

1 **A computational study of soot formation and flame structure of coflow**
2 **laminar methane/air diffusion flames under microgravity and normal**
3 **gravity**

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1 A computational study of soot formation and flame structure of coflow 2 laminar methane/air diffusion flames under microgravity and normal 3 gravity

4 A numerical study is conducted of methane–air coflow diffusion flames at
5 microgravity (μg) and normal gravity (1g), and comparisons are made with
6 experimental data in the literature. The model employed uses a detailed gas phase
7 chemical kinetic mechanism that includes PAH formation and growth, and is
8 coupled to a sectional soot particle dynamics model. The model is able to
9 accurately predict the trends observed experimentally with reduction of gravity
10 without any tuning of the model for different flames. The microgravity sooting
11 flames were found to have lower temperatures and higher volume fraction than
12 their normal gravity counterparts. In the absence of gravity, the flame radii
13 increase due to elimination of buoyancy forces and reduction of flow velocity,
14 which is consistent with experimental observations. Soot formation along the
15 wings is seen to be surface growth dominated, while PAH condensation plays a
16 more major role on centerline soot formation. Surface growth and PAH growth
17 increase in microgravity primarily due to increases in the residence time inside
18 the flame. The rate of increase of surface growth is more significant compared to
19 PAH growth, which causes soot distribution to shift from the centerline of the
20 flame to the wings in microgravity.

21 **Keywords:** laminar diffusion flame, methane–air, microgravity, soot formation,
22 numerical modelling

23 Introduction

24 Soot formation plays a crucial role within the context of fire spreading in microgravity,
25 i.e. fire safety in manned spacecraft. Elongated formation residence times due to the
26 absence of buoyancy forces substantially increase concentration of soot in flames under
27 microgravity conditions, enhancing radiation heat transfer and fire spreading. As such,
28 flames established in microgravity rely on empirical models for characterisation of soot
29 formation [1,2]. The emergence of detailed soot models has facilitated the gain of in
30 depth insight on soot production phenomena in flames established in microgravity.

1 Furthermore, a scenario, which simplifies some aspects of soot generation, is highly
2 desirable for academic study. Soot particles are generated in high temperature, fuel rich
3 regions when burning a variety of fuels. Soot formation is a complex phenomenon,
4 which involves several concurrent chemical and physical processes. Reducing gravity in
5 a coflow diffusion flame is such an example. Elimination of buoyancy simplifies the
6 flow field and simultaneously enhances residence time within the flame. By
7 investigating the structural differences between normal and microgravity flames, an
8 improved understanding of the factors that affect soot formation can be achieved.

9 Both experimental and numerical methods have been utilized to quantify the
10 effect of gravity on soot formation. Ku et al. [3,4] experimentally investigated soot yield
11 and morphology for various coflow diffusion flames under reduced gravity in the
12 NASA Lewis Research Center's 2.2-s drop tower. In microgravity the soot volume
13 fractions measured using laser extinction increased by a factor 2-4 compared to the
14 same flame under normal gravity. A factor of 2 increase in the measured maximum soot
15 volume fraction and a 40% increase in primary particle diameter at 0g for laminar
16 nitrogen diluted acetylene jet diffusion flames have also been reported by Megaridis et
17 al. [5,6]. The trend of increase in both soot mass and size have been experimentally
18 observed by Walsh et al. [7], Jeon and Choi [8], and Reimann et al. [9,10] using
19 multiple optical measurement techniques. Experimental observations of non-buoyant
20 round laminar jet diffusion flames were made for various jet flames burning in still air
21 by Diez et al. [11] on board the Space Shuttle Columbia to correlate soot volume
22 fraction with estimated mixture fraction for a selection of fuels/pressures. These
23 experiments provide an important foundation for the study of microgravity soot
24 formation.

1 Simulations of a laminar ethylene-air diffusion flame burning in quiescent air
2 were conducted by Kaplan et al. [12] in reduced gravity. Their unsteady simulations
3 showed a factor of 10 increase in peak soot volume fraction from 1g to μg . More
4 importantly, their simulation did not reach steady-state conditions within 3s of imposing
5 microgravity which is the time period in drop tower experiments. Kong and Liu [13,14],
6 Liu et al. [15], and Charest et al. [16,17] also numerically investigated the influence of
7 gravity on laminar coflow diffusion flames, demonstrating that the μg flame has lower
8 temperatures, thicker soot regions and higher soot volume fractions than the 1g flame.
9 These numerical investigations employed semi-empirical two equation models which
10 based the formation and growth of soot only on acetylene. While these works were
11 major advances in modelling ability, they were performed for conditions different from
12 available experimental data. Therefore, no direct comparison with experiment was
13 available, making it difficult to assess and analyze the computational models.

14 Recently, Ma et al. [18] measured soot and temperature in a set of methane/air
15 coflow laminar diffusion flames in normal and microgravity on board the International
16 Space Station. From the experiments, the peak soot volume fractions of the
17 microgravity flames were found to be higher than the normal gravity flames by a factor
18 of 4–8. The distribution of peak soot also shifted from the centerline at 1g to the wings
19 at μg . The flame temperature in μg is shown to be lower than its 1g counterpart due to
20 higher radiative loss. The study provides high-fidelity experimental data for model
21 validation.

22 A detailed numerical model, *CoFlame*, was developed by Thomson, Dworkin,
23 and coworkers [19–26]. It employs a polycyclic aromatic hydrocarbon (PAH) based
24 sectional particle dynamics model, tracking shape and size distribution. *CoFlame*
25 accounts for all processes that are known to occur in soot formation, including PAH

1 growth, particle inception, surface growth via hydrogen-abstraction-carbon-addition
2 (HACA) and PAH condensation, oxidation, coagulation, fragmentation, gas phase
3 scrubbing, soot particle diffusion, and radiation [27,28]. In the present study, CoFlame
4 is combined with a chemical mechanism by Chernov et al. [25] that includes numerous
5 reaction pathways for formation of PAHs up to five aromatic rings. Additionally,
6 conjugate heat transfer (CHT) between the fluid streams and the fuel tube is modeled to
7 capture the effect of fuel tube preheating [26]. This highly detailed model is employed
8 to predict soot in a set of methane/air coflow laminar diffusion flames [18] in normal
9 gravity and microgravity. The mechanisms by which microgravity affects soot and
10 flame structure are investigated. The present work builds upon the work done by Ma et
11 al. [18], in which the same flame was studied, by linking the solid phase to PAHs up to
12 A5, and considering conjugate heat transfer. Therefore, further insights into the effects
13 of gravity were obtained.

14 **Flame and model description**

15 The burner consists of an annular fuel tube located in the center of a 76mm wide square
16 duct with rounded corners. Details of the burner construction and operation are provided
17 in [18]. Six methane/air flames have been studied under both microgravity and normal
18 gravity. Details are provided in Table 1. The square duct is approximated as a co-
19 annular tube with an identical cross-sectional area (radius $r_0=4.288$ cm). Soot volume
20 fractions and temperatures were measured using a color-ratio pyrometry approach based
21 on a color DSLR camera [18]. Although most of the discussion will be focused around
22 the first two flames, for the soot models it is essential to illustrate that the model
23 predictions are applicable to a range of conditions. Narrow application of models tends
24 to hinder their development and the value and knowledge obtained from their studies,
25 hence the inclusion of the other data.

1 CoFlame [29] is adapted to model the flames in this investigation. For the
2 gaseous phase, the fully coupled elliptical conservation equations for mass, momentum,
3 energy, species mass fraction, soot aggregate number densities, and primary particle
4 number densities are solved. The model utilizes the axi-symmetrical nature of the flame,
5 and equations are solved in the two-dimensional (z and r) cylindrical co-ordinate
6 system. All boundary conditions including inlet velocities are set in accordance to the
7 values reported in [18,30]. A detailed description of the governing equations, and
8 solution methodology can be found in [19–26,29]. The Chernov et al. [25] chemical
9 mechanism, being reduced to 94 species and 754 reactions by eliminating methanol
10 chemistry, describes the oxidation of the fuel and the formation of PAHs.

11 The three heaviest PAH species in the mechanism has been chosen as the PAH
12 species that interact with the soot particles which are benzo(a)pyrene ($C_{20}H_{12}$),
13 benzo(a)pyrenyl ($C_{20}H_{12}$), and benzo(ghi)fluoranthene ($C_{18}H_{10}$). The 5-ring PAHs are
14 used as nucleating species because they are more representative of PAH stacks observed
15 in soot particles [31–33]. The 5-ring model is slightly more versatile. An extensive
16 discussion on this subject can be found in Eaves et al. [29] and Saffaripour et al. [33].
17 Nucleation is modeled based on the collision of two PAHs in the free-molecular regime
18 as in [33]. The HACA mechanism [34,35] is used to describe soot surface growth. As in
19 previous studies [26,33,36], a constant value of 1.0 is used for the surface site density
20 parameter α , which is a semi-empirical parameter that reconciles the inaccuracies of
21 treating the soot surface like a corresponding PAH. PAH condensation is modelled
22 based on collision theory between PAH molecules and aggregates [37]. All model
23 parameters are consistent with parameters described by Eaves et al. [29], which was
24 validated for soot formation in both methane and ethylene coflow flames. Details of the
25 sectional model and transport equations can be found in [29].

1 **Results and discussion**

2 Temperature is important to soot formation; especially the hydrogen abstraction
3 reaction which has the highest activation energy [37]. Radiation heat losses from soot
4 will influence flame temperature as well. According to Liu et al. [15], these radiation
5 heat losses are more significant at microgravity than 1g and completely alter the
6 structure of the flame.

7 In order to examine the performance of the model in predicting temperature and
8 demonstrate structural differences between the microgravity and normal gravity flames,
9 computed and measured temperature maps for the F1 and F1g flames are depicted in
10 Figure 1 and radial temperature profiles at three different heights are provided in Figure
11 2. Ma et al. [18] estimated that measured microgravity soot temperature is
12 underestimated by up to 50K due to variation in range of the soot emission properties
13 (i.e., dispersion exponent). The 50K underestimation is in addition to other uncertainties
14 associated with the color-ratio pyrometry. The soot self-absorption along the optical
15 pathway of collection is not accounted for, which can lead to an additional significant
16 underestimation of the soot temperature. Also, the absorption coefficient fields (or
17 equivalently the soot emissivity fields) at the different wavelengths used by Ma et al.
18 [18] are not directly measured, which means that a model for soot refractive index as a
19 function of wavelength is implicitly required, which can lead to significant
20 discrepancies [38]. Hence, temperature predictions agree with experimental data for
21 both flames. The exception is that temperature is overpredicted on the centerline, which
22 is due to underprediction of soot and resulting radiation. The overprediction of
23 temperature for the microgravity flame is more significant than for the normal gravity
24 flame, consistent with [15]. The maximum temperature difference between the
25 predicted and experimental data for the microgravity flame is twice that of the normal

1 gravity flame. Similar trends in prediction of temperature are observed for the other
2 flames for which results are included in the Supplementary Data.

3 The flame radius (r_f) is defined as the maximum radius of the 1% CH
4 concentration isopleth [30]. A maximum increase in flame radius due to an absence of
5 gravity is observed in the F1 flame, where r_f is 60% larger than in the F1g flame. For
6 the F2 and F3 flames, r_f is predicted to increase 47% and 21%, respectively. For the
7 normal gravity flames, r_f is relatively constant between the three flames (between 5.1–
8 4.2mm) which is consistent with the experimental and theoretical studies that r_f is
9 proportional to $St^{0.25}$ (St is the Stokes number) [30]. In the microgravity flames
10 however, the average velocity which is the dominant variable in St , is highly dependent
11 on the inlet flow rate. Therefore, the lowest increase in r_f is observed for the F3 flame
12 which has the closest velocity field to its 1g counterpart.

13 The predicted soot volume fraction maps and the experimental data from [18]
14 for the F1 and F1g flames are demonstrated in Figure 3. The computed maximum soot
15 volume fraction is underpredicted in both 1g and μg by a factor of 10 compared to the
16 experimental data. Although the measured microgravity soot volume fraction is possibly
17 overestimated by up to a factor of 2 due to the uncertainty of the soot emission
18 properties [18], that is not enough to put the predicted peak soot within the range of the
19 uncertainty of the experimental data. The same trend of underprediction of peak soot in
20 comparison to experimental data is found for the remaining flame soot predictions as
21 well, which are depicted in the Supplementary Data. Despite underprediction, the
22 predicted radial distribution of soot particles is in agreement with the experimental data.
23 It should be noted that the soot predictions are in better agreement with experimental
24 data compared to earlier numerical studies of soot in these flames [18] (see Table 2).

1 Despite the underprediction of soot volume fraction, the model is able to capture
2 the transition of location of peak soot between different flames which was observed
3 experimentally. For the third set of the flames, the absence of gravity causes the peak
4 soot volume fraction to increase but the location of the peak remains on the centerline of
5 the flame. However, for the other two sets, in addition to increases in the maximum soot
6 volume fraction in microgravity, a shift is observed in the location of the peak. The peak
7 soot concentration location moves from the centerline in the 1g flame toward the
8 annular region of the flame (wings) in its μg counterpart. The soot model is accurately
9 capturing this transition for all flames.

10 Thermal diffusion forces can shift heavy molecules and soot in the coflow
11 diffusion flames towards the centerline, as was shown in Giovangili and coworkers
12 [39,40]. These findings emphasize the importance of including relevant transport
13 phenomena in the model which was incorporated in the soot model for this study.
14 However, the shift observed here is not an artefact of changing transport model
15 parameters. Instead, the shift is between two flames with different gravitational
16 conditions observed experimentally and in the numerical predictions. Since the
17 convection forces due to buoyancy are much stronger in the 1g flame it would lessen the
18 relative effect of diffusion forces on the shift compared to the μg flame. Therefore, the
19 diffusion forces are having a much stronger effect in pushing the particles towards the
20 centreline. Despite this effect, the soot is formed mostly on the wings. Hence, the shift
21 cannot be attributed solely to the diffusion forces although they are present and are
22 responsible for part of the shift towards the centerline in all of the coflow diffusion
23 flames.

24 In addition, the predictions show a factor of 10 increase in soot volume fraction
25 in microgravity which is in good agreement with the experimental.

1 A sophisticated soot and PAH simulation too the Coflame code includes state-
2 of-the-art soot formation processes and PAH growth pathways. The Coflame code has
3 been validated against experimental data for an axi-symmetric pipe experiencing a
4 sudden expansion, and substantial data available for ethylene–air and methane–air
5 diffusion flames [29], high pressure [41], partially premixed combustion [19], diluted
6 flames, and different burner geometries [21]. In addition, the soot model has been
7 validated against experimental data for several premixed flames [32,42]. The facts that
8 the validation here has been extended to six different flames and the model is able to
9 correctly capture the trend of increase of soot due to micro-gravity and the transition of
10 soot maximum concentration location from wings toward centerline, in addition to
11 previous extensive validation of the Coflame code, allow further insights to understand
12 soot formation in microgravity.

13 Predicted soot peak primary particle diameter, primary particle number density,
14 and number of particles per aggregate along the pathline of maximum soot in the
15 annular region, and along the centerline of the flames are summarized in Table 3. The
16 particle pathline follows the particle trajectory and includes the thermophoretic velocity.
17 In the F3g flame, the experiments and model prediction shows the soot to be formed in
18 a very narrow area in the vicinity of the centreline, causing the wings to be irrelevant to
19 this analysis. In general, the model predicts an increase in particle diameters in
20 microgravity. However, the rate of increase is higher on the wings than on the centerline
21 which is also consistent with the trend of increase in soot volume fraction. The increase
22 in the particle diameter observed is within the range that is experimentally observed by
23 Ku et al. [3] and Reimann et al. [9].

24 An interesting trend is observed for the number density of particles. On the
25 wings, the number density remains constant as the gravity disappears. Number density

1 of particles is primarily controlled by the nucleation process [29,43]. Nucleation is
2 assumed to be a function of collision rate between PAHs [37] which is proportional to
3 temperature and PAH concentrations. Predicted nucleation on the wings is highest
4 downstream of the fuel tube near the flame front where the temperature is close to its
5 peak. The fact that number density remains unchanged indicates that the cumulative
6 effect of different parameters is such that nucleation remains constant. One explanation
7 is that since the nucleation process in this region is fast, the increase in the residence
8 time due to microgravity has an insignificant effect on the nucleation rate. On the
9 centerline, an increase in number density of particles is predicted as the buoyancy is
10 eliminated, which can be attributed to the increase of PAH concentration (see Figure 5)
11 as well as residence time. Primary particles per aggregate (n_p) increases by reducing
12 buoyancy both on the wings and on the centerline. Since the number density of particles
13 is also increased on the centerline, the effect of increasing residence time in
14 microgravity conditions is more pronounced on n_p .

15 Mass based contributions of different growth and oxidation processes for the F1
16 and F1g flames along the wings and centerline are presented in Figure 4. This figure
17 illustrates the integrated contributions of different processes to soot mass, with respect
18 to time along the particle pathline. The horizontal axis indicates the height up to which
19 the values are time-integrated. This set of data can be used to better understand the
20 formation pathways of soot and interpret the mechanism for the shift of the location of
21 peak soot from the centerline to the wings in microgravity. A common pattern among
22 all cases is that soot mass yield starts with the nucleation of soot particles. However, the
23 PAH contribution to soot mass does not prevail further downstream. The halt of PAH
24 addition before entering the oxidation zone is a consequence of PAH depletion. After
25 the inception stage, HACA growth comes into effect and continues all the way through

1 the oxidation zone where all soot mass is depleted by O₂ and OH oxidation. Since these
2 are non-sooting flames, the mass production of HACA + PAH addition balance the
3 mass consumption via oxidation downstream of the flame front.

4 From comparisons of contributions of different processes in 1g and μg, the
5 following observation is made. On the wings, the soot mass growth is dominated by
6 HACA, accounting for 89% of peak soot mass in 1g and 93% in μg condition. In
7 microgravity on the wings, the contribution of HACA increased by a factor of 12 while
8 the PAH contribution to soot mass increased by a factor of 7. On the centerline a
9 different behaviour is observed; HACA is still responsible for most of the soot mass. In
10 normal gravity, 93% of the peak soot mass is gained through HACA growth. Under
11 microgravity the balance for soot yield is shifted more toward PAH addition. PAH
12 addition accounts for 36% of the total soot mass in the F1 flame on the centerline. In
13 comparison to 1g, the mass addition by PAH is increased by a factor of 11 while HACA
14 only increased by a factor of 4. From these observations, it can be concluded that slower
15 increase in HACA on the centerline is the reason for the shift of peak soot location from
16 the centerline in the F1g flame to the wings in the F1 flame.

17 Figure 5 illustrates the variation of acetylene, benzene, and pyrene mole
18 fractions, which are all soot precursors, along the wings and centerline of the F1 and
19 F1g flames. As presented in Table 4, the carbon addition step in HACA for soot and
20 PAHs is most often considered to be by acetylene and the hydrogen abstraction to be via
21 H radical [28,34,35]. Therefore, studying acetylene behaviour in the flame can be used
22 to comprehend HACA growth. The C₂H₂ levels in the F1g flame are comparable on the
23 wings and on the centerline. By comparing these results with the HACA contribution to
24 soot mass in Figure 4, it can be concluded that the balanced concentration of C₂H₂ on
25 the wings and centerline translates to balanced HACA growth in these regions.

1 However, in the absence of buoyancy acetylene concentrations decrease both on the
2 wings and on the centerline while HACA increases considerably. Hence, the increase of
3 HACA is not a direct result of increasing C_2H_2 . Instead, HACA is increased due to more
4 consumption of C_2H_2 which leads to lower levels of C_2H_2 under microgravity. As
5 presented in Table 4, the reactions with highest rate for HACA are the C_2H_2 addition
6 and H abstraction. H radical, often the bottleneck of the HACA growth [21,29], also
7 displays a similar behaviour; the increase in HACA growth as a result of prolonged
8 residence time comes at the cost of consumption of H and C_2H_2 from the gas phase.

9 Benzene and pyrene represent the behaviour of small aromatics and moderate
10 PAH species in the flame, respectively. The model predictions suggest that as gravity
11 reduces, the concentration of benzene and pyrene increase everywhere but not at the
12 same rate. The increase in aromatic compounds on the centerline is more significant
13 than on the wings. Peak benzene mole fraction is increased by 71% on the centerline
14 while it increases by only 14% on the wings. Also, these results show that the
15 relationship between benzene and pyrene is not direct. Aforementioned growth in
16 benzene concentration results in a 150% increase of pyrene on the wings and a factor of
17 5.6 increase in pyrene concentration on the centerline. The enormous increase in heavier
18 PAH species on the centerline is also one of the reasons for PAH addition becoming
19 more dominant on the centerline.

20 Finally, to leverage the power of the numerical model, a sensitivity analysis is
21 conducted to quantify the effect of HACA soot growth rate and residence time on soot
22 formation. The soot particle surface reactivity parameter (α) [21] which directly
23 influences HACA, is increased from 1 to 8. While these values are unphysical, they
24 allow further understanding on needed areas of model development. The predicted soot
25 volume fraction in the F1 flame with various α along the wings and centerline are

1 depicted in Figure 6. The results with $\alpha = 1$ represent the results provided in previous
2 sections and the rest shows the effect of enhancing HACA surface growth. On the
3 wings, the soot volume fraction monotonically increases with α . With $\alpha = 8$, enough
4 soot is produced to overcome the previous underprediction. These results show that
5 there is a potential to investigate new surface growth venues to upgrade soot model
6 predictions on the wings. Contrarily, on the centerline, soot mass becomes insensitive to
7 α due to depletion of H radical. Although appreciable amounts of C_2H_2 are present in
8 the mixture, due to a lack of hydrogen-abstraction, soot particles are unable to absorb
9 them. These results reiterate previous study conclusions which suggested that the
10 underprediction of centerline soot is a result of PAH chemistry [23,36]. Thus, it is
11 necessary to revisit PAH growth pathways and invest in introducing novel growth
12 pathways for PAHs.

13 In flame F1g, when α is increased from 1 to 8, the peak soot location moves
14 toward the wings. However, with $\alpha=8$, the model overpredicts soot volume fraction by
15 an order of magnitude, suggesting that in that case the model is not correct. Moreover,
16 with $\alpha=8$ the difference between the soot volume fraction on the wings and centerline of
17 the F1g flame is not drastic, as it is for the μg flame.

18 The absence of gravity causes an increase in the residence time in the flame.
19 This increase in residence time as has been shown in Figure 1 is accompanied by a
20 change in temperature field both in the temperature magnitude and distribution. To
21 separate the influence of temperature and residence time, a numerical experiment has
22 been conducted in which the temperature field has been fixed to the F1 flame
23 temperature but the gravity has been set to 1g. This configuration will impose a
24 residence time close to the F1g flame. The result is a complete disappearance of soot
25 particles everywhere in the flame. A substantial reduction of PAH precursors (propargyl

1 and *i*-C₄H₅) is caused by the lack of formation of PAHs. As a result, no soot is formed
2 in this configuration. This experiment alludes that the temperature field distortion as a
3 result of the absence of gravity has a negative effect on soot formation. In conclusion,
4 the increase of soot formation as a result of an increase in residence time is so
5 significant that it can overcome the temperature effect and leads to an order of
6 magnitude increase in soot formation.

7 **Conclusions**

8 Laminar, sooting coflow methane–air diffusion flames at microgravity and normal
9 gravity are numerically simulated using a detailed PAH-based sectional soot model. The
10 trends observed in the experimental data are captured by the model. The flame
11 temperature in microgravity is shown to be lower than its 1g counterpart with
12 experimental data. Computations of both 1g and μ g flames yield agreement in flame
13 shape with experimental results. As in other investigations, soot volume fraction is
14 underpredicted. The peak soot volume fraction in microgravity flames was found to be
15 higher than in normal gravity by a factor of 10. Primary particle diameters and number
16 of particles per aggregate were found to increase in the absence of gravity.

17 Experimental data shows a shift of distribution of peak soot from the centerline
18 of the flame at 1g to the wings at μ g. The computational model successfully captured
19 this soot migration and it was attributed to a considerable increase in HACA surface
20 growth on the wings in microgravity.

21 Soot formation is seen to be surface growth dominated in the studied flames,
22 while the significance of PAH growth increases on the centerline for microgravity
23 flames. HACA and PAH growth increase along the wings and centerline of the flames
24 with elimination of gravity due to increases of the growth residence time. The HACA

1 growth on the wings was found sensitive to the soot surface reactivity parameter
2 whereas soot on the centerline is shown to be insensitive to it due to reduced H radical
3 concentration.

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12

1 **Tables**

2 Table 1 Burner geometry and inlet boundary of the simulated laminar diffusion
3 flames

Flame des.	Gravity	Velocity(cm/s)		Fuel tube ID (mm)
		Fuel	Air	
F1	0	46	18	3.23
F1g	1g	46	18	3.23
F2	0	169	18	1.59
F2g	1g	169	18	1.59
F3	0	169	40	1.59
F3g	1g	169	40	1.59

4

5

1 Table 2 Comparison of maximum soot volume fraction predicted by Coflame
2 model, and Ma et al. [18] with experimental data [18].

Flame	Coflame	Ma et al.	Exp.
F1	0.5	0.2	4.4
F1g	0.05	0.18	0.5
F2	0.2	0.16	2.0
F2g	0.02	0.12	0.4
F3	0.1	0.12	1.3
F3g	0.01	0.09	0.4

3

4

1 Table 3 Predicted maximum soot primary particle diameter (D_p), number density
 2 of primary particles (N_p), and number of particles per aggregate (N_p/N) along the
 3 pathline of maximum soot on the wings (W) and the centerline (C) of a set of methane
 4 flames under normal and microgravity conditions.

Flame	D_p (nm)		N_p (#/cm ³)		N_p/N	
	W	C	W	C	W	C
F1	23	17	5.E10	1.E11	12	26
F1g	9	9	5.E10	7.E10	7	9
F2	18	14	4.E10	8.E10	10	20
F2g	9	8	4.E10	5.E10	5	6
F3	11	11	5.E10	7.E10	9	14
F3g	–	9	–	4.E10	–	4

5

6

1 Table 4 HACA-based soot surface growth and oxidation reactions [34],
 2 $\square = \square\square\square\square - \square\square/\square\square$.

No.	Reaction	$A \left(\frac{\square\square^3}{\square\square\square.\square} \right)$	b	E_a $\left(\frac{\square\square\square\square}{\square\square\square} \right)$
S1	$C_{\text{soot}}\text{-H} + \text{H} \rightleftharpoons C_{\text{soot}}\square + \text{H}_2$	4.2×10^{13}	0.0	13.0
S2	$C_{\text{soot}}\text{-H} + \text{OH} \rightleftharpoons C_{\text{soot}}\square + \text{H}_2\text{O}$	1.0×10^{10}	0.73	1.43
S3	$C_{\text{soot}}\square + \text{H} \longrightarrow C_{\text{soot}}\text{-H}$	2.0×10^{13}	0.0	0.0
S4	$C_{\text{soot}}\square + \text{C}_2\text{H} \longrightarrow C_{\text{soot}}\text{-H} + \text{H}$	8.0×10^7	1.56	3.8
S5	$C_{\text{soot}}\square + \text{O}_2 \longrightarrow 2\text{CO} + \text{product}$	2.2×10^{12}	0.0	7.5
S6	$C_{\text{soot}}\text{-H} + \text{OH} \longrightarrow \text{CO} + \text{product}$		$Y_{\text{OH}} = 0.13$	

3

4

1 **List of figure captions**

2 Figure 1. Computed temperature (K) isopleths and measured soot temperature (K)
3 from [18] under (left) normal gravity and (right) microgravity for a CH₄ flame with 3.23
4 mm ID nozzle and average fuel speed of 46 cm/s.

5 Figure 2. Computed temperature (K) radial profiles and measured soot
6 temperature (K) from [18] at three different heights above the burner under (left) normal
7 gravity and (right) microgravity for a CH₄ flame with 3.23 mm ID nozzle and average
8 fuel speed of 46 cm/s.

9 Figure 3. Comparison of soot volume fraction maps (ppm) with measured soot
10 volume fractions (ppm) from [18] under (left) normal gravity and (right) microgravity
11 for a CH₄ flame with 3.23 mm ID nozzle and average fuel speed of 46 cm/s.

12 Figure 4. Total mass contribution by HACA surface growth, PAH addition
13 (nucleation and condensation), and the amount of soot mass depleted by oxidation for a
14 soot particle travelling (top) along the pathline of maximum soot on the wings, and
15 (bottom) along the centerline for a CH₄ flame with 3.23 mm ID nozzle and average fuel
16 speed of 46 cm/s under normal and microgravity.

17 Figure 5. Acetylene, benzene and pyrene mole fractions as a function of height
18 above the burner along the pathline of maximum soot on the wings and along the
19 centerline for a CH₄ flame with 3.23 mm ID nozzle and average fuel speed of 46 cm/s
20 under normal and microgravity.

21 Figure 6. Predicted soot volume fraction (ppm) using various surface reactivity
22 parameter as a function of height above the burner (top) along the pathline of maximum
23 soot on the wings and (bottom) along the centerline for a CH₄ flame with 3.23 mm ID
24 nozzle and average fuel speed of 46 cm/s under microgravity.