# Evaluation of Finite Element Software for 

## Pavement Stress Analysis

## By

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A project

## Presented to Ryerson University

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# Evaluation of Finite Element Software for 

## Pavement Stress Analysis

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#### Abstract

Different approaches are usually taken when designing flexible and rigid pavement: the rigid concrete slab carries major portion of the traffic load; while for flexible pavement, external loads are distributed to the subgrade because of the relative low modulus of elasticity of asphalt layer comparing to concrete in the case of rigid pavement.


Pavement engineering has gone through major developments; the transition from Empirical Design Method to Mechanistic-Empirical Methods is becoming a near-future trend. The Mechanistic-Empirical Method has two components: (1) stress, strain and deflection are calculated based on analyzing mechanical characteristics of materials; (2) critical pavement distresses are quantitatively predicted by experimental calibrated equations. Hence, stress analysis has become an important role in pavement engineering.

The most practical and widely used stress analysis method for flexible pavement is Burmister's Elastic Layered Theory; and for analyzing rigid pavement is Finite Element Method. KENSLABS and STAAD-III are both Finite Element software; KENSLABS is designed specifically for concrete pavement stress analysis, therefore it is more user-friendly for pavement design; STAAD-III is more suitable for general plane and space structures. The project compares the use of both software for stress analysis in rigid pavement in term of simplicity and precision.

## Acknowledgements

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## Executive Summary

The first chapter of this report briefly introduced the pavement types, available pavement design methods and objectives of this project. There are two major types pavement: flexible or asphalt pavement and rigid or concrete pavement. As an important component of Mechanistic-Empirical Method, the most practical stress analysis method for flexible pavement is Burmister's Elastic Layered Theory and for rigid pavement is Finite Element Method.

The second chapter of this report reviewed critical stress, strain and deflection, which would induce major types of distress in flexible pavement and rigid pavement. Horizontal tensile strain at the bottom of the asphalt and the vertical compressive strain at the top of the subgrade are two critical measurements for predicting flexible pavement distresses such as fatigue cracking and rutting. Flexural stress in concrete slab is a major factor controlling rigid pavement distress such as transverse cracking. The application of Burmister's Elastic Layered Theory in flexible pavement design and Finite Element Method in rigid pavement design will also be elaborated in this Chapter.

Two available Finite Element software, KENSLABS and STAAD-III are compared in Chapter 3 by solving two pavement design examples. As a software designed specifically for rigid pavement stress analysis, KENSLABS has several advantages in terms of simplicity; as a powerful while complicated finite element software, STAAD-III is more suitable for analyzing general plane and space structures. The last chapter presents the conclusions of this project.

## Chapter 1: Introduction

### 1.1 Definition of pavement types

There are mainly two types of pavement structure currently in use around the world: flexible pavement and rigid pavement. The major difference between these two types of pavement is: flexible pavement has an asphalt concrete surface layer at the very top, base and/or subbase layer in the middle, and natural subgrade layer at the very bottom; while for rigid pavement, regardless as continuously reinforced concrete or lean concrete pavement, it has a concrete slab sitting right above the soil, a layer of base course may sit in-between. Compositions of two types of pavement are simply illustrated as below in Fig. 1 and Fig. 2:


Fig. 1 Flexible Pavement Composition ${ }^{(2)}$


Fig. 2 Rigid Pavement Composition ${ }^{(2)}$

### 1.2 Pavement design methods

Pavement design has evolved from art to science during the past two centuries. The first asphaltpaved roadway in the United States was constructed in 1870 at Newark by a Belgian Chemist named Edmund J. DeSmedt; ${ }^{(1)}$ and the first Portland cement concrete pavement in the United States was believed to be laid in 1893 in Bellefontaine, Ohio by J. Y. McClintock. ${ }^{(1)}$

As a simple design approach, the Empirical Design Method has been adopted by many transportation agencies nowadays. By inputting environmental, material and traffic data, Empirical Method is able to provide design engineers the pavement design thickness by using calibrated equations and Nomographs. Unlike the Empirical Design Method, MechanisticEmpirical Method is mainly based on analyzing mechanical characteristics of materials. Stress, strain and deflection can be calculated by using Elastic Layered Theory (flexible pavement) and Finite Element Method (rigid pavement); then critical pavement distresses can be quantitatively predicted by experimental calibrated models. ${ }^{(3)}$ With the development of high-speed calculation with computer software, critical stresses and deflections of pavement under traffic and environmental loadings is much more convenient to be predicted than the past.

### 1.3 Objectives

The objectives of this project are firstly, to review critical stress, strain and deflection that would induce major types of distress in flexible pavement and rigid pavement. Distress is an important consideration in terms of pavement performance. The magnitude/frequency of loading and number of load repetitions during the design period are major factors contribute to the damage effects of pavement structures. For flexible pavement, primary distresses are rutting, fatigue
cracking and thermal cracking; ${ }^{(3)}$ for concrete pavement, major distresses are transverse cracking and faulting. ${ }^{(3)}$

Secondly, available stress analysis methods will be reviewed. As a realistic approach for analyzing flexible pavement responses, Burmister's Elastic Layered Theory will be briefly introduced; and as a common structural analyzing tool, Finite Element Method and how the method is applied in solving rigid pavement problem will be elaborated.

Finally, two available Finite Element software: $K E N S L A B S$ and $S T A A D-I I I$ will be compared by solving two pavement design examples. The results from running two software will be explained and analyzed. A brief introduction on design procedures and reports generated by both software are included in Appendices.

# Chapter 2: Review of available stress analysis methodology <br> <br> For flexible and rigid pavement 

 <br> <br> For flexible and rigid pavement}

From the standpoint of stress analysis, the essential differences between flexible and rigid pavement is: because of the large modulus of elasticity of the rigid concrete slab, major portion of the load transferred from the tires of the vehicles will be carried by the concrete slab itself, thus the main goal of the rigid pavement design is to build a strong concrete slab. While for flexible pavement, external loads will be distributed to the subgrade layer because of the relative low modulus of elasticity of asphalt layer and other upper layers, the design process consists of choosing optimum layer combinations and structural design of each layer component.

### 2.1 Stress analysis for flexible pavement

The simplest way to solve flexible pavement problem is to consider the pavement system as a homogeneous half-space with infinite area and depth, the original theory developed by Boussinesq $(1885)^{(1)}$ was based on a concentrated load applied on an elastic half space, however, this assumption is not realistic. First of all, the subgrade soils are not elastic even though under the moving traffic load some of the deformation is recoverable; secondly, flexible pavement is a layered system with better materials at the top; each layer has different material characteristics, such as modulus of elasticity E and Poisson's ratio $v$.

The Elastic Layered Theory developed by Burmister (1943) is a realistic solution to treat the flexible pavement as a multi-layered system. Some basic assumptions when applying this theory are: ${ }^{(2)}$

1. Pavement layers extend infinitely in the horizontal direction.
2. The bottom layer (usually the subgrade) extends infinitely downward.
3. Materials are not stressed beyond their elastic ranges.
4. Continuity conditions are satisfied at the layer interfaces.


Fig. 3 Flexible pavement stress analysis ${ }^{(2)}$

From the above illustration, according to the mechanistic theories, at any point within the layered system, there are 9 stresses: 3 normal stresses $\sigma_{\mathrm{z}}, \sigma_{\mathrm{r}}, \sigma_{\mathrm{t}}$ and 6 shear stresses, $\tau_{\mathrm{tz}}, \tau_{\mathrm{z}}, \tau_{\mathrm{tr}}, \tau_{\mathrm{rr}}, \tau_{\mathrm{ra}}, \tau_{\mathrm{zr}}$, and the following relationships exist:
$\tau_{\mathrm{tz}}=\tau_{\mathrm{zt}}$
$\tau_{\mathrm{tr}}=\tau_{\mathrm{rt}}$
$\tau_{\text {rz }}=\tau_{\text {zu }}$

Typically, flexible pavement is designed to put materials with higher modulus of elasticity at the upper layers in order to reduce stresses and deflection transferred to the subgrade layer. Basing on the elastic body theory, the basic stress and strain relationship should be satisfied within this axially symmetrical system.

### 2.1.1 Vertical compressive stress



Fig. 4 Distribution of vertical compressive stress with Modulus of Elasticity ${ }^{(2)}$

From above Fig. 4, vertical compressive stress at the top surface $\sigma_{z}=-P$, where $P$ stands for the external load applied on the unit circular area. According to the graph, it is self-explanatory that to increase the modulus of elasticity, or in other words, to increase the rigidity of the surface layer can reduce vertical compressive stress transferred to lower layers. This explains why we design the flexible pavement as materials with higher modulus of elasticity at the upper layers. On the other hand, under same traffic loading conditions, soft asphalt mixtures or excessive air voids or inadequate compaction in any pavement layer are tend to be more susceptible to rutting damages. ${ }^{(3)}$

From the Fig. 5 below it could also be seen that $\sigma_{z}$ will be reduced when $h_{1} / h_{2}$ increase or $1 / h_{2}$ decrease. That means if $h_{2}$, the thickness of the base course layer should remain constant, to increase $h_{1}$, the thickness of surface layer will bring the vertical compressive stress down.


Fig. 5 Distribution of vertical compressive stress with layer thickness ${ }^{(2)}$

### 2.1.2 Horizontal shearing stress

Fig. 6 simply illustrates the distribution of shearing stress over layers. From the figure, the maximum $\tau_{\text {rz }}$ occurs approximately at the midpoint of the surface layer; and at the top surface horizontal shear stress $\tau_{\mathrm{rz}}=0$. To increase the modulus of elasticity of the surface layer will cause the shearing stress to be increased. Inadequate shear strength of the asphalt mixture will induce two-dimensional movement under heavy traffic, ${ }^{(3)}$ and this will induce two-dimensional rutting damage. As a major type of distress in asphalt pavement, rutting appears as a surface depression in the wheel paths. Rutting damage relates to material properties and loading, major portion of the rutting occur within the asphalt layer.


Fig. 6 Distribution of horizontal shear stress with Modulus of Elasticity ${ }^{(2)}$

### 2.1.3 Horizontal tensile stress

Horizontal tensile stress would normally be developed at first interface, right underneath the asphalt layer. The simplified distribution pattern is as shown below Fig. 7.


Fig. 7 Distribution of horizontal tensile stress with Modulus of Elasticity and layer thickness ${ }^{(2)}$

The information we can get from Fig. 7 is: to increase $E_{1} / E_{2}$ or reduce $h_{1} / h_{2}, \sigma_{r}$ will significantly increase; but when asphalt layer thickness approaches zero, there would be no horizontal tensile stress exists.

When horizontal tensile stress at the bottom of the asphalt layer exceed its tensile strength under repeated loading applications, cracks will form and propagate to the surface, this type of distress is called fatigue cracking which is the result of repeated bending of asphalt layer under traffic. ${ }^{(3)}$

Many factors contribute the forming of fatigue cracking, material properties and layer thickness, traffic load and number of load repetitions, temperature and environmental conditions. Studies
have shown that 3 to 5 inches thick asphalt is most susceptible to fatigue cracking damage. ${ }^{(3)}$ The proper thickness of asphalt layer must be either as thin as practical or as thick as possible. The flexible pavement with very thin asphalt layer tend to have less problem with fatigue cracking since compressive effect tend to be more significant.

### 2.1.4 Vertical deflection

Studies have shown that about 70 to $95 \%$ of the surface deflection is the function of the elastic compression on the subgrade layer. ${ }^{(3)}$ Based on mechanical theories about stress/strain relationship, in order to minimize surface deflections, vertical compressive stress should be kept low. From the above analysis about vertical compressive stress, to increase $E_{1}$ or $h_{1}$ at the surface layer when thickness of base course layer remains the same can all help to reach this goal.

Excessive vertical deflection on subgrade layer will cause rutting damage. Rutting is an incremental plastic damage throughout the layered system. National Cooperative Highway Research Program conducted by State of Illinois suggested that the total rutting in the pavement structure is to sum up the permanent deformation at asphalt layer, granular base layer and subgrade layer. Currently there are several models available for estimating plastic strain within asphalt layer such as Ohio State Model, ${ }^{(3)} \epsilon_{\mathrm{p}}=\mathrm{a}_{\mathrm{l}} \times\left(\epsilon_{\mathrm{r}}\right)^{\mathrm{a} 2} \times \mathrm{N}^{\mathrm{a} 3}$ and Illinois Model, ${ }^{(3)} \epsilon_{\mathrm{p}}=\mathrm{a}_{1} \times \epsilon_{\mathrm{r}}$ $\times \mathrm{T}^{\mathrm{a} 2} \times \mathrm{N}^{\mathrm{a} 3}$. In both models, plastic strain $\epsilon_{\mathrm{p}}$ is a function of $\epsilon_{\mathrm{r}}$, the resilient strain (also called recoverable strain) and N , number of repetitions of loading. In Illinois model, T , the temperature effect is also considered when estimating plastic strain in asphalt layer, since as a viscoelastic material, strain in asphalt is a time- and temperature-dependent value. ' $a_{n}$ ' in above equations represent calibration factors.

### 2.2 Stress analysis for rigid pavement

Stresses in rigid pavement are resulted from a variety of reasons: loading from the wheels, temperature or moisture changes, and deflection or deformation of the base course layer or subgrade layer.

Two factors will affect the decision on whether to use plain concrete or reinforced concrete pavement: (1) spacing of joints and (2) whether a base course is used over natural subgrade layer. In most cases, if the slab length is less than 20 feet and a cement-treated base course is used; plain concrete pavement would be a cost-saving choice. ${ }^{(2)}$ When the joint spacing increased to more than 40 feet, wire mesh reinforcement should be used for crack control purpose only. ${ }^{(2)}$

Fig. 8 as shown below illustrates the general layout of rigid concrete pavement. Dowel bars are used at transverse joints to act as a load-transferring media; dowel bars normally are heavy steels and will be placed at relative close intervals; they should also be smooth and lubricated at one end to allow slab movements.

Dowel bars for transverse joints (expansion or contraction)


Tie bars for longitudinal joints
Fig. 8 General layout of rigid concrete pavement ${ }^{(1)}$

Tie bars are used along longitudinal joints, the function of the tie bars is tying two slabs together to improve load transfer and prevent joints from opening and/or faulting. ${ }^{(8)}$ As recommended in the design manual of Illinois Department of Transportation, tie bar must be deformed and hooked and must be firmly anchored into the concrete; tie bars should be placed at relatively large intervals.

Basically there are four groups of joints for rigid concrete pavements: contraction joints, expansion joints, construction joints and warping joints. ${ }^{(2)}$ They are illustrated in Fig. 9 and 10.


Fig. 9 Contraction joint (a) and Expansion joint (b) $)^{(2)}$

(c) Construction Joint

(d) Warping Joint

Fig. 10 Construction joint (c) and Warping joint (d) ${ }^{(2)}$

Each kind of joint has its specific functions. The intention of applying contraction joints is to relieve the horizontal tensile stress due to contraction and curling/warping of concrete slab. Normal joint opening is about 0.25 inch. Dowel bar may or may not be used at the contraction joints location. ${ }^{(2)}$ When dowel bars are not used, load transfer can be achieved by grain interlock of the lower portion of the slab.

Expansion joints should allow concrete to expand, joint opening is about 0.75 inch. Since the gap is relatively big, infiltrations of subgrade materials into the expansion joints may cause the joints
to expand, and sometimes, inadequate load transfer may cause the pumping of material underneath the concrete slab, thus expansion joints are used more often for airport pavements.

Construction joints are also about 0.25 inch wide, the function is to transit from old concrete to the construction of new concrete. It can be keyed construction joints as shown in Fig. 10 or butt type construction joints which are more commonly used.

Warping joints are used to control longitudinal cracks to occur along the centerline of the pavement. Tie bars should be used to connect two pieces of slab, load transfer can be achieved by the grain interlock of the lower portion of the slab.

### 2.2.1 Curling stress

When the concrete pavement is exposed to the sun during the day, the top of the slab warms faster than the subgrade, if the temperature gradient occurs through the depth of the slab, curling stress will be induced.


Fig. 11 Curling effect due to temperature gradient ${ }^{(2)}$

Regardless the slab to curl upward or downward, the weight of the slab will always tend to hold the slab in its original position, thus the stresses are induced. Based on the work by Westergaard and Bradbury, curling stress at the edge and any point inside the slab can be calculated by the following equations. ${ }^{(1)}$

## For edge stresses:

$\sigma=\left(\mathrm{C} \times \mathrm{E} \times \alpha_{\mathrm{t}} \times \Delta \mathrm{t}\right) / 2$
For interior stresses:

$$
\begin{align*}
& \sigma_{y}=\left[\left(\mathrm{E} \times \alpha_{\mathrm{t}} \times \Delta \mathrm{t}\right) / 2\right] \times\left[\left(\mathrm{C}_{\mathrm{y}}+v \times \mathrm{C}_{\mathrm{x}}\right) /\left(1-v^{2}\right)\right]  \tag{Eq.2.2}\\
& \sigma_{\mathrm{x}}=\left[\left(\mathrm{E} \times \alpha_{\mathrm{f}} \times \Delta \mathrm{t}\right) / 2\right] \times\left[\left(\mathrm{C}_{\mathrm{x}}+v \times \mathrm{C}_{\mathrm{y}}\right) /\left(1-v^{2}\right)\right] \tag{Eq.2.3}
\end{align*}
$$

Within the above interior curling stresses equations, $C_{x}$ and $C_{y}$ are curling stress coefficients. They are fixed values based on the chart provided by Bradbury. ${ }^{(1)}$ The first equation measures interior curling stress due to bending in $y$ direction, and the second equation measures interior curling stress due to bending in $x$ direction; $E$ is the modulus of elasticity of concrete (lbs/sq. inch); $v$, the Poisson's ratio; $\alpha_{\mathrm{i}}$, coefficient of thermal expansion (inch $/ \mathrm{inch} /{ }^{\circ} \mathrm{F}$ ) and $\Delta \mathrm{t}$ is the temperature differential $\left({ }^{\circ} \mathrm{F}\right)$.

As a major distress in rigid pavement, transverse cracking is the combined effect of external load and temperature. When a heavy load is near the longitudinal edge of the slab, midway between the transverse joints, a severe tensile bending stress will occur at the bottom of the slab. The situation will become worse when the top of the slab is warmer than the bottom of the slab. Repeated heavy loading under this condition would result fatigue damage along the bottom edge
of the slab, which will induce a transverse crack and propagate to the surface of the pavement. When top of the slab is cooler than the bottom, repeated heavy loading would result in fatigue damage at the top of the slab, which eventually will induce transverse cracks initiated from the surface of the pavement. ${ }^{(3)}$

Many factors contribute to the forming of fatigue cracking, not only slab thickness and concrete strength, joint spacing also play a big role in the formation of transverse cracks. The effective way to minimize transverse cracking is to increase slab thickness, reduce joint spacing or use a widened slab, and if possible, to provide a stabilized base layer.

### 2.2.2 Frictional stress

From the above discussion, we understand that curling stresses would occur with temperature gradient through the depth of the concrete slab; while uniform temperature changes will cause the concrete slab to contract or expand, and friction stresses are resulted from frictional resistance in-between the slab and subgrade. Frictional resistance is critical for long slabs (normally refer to the slabs more than 100 feet long), ${ }^{(2)}$ while for short and average length slabs (less than 40 feet), ${ }^{(2)}$ frictional resistance value is not considered as significant.

The purpose of analyzing frictional stress is to determine the spacing between contraction joints and the number of tie bars. For pavement with long joint spacing, steel wire mesh must be used to take care of the tensile stress induced by friction.


Fig. 12 Distribution of frictional resistance ${ }^{(1)}$

According to the above illustration, Kelly has suggested an equation to measure the frictional stress: ${ }^{(1)}$
$\sigma_{\mathrm{c}}=\gamma_{\mathrm{c}} \times \mathrm{L} \times \mathrm{f}_{\mathrm{a}} / 2$
Where $\sigma_{\mathrm{c}}$ is the frictional stress ( $\mathrm{lbs} / \mathrm{sq}$. inch); $\gamma_{\mathrm{c}}$ is unit weight of concrete (lbs/cubic inch); L: the length of slab and $f_{a}$ is the average coefficient of friction between slab and subgrade layer, in most occasions $f_{a}=1.5$.

### 2.2.3 External loading stress

Westergaard had extraordinary contributions on the pavement thickness design. ${ }^{(2)} \mathrm{He}$ developed a set of equations for determining stresses and deflections in concrete pavements under different loads conditions: at the interior of the slab, at free edges and at the comers. These equations allow the prediction of critical stresses under extreme conditions of loading cases. ${ }^{(2)}$

## For corner case,

$\sigma_{\mathrm{c}}=\left(3 \times \mathrm{P} / \mathrm{h}^{2}\right) \times\left[1-\left(\mathrm{a}_{1} \times 1.414 / \ell\right)^{0.6}\right]$
For interior case,
$\sigma_{\mathrm{i}}=\left(0.316 \times \mathrm{P} / \mathrm{h}^{2}\right) \times\left[4 \times \log _{10}(\ell / \mathrm{b})+1.069\right], \mathrm{b}=\left(1.6 \times \mathrm{a}^{2}+\mathrm{h}^{2}\right)-0.675 \mathrm{~h}$
For edge case,
$\sigma_{e}=\left(0.572 \times P / h^{2}\right) \times\left[4 \times \log _{10}(\ell / b)+0.359\right], b=\left(1.6 \times \mathrm{a}^{2}+\mathrm{h}^{2}\right)-0.675 \mathrm{~h}$

Where P is the external load; $\ell$ is the radius of relative stiffness of concrete to subgrade, ${ }^{(1)} \ell=$ $\left[\left(\mathrm{E} \times \mathrm{h}^{3}\right) /\left(12 \times\left(1-v^{2}\right) \times \mathrm{k}\right)\right]^{0.25} \cdot \ell$ is a function of E , the modulus of elasticity of concrete; $v$, the Poisson's ratio of concrete; $k$ : the modulus of subgrade reaction and $h$ : the thickness of concrete slab; $a_{1}$ is called contact radius, it is the distance from the point where load applies to corner of slab; $a$ is the distance from the point where load applies to the point tangent to slab edge.

Basing on Westergaard theory, in 1950 's, Pickett and Ray developed a set of charts called 'influence charts' which allow analyzing stress and deflection more conveniently. ${ }^{(1)}$ The charts were developed basing on two assumptions: (1) pavement built on dense liquid foundation (2) pavement itself is an elastic solid. Liquid foundation is also called Winkler foundation, liquid foundation assumes under vertical force, vertical deflection at node $i$ is independent of deflections at neighboring nodes.

The procedure to calculate stress and deflection is relative simple by using influence chart. The first step is to find the size of tire imprint by the given tire pressure; second step is to trace the
tire imprints on the influence chart with using appropriate radius of relative stiffness as a scale factor and the third step is to count number of blocks within the imprint area and finally solve a set of equations to find deflection and stress.

### 2.2.4 Dowel bar stress

Upon above discussion, we understood that dowel bars are used at transverse joints to transfer loads when "grain interlock" is hardly achieved or heavy loads shall be applied, dowel joints are normally applied for joint openings from 0.04 to 0.25 inch wide.

The design of dowel bars are based on the analysis of concrete bearing stresses under dowel bars. Following assumptions should be applied in order to simplify the case: (1) it is assumed that dowel bars are perfectly aligned and free to move and (2) characteristics of subgrade materials are overlooked; dowel bar diameter and length design is based on pavement thickness only.


Fig. 13 Deflected shape of dowel bar under load ${ }^{(2)}$


Fig. 14 Dowel bar deformation under load ${ }^{(1)}$

Based on the above assumptions, Timoshenko, Friberg $(1940)^{(1)}$ developed following equations to estimate bearing stress on concrete underneath the dowel. For a joint opening $z$, if the dowel bar is subjected to external load $P_{t}$, the moment at the dowel-concrete interface is $M_{0}=-P_{t} \times z / 2$; the deflection of the dowel at the joint can be estimated by: ${ }^{(1)}$
$\mathrm{y}_{0}=\mathrm{P}_{\mathrm{t}} \times(2+\beta \times \mathrm{z}) /\left(4 \times \beta^{3} \times \mathrm{E}_{\mathrm{d}} \times \mathrm{I}_{\mathrm{d}}\right)$

Where $\beta$ is the relative stiffness of dowel bar versus concrete,
$\beta=\left[(\mathrm{K} \times \mathrm{d}) /\left(4 \times \mathrm{E}_{\mathrm{d}} \times \mathrm{I}_{\mathrm{d}}\right)\right]^{1 / 4}$

At above equation, where K is the modulus of dowel support, K value is also called steelconcrete K value, it expresses the bearing stress in the concrete developed under a unit deflection of dowel bar. ${ }^{(9)}$ Due to the difficulty of establishing the value of K theoretically, K value is normally selected from $3 \times 10^{5}$ to $1.5 \times 10^{6} \mathrm{pci}^{(1)}{ }^{( } \mathrm{d}$ ' is the diameter of dowel bar; and $\mathrm{E}_{\mathrm{d}}$, $\mathrm{I}_{\mathrm{d}}$ are modulus of elasticity of the dowel bar and the moment of inertia of the dowel bar respectively.

The actual bearing pressure on the concrete at the joint face $\sigma=K \times y_{0}$ has to be compared with allowable bearing stress of concrete since the concrete is much weaker than the steel. The allowable bearing stress of concrete can be estimated by $f_{b}=\left[(4-d) \times f_{c}{ }^{\prime}\right] / 3$ (American Concrete Institute, 1956), ${ }^{(1)} \mathrm{d}$ is the dowel diameter in inches and $\mathrm{f}_{\mathrm{c}}$ ' is the ultimate compressive strength of concrete. The load transferring capacity of a single dowel is determined by allowable bearing stress of concrete, the load transferring capacity equals to the actual bearing pressure $\sigma$ divided by allowable bearing stress $\mathrm{f}_{\mathrm{b}}$ and times $100{ }^{(2)}$ Load transferring capacity also depends upon the length of embedment. Tests indicated that for a 0.75 -inch diameter dowel, the length of embedment should be about 8 times of diameter. ${ }^{(2)}$ For bigger size dowels, the length of embedment decreases.

Since dowel bars work as a group, as per the study by Friberg (1940), ${ }^{(1)}$ when the load is applied at a joint, the dowel bars right under the applied load carry a major portion of load. Other dowel bars carry a progressively reduced amount of load, this result a portion of load can be transferred into the next slab.

The Fig. 15 below shows that the load carrying capacity of a group of dowels is equal to the sum of loads carried by each dowel. And in the case of superimposing effect of two wheels, similar as shown above, the load carried by the dowel at point A should be the load carried by the dowel at A due to load at A plus the load carried by the dowel at A due to load at B. From the point where load applies, load transferring capacity will reduce to zero at a distance equals to 1.8 times radius of relative stiffness of concrete versus subgrade.


Fig. 15 Loads distribution on dowel group ${ }^{(1)}$

Joint faulting is mainly caused by repeated heavy loading crossing transverse joints; it appears as base material build-up at one slab corner, and loss of support at adjacent corner. ${ }^{(3)}$ It has adverse impact on riding quality. For pavements with dowel bars as load transferring device, faulting is assumed mainly caused by erosion of the concrete around dowel bars under repeated traffic loading. Since the concrete bearing stress under dowel can be calculated, faulting can be predicted by calibrated equation. ${ }^{(3)}$ There are many ways to reduce the formation of joint faulting, either to increase the load transfer efficiency by increase the diameter of the dowel bars
or to reduce the joint spacing, or move the heavy load from slab corners by widening the slab or provide stabilized base course.

### 2.2.5 Reinforcing steel and tie bar stress

As per above mentioned, the design of dowel bars is conducted by analyzing the concrete bearing stress and actual load transferring capacity of dowel; while for reinforcing steel wire mesh and tie bar, the design is mainly based on the analysis of frictional stress induced by uniform temperature changes.

Wire mesh reinforcement will not add structural capacity to the pavement slab, it can tie cracked concrete pavement together and control crack developments in order to allow effective load transferring efficiency through grain interlock. Basing on the assumption that tensile stress along the slab length induced by the frictional stress is carried by wire mesh reinforcement, the area of steel can be approximately estimated.

Similar to the design of wire mesh reinforcement, tie bars are used along longitudinal joints to tie pavement slabs together, the approximate area of tie bars required for connecting longitudinal joints can be estimated by the same concept that tensile stress along the slab lane width induced by the frictional stress is carried by the tie bar steels.

The amount of steel required depends upon the size of the slab. For short slabs, steel could be reduced or omitted. When the pavement is designed without transverse joint, as it is called
continuously reinforced slab, adequate steels must be provided to ensure cracks to be tightly closed.

### 2.2.6 Thickness design criteria

From the above discussion we could see that stress-inducing factors in rigid concrete pavement are quite complex, while some factors could be ignored for thickness design. Besides subgrade type, curling stress will not be taken into consideration when deciding thickness of pavement. Herewith we applied the conclusive statement by E. J. Yoder: ${ }^{(2)}$ "Joints and steel are used to relieve and/or take care of warping (curling) stresses, and the design, then, is based upon load alone when considering thickness. This principle is so important that it must be clearly understood by the designer. Recall that a joint is nothing more than a designed crack".

Department of Transportation of individual states has developed their own minimum pavement thickness criteria. For instance, the Colorado Department of Transportation has stated in its Pavement Design Manual that the minimum thickness of Portland Cement concrete pavement is 8 inches for traffic greater than 1 million 18 K ESAL and 6 inches for traffic less than or equal to 1 million 18 K ESAL. ${ }^{(4)}$

### 2.3 Application of Elastic Layered Theory in flexible pavement design

Flexible pavements are layered systems with better materials on the top and being represented by a homogeneous mass is not realistic. Burmister's layered theory is more appropriate when we try to simulate the actual stress state of flexible pavements. ${ }^{(1)}$ Burmister first developed solutions for a two layered system which is more suitable for a full-depth asphalt construction, and then he
extended the solutions to a three-layered system. With the advent of computer software, the elastic layered theory can be applied to a multilayered system with any number of layers. The application of elastic layered theory on solving flexible pavement problem will be briefly introduced as below.

In order to use elastic layered theory to solve flexible pavement problem, a stress function $\Phi$ should be defined for each layer in the system. $\Phi$ is a function of $\mathrm{r}, \mathrm{z}, \mathrm{H}$ and vertical load, where $r$ is the radial distance from the study point to the external load; $z$ is the depth from the study point to the surface; H is the distance from the surface to the upper boundary of the lowest layer. $\Phi$ also contains integration constants; each layer has a stress function $\Phi$ with different integration constants.

Once the stress function $\Phi$ is defined, stress and strain within the layered system can all be expressed by the stress function $\Phi$; the stress function can also allow stress and strain to satisfy all elastic stress/strain relationships. By the classical theory of elasticity, stress function should satisfy the governing differential equation: $\nabla^{4} \Phi=0$, where $\nabla$ is called Laplace operator, $\nabla^{2}=$ $\partial^{2} / \partial r^{2}+(1 / r) \partial / \partial r+\partial^{2} / \partial z^{2}$.

To apply the continuity conditions basing on the assumption that all layers are fully bonded with the same vertical compressive stress, horizontal shear stress, vertical displacement and radial displacement at every point along the layer interface, the following relationship should be satisfied: $\left(\sigma_{z}\right)_{i}=\left(\sigma_{z}\right)_{i+1} ;\left(\tau_{\mathrm{rz}}\right)_{\mathrm{i}}=\left(\tau_{\mathrm{rz}}\right)_{\mathrm{i}+1} ;(\omega)_{\mathrm{i}}=(\omega)_{i+1} ;(\mathfrak{u})_{\mathrm{i}}=(\mathfrak{u})_{i+1}$. And if the ith layer interface
is unbonded or frictionless, the continuation of shear stress and radial displacement must be zero, $\left(\sigma_{\mathrm{z}}\right)_{\mathrm{i}}=\left(\sigma_{\mathrm{z}}\right)_{\mathrm{i}+1} ;\left(\tau_{\mathrm{rz}}\right)_{\mathrm{i}}=0 ;(\omega)_{\mathrm{i}}=(\omega)_{\mathrm{i}+1} ;\left(\tau_{\mathrm{rz}}\right)_{\mathrm{i}+1}=0$.

At the upper surface, as we have discussed before under Fig. 4 and Fig. 6, vertical compressive stress under the external load $\left(\sigma_{z}\right)_{1}=-P$, where $P$ is the vertical load applied on a unit area; horizontal shear stress $\left(\tau_{r 2}\right)_{1}=0$. With the application of these boundary and continuity conditions, unknown integration constants for each layer can be found, and stress and displacement can be determined.

Comparing with Boussinesq's elastic homogeneous half-space theory, Burmister's elastic layered theory is more practical for analyzing flexible pavement problems. Based on two-layered and three-layered system, various charts were developed for determining pavement responses conveniently. With the application of elastic layered theory in multilayered pavement system, available flexible pavement stress analysis software such as $K E N L A Y E R$ developed by Hung $(1985){ }^{(1)}$ and his colleagues makes multilayered pavement system stress analysis problems much easier to solve.

### 2.4 Application of Finite Element Method in rigid pavement design

Finite element method is a numerical technique for solving problems. By finite element method, a continuous physical problem will be transformed into a discretized finite element problem. The following paragraphs elaborate how finite element method can be applied in the case of concrete pavement design.

### 2.4.1 Discretize the continuum

For rigid concrete slabs, the shape of rectangular is selected to discretize the whole concrete slab into numerous finite elements.


Fig. 16 Discretize rectangular concrete slab ${ }^{(1)}$

Each nodal point $(i, j, k, l)$ at each rectangular element has three degrees of freedom $\left(\theta_{\mathrm{x}}, \theta_{\mathrm{y}}, \omega\right): \theta_{\mathrm{x}}$, the rotation about x axis; $\theta_{\mathrm{y}}$, the rotation about y axis and $\omega$ is the deflection along z axis. Nodal displacements for each rectangular element can be represented by the matrix as shown below which contains 12 items.
$\{\delta\}^{\mathrm{e}}=\left[\begin{array}{llll}\delta_{\mathrm{i}} & \delta_{\mathrm{j}} & \delta_{\mathrm{k}} & \delta_{\mathrm{l}}\end{array}\right]=\left[\omega_{\mathrm{i}} \theta_{\mathrm{xi}} \theta_{\mathrm{yi}} \omega_{\mathrm{j}} \theta_{\mathrm{xj}} \theta_{\mathrm{yj}} \omega_{1} \theta_{\mathrm{xl}} \theta_{\mathrm{yl}} \omega_{\mathrm{k}} \theta_{\mathrm{xk}} \theta_{\mathrm{yk}}\right]$

### 2.4.2 Select interpolation functions

A polynomial containing $x$ and $y$ coordinate will be chosen, the degree of the polynomial will be based on the number of nodes assigned to the element. Since each elements contains four nodes, a forth degree polynomial is assigned to represent vertical deflection $\omega$, since for slender slabs
under flexural bending, displacements, internal forces and stresses can all be expressed in term of $\omega\left(\theta_{\mathrm{x}}=-\mathrm{d} \omega / \mathrm{dy}\right.$ and $\left.\theta_{\mathrm{y}}=\mathrm{d} \omega / \mathrm{d} \mathrm{x}\right)$.
$\omega=a_{1}+a_{2} x+a_{3} y+a_{4} x^{2}+a_{5} x y+a_{6} y^{2}+a_{7} x^{3}+a_{8} x^{2} y+a_{9} x y^{2}+a_{10} y^{3}+a_{11} x^{3} y+a_{12} x y^{3}$

We randomly pick one element of size $a b y b$, if the center of the rectangular element is defined as $(0,0)$ of the xyz coordinate system, the coordinate reading of each node within the element can be found; and by inputting ( $\mathrm{x}, \mathrm{y}$ ) coordinate of each node, deflection $\omega$ at each node can all be expressed by above polynomial with $\mathrm{a}_{\mathrm{n}}$ as unknown coefficients. For instance, $(-\mathrm{a} / 2)$ and ( $\mathrm{b} / 2$ ) are the x and y coordinate for the nodal point i , deflection $\omega$, rotation $\theta_{\mathrm{x}}$ and $\theta_{\mathrm{y}}$ at point i can be expressed as three functions as shown below:
$\omega \mathrm{i}=\mathrm{a}_{1}+\mathrm{a}_{2} \times(-\mathrm{a} / 2)+\mathrm{a}_{3} \times(-\mathrm{b} / 2)+\mathrm{a}_{4} \times\left(\mathrm{a}^{2} / 4\right)+\mathrm{a}_{5} \times(\mathrm{ab} / 4)+\mathrm{a}_{6} \times\left(\mathrm{b}^{2} / 4\right)+\mathrm{a}_{7} \times\left(-\mathrm{a}^{3} / 8\right)+\mathrm{a}_{8} \times(-$ $\left.a^{2} b / 8\right)+a_{9} \times\left(-a b^{2} / 8\right)+a_{10} \times\left(-b^{3} / 8\right)+a_{11} \times\left(a^{3} b / 16\right)+a_{12} \times\left(a b^{3} / 16\right)$
$\left(\theta_{x}\right) i=-d \omega / d y=-\left[a_{3}-a_{5} \times(a / 2)-a_{6} \times b+a_{8} \times\left(a^{2} / 4\right)+a_{9} \times(a b / 2)+a_{10} \times\left(3 b^{2} / 4\right)-a_{11} \times\left(a^{3} / 8\right)-\right.$ $\left.\mathrm{a}_{12} \times\left(3 \mathrm{ab}^{2} / 8\right)\right]$
$\left(\theta_{y}\right) i=d \omega / d x=a_{2}-a_{4} \times a-a_{5} \times(b / 2)+a_{7} \times\left(3 a^{2} / 4\right)+a_{8} \times(a b / 2)+a_{9} \times\left(b^{2} / 4\right)-a_{11} \times\left(a^{2} b / 8\right)-a_{12}$ $\times\left(b^{3} / 8\right)$


Fig. 17 Nodal displacement ${ }^{(6)}$

And for the rest three nodes $j(-a / 2, b / 2), k(a / 2,-b / 2)$ and $1(a / 2, b / 2)$, similarly, there are three polynomials to represent $\left(\theta_{\mathrm{x}}, \theta_{\mathrm{y}}, \omega\right)$ for each node respectively. For such a rectangular element containing four nodes, there are totally 12 equations to represent the displacement matrix, in order to simplify, the displacement matrix can be expressed by $[\delta]=[\mathrm{C}][\mathrm{a}]$, where $[\mathrm{C}]$ is a 12 by 12 matrix containing a and $b$ (size of the element), and [a] contains $a_{1}$ to $a_{12}$ twelve unknown coefficients.

### 2.4.3 Define the element property

Stiffness matrix for the element shall be established at this step. Stiffness is the resistance of an elastic body to the deflection induced by an external load. Stiffness can be measured by $\mathrm{k}=\mathrm{P} / \delta$, where $P$ is the external force and $\delta$ represent the displacement. When both force and deflection are vectors, stiffness matrix can be expressed as a function of two major characteristics of material properties: E, the modulus of elasticity and $v$, the Poisson's ratio.

For elastic body, the following stress/strain relationships exist:

$$
\begin{align*}
& \sigma_{\mathrm{x}}=\mathrm{E} \times\left(\epsilon_{\mathrm{x}}+v \times \epsilon_{\mathrm{y}}\right) /\left(1-v^{2}\right)  \tag{Eq.2.15}\\
& \sigma_{\mathrm{y}}=\mathrm{E} \times\left(\epsilon_{\mathrm{y}}+v \times \epsilon_{\mathrm{x}}\right) /\left(1-v^{2}\right)  \tag{Eq.2.16}\\
& \tau_{\mathrm{xy}}=\mathrm{E} \times \gamma_{\mathrm{xy}} /[2 \times(1+v)] \tag{Eq.2.17}
\end{align*}
$$

And
$\epsilon_{\mathrm{x}}=\mathrm{du} / \mathrm{dx}=-\mathrm{z} \times \mathrm{d}^{2} \omega / \mathrm{dx}^{2}$
$\epsilon_{\mathrm{y}}=\mathrm{d} \nu / \mathrm{dy}=-\mathrm{z} \times \mathrm{d}^{2} \omega / \mathrm{dy}^{2}$
$\gamma_{x y}=d u / d x+d \nu / d y=-2 z \times d^{2} \omega / d x d y$

Where $\epsilon_{\mathrm{x}}$, elastic strain along x direction equals du/dx, the deformation per unit length of the object, du represents the size changes along $x$ direction and $d x$ represent the size of the object along x direction. $\mathrm{d} \nu$ and dy represent size changes and size of the object along y direction respectively.

### 2.4.3.1 Stiffness of concrete slab

For slender slabs subject to flexural bending as shown below in Fig. 18, bending moment can be expressed in the following form:
$\mathrm{M}_{\mathrm{x}}=\int_{-1 / 2, v 2)} \mathrm{z} \sigma_{\mathrm{x}} \mathrm{dz}$
$\mathrm{M}_{\mathrm{y}}=\int_{\mathrm{t}-12, \mathrm{t} 2 \mathrm{z})} \mathrm{z} \sigma_{\mathrm{y}} \mathrm{dz}$
$M_{x y}=\int_{-t 2, t 2)} \mathrm{z} \tau_{x y} \mathrm{dz}$


Fig. 18 Slender slab subjected to bending moment

Integrating the above moment equations, the moment matrix can be expressed as $[M]=[D]$ $[\psi],{ }^{(6)}$ where [D] is called modulus of rigidity of the slab, ${ }^{(1)}$ it is comprised of E, modulus of elasticity, $v$ Poisson's ratio and $t$, thickness of the slab.
$[\mathrm{D}]=\left[\left(\mathrm{E} \times \mathrm{t}^{3}\right) /\left(12 \times\left(1-v^{2}\right)\right)\right]\left[\begin{array}{lll}1 & v & 0 \\ v & 1 & 0 \\ 0 & 0 & (1-v) / 2\end{array}\right]$
$[\psi]=\left[\begin{array}{l}-d^{2} \omega / d x^{2} \\ -d^{2} \omega / d y^{2} \\ -d^{2} \omega / d x d y\end{array}\right]$

Element stiffness matrix can always be expressed in the following form: ${ }^{(6)}$
$\left.\left[K_{1}\right]\right]^{e}=\iint\left([B]^{T}[D][B] d x d y\right.$

Where, $[B]$ is called displacement differentiation matrix. It is obtained by differentiation of displacement expressed through shape functions and nodal displacements. For better understanding of the form of $[B],[B]=[Q][C]^{-1}$ and $[B]^{T}=[Q]^{T}[C]^{-1 T}$, as explained before, $[C]$ is a 12 by 12 matrix containing $a$ and $b$ (size of the element), [ Q ] is a 3 by 12 matrix, it equals inputting coordinate reading at one node into $[\psi]$. Thus $[\mathrm{K}]$ is a matrix containing only material properties, i.e. E and $v$.

In order to simulate real world situation, the stiffness of pavement should include stiffness of concrete slab, foundation and the joint: $[\mathrm{K}]^{\mathrm{e}}=\left[\mathrm{K}_{1}\right]^{\mathrm{e}}+\left[\mathrm{K}_{\mathrm{II}}\right]^{\mathrm{e}}+\left[\mathrm{K}_{1 I I}\right]^{\mathrm{e}(1)}$

### 2.4.3.2 Stiffness of foundation

Two different types of foundation can normally be considered: Winkler foundation and solid foundation. Winkler foundation was also called liquid foundation which assumes vertical force at node $i$ depends only on the vertical deflection at node $i$ and is independent of all other nodes.


Fig 19 Winkler foundation ${ }^{(1)}$
The stiffness of the Winkler foundation can be expressed as: $\mathrm{k}=\mathrm{P} / \omega$, where k is the modulus of subgrade reaction; P is the unit force and $\omega$ stands for the vertical deflection.

Assume an element of size $a$ by $b$, nodal points right locate at the corner of the slab, total resisting force from the foundation can be expressed as $F=\mathbf{a} \times \mathbf{b} \times \mathbf{k} \times \omega .{ }^{(1)}$ Since from the
assumption, every nodal point is independent of deflections from other nodal point, resisting force from each nodal point can be expressed as $F=a \times b \times k \times \omega / 4$. Transforming into matrix form $[\mathrm{F}]=[\mathrm{K}][\delta]$, we can easily get foundation stiffness matrix $\left[\mathrm{K}_{I I}{ }^{\mathrm{c}}=(\mathrm{k} \times \mathrm{a} \times \mathrm{b} / 4)\left[\begin{array}{lll}1 & 0 & 0\end{array}\right] . .^{(1)}\right.$

Winkler foundation oversimplified the actual characteristics of soil foundation. A solid foundation model is believed to be more realistic: the deflection at any nodal point depends not only on the force at the node itself but also on the forces at all other nodes. The following equation is used to determine the stiffness matrix: $\omega_{\mathrm{i}, \mathrm{j}}=\mathrm{P}_{\mathrm{j}}\left(1-v_{\mathrm{f}}\right) /\left(\pi \times \mathrm{E}_{\mathrm{f}} \times \mathrm{d}_{\mathrm{i}, \mathrm{j}}\right) .{ }^{(1)}$ Where $\omega_{\mathrm{i}, \mathrm{j}}$ is the deflection at nodal point i due to force at nodal point j ; $\mathrm{P}_{\mathrm{j}}$ is the force at nodal point j ; $v_{\mathrm{f}}$ is the Poisson's ratio for foundation and $\mathrm{d}_{\mathrm{i}, \mathrm{j}}$ is distance between nodal point i and j . The stiffness matrix of the foundation element is defined as deflection at one node due to forces at all other nodes. In order to determine the stiffness matrix, a unit concentrated force can be transferred as distributed load over a " $4 \times \mathrm{a} \times \mathrm{b}$ " area as a uniform pressure $1 /(4 \times \mathrm{a} \times \mathrm{b})$, then integrate over the whole area.



Fig. 20 Solid foundation ${ }^{(1)}$

### 2.4.3.3 Stiffness of joints

The stiffness of joint is represented by shear spring constant $C_{\omega}$, which can be expressed by: $\mathrm{C}_{\omega}=$ shear force per unit length of joint / difference in deflection between two slabs.


Fig. 21 Shear transfers through joint by grain interlock ${ }^{(1)}$

When we use grain interlock to transfer the load over joint as shown above in Fig. 21, the shear force can be expressed as $\mathrm{F}=\mathrm{L} \times \mathrm{C}_{\omega} \times \omega_{\mathrm{d}}$, where F is the nodal force applied to both slabs; L is the average spacing between nodes at the joint and $\omega_{d}$ is the deflection between two slabs.

When dowel bars are used to transfer loads, the situation can be simulated by Fig. 22, and the deflection between two slab can be expressed as $\omega_{d}=\Delta S+2 \times y_{0}$. Where $\Delta S$ is the deformation of dowel under load and $y_{o}$ is deformation of concrete under the dowel. Both items can be expressed as a function of $P_{t}$ which is the force applied on each dowel bar and equals to $F \times S_{b} / L$, where $F$ is nodal force applied at the slab, $S_{b}$ is dowel spacing and $L$ is the average nodal spacing at joint. By knowing $\omega_{d}, C_{\omega}$, shear spring constant at the joint can be defined.


Fig. 22 Shear transfers through joint by dowel bar ${ }^{(1)}$

### 2.4.4 Assemble the element equations

The core equation of the finite element method is $[F]=[K][\delta],{ }^{(6)}$ where $[F]$ is the force vector; $[\mathrm{K}]$ is the stiffness matrix and $[\mathrm{K}]=\left[\mathrm{K}_{1}\right]+\left[\mathrm{K}_{\mathrm{II}}\right]+\left[\mathrm{K}_{\mathrm{III}}\right] ;[\delta]$ is the displacement vector. $[\mathrm{F}]$ could contain external force and thermal stress due to temperature changes. Since $[\mathrm{F}]$ and $[\mathrm{K}]$ contain all known information, [ $\delta]$, the displacement vector can be found by solving a group of equations.

In order to keep the consistency, external load $P$ should be converted into 12 nodal loads (at 3 directions at each nodal point and totally 4 nodal points for one element).

External load vector can be expressed by the following form:

$$
\{F\}^{e}=\left[\begin{array}{lllllll}
Z_{i} & T x_{i} & T_{y i} & Z_{j} & T x_{j} & T_{y j} & Z_{l} \tag{Eq.2.27}
\end{array} \mathrm{Tx}_{1} \mathrm{~T}_{\mathrm{yl}} Z_{k} \quad \mathrm{Tx}_{\mathrm{k}} \mathrm{~T}_{\mathrm{yk}}\right]
$$

From the Fig. 23 below we can see that, since the slab only subject to vertical loading, the ' $T$ ' item at the above matrix should be equal zero.


Fig. 23 Finite Element under external load ${ }^{(6)}$

If the curling effect considered, the above matrix can be expressed by another format:
$[\mathrm{F}]=\left[\mathrm{K}_{\mathrm{p}}\right][\delta]-\left[\mathrm{K}_{\mathrm{f}}\right]\left[\delta^{\prime}\right] .{ }^{(1)}$ Where $\left[\mathrm{K}_{\mathrm{p}}\right]$ is the stiffness matrix of the slab including the joint; $[\delta]$ is the nodal displacement of the slab; $\left[\mathrm{K}_{f}\right]$ is the stiffness matrix of the foundation; $[\delta$ '] is the nodal displacement of the foundation. The items in the [ $\delta$ '] should equal to the numbers in [ $\delta$ ] subtracted by the curling deformation, and the rotation items should equal zero in [ $\delta^{\prime}$ ], since foundation will have no rotations.

### 2.4.5 Solve the global equation system

Once the element equation $[F]^{e}=[K]^{e}[\delta]^{e(6)}$ be set up, the global equation system will be established in the similar format: $[\mathrm{F}]^{\mathrm{T}}=[\mathrm{K}]^{\mathrm{T}}[\delta]^{\mathrm{T}(6)}$ basing on the element equation. The global stiffness matrix and the global load vector can normally be expressed in the following form: ${ }^{(6)}$
$[\mathrm{K}]^{\mathrm{T}}=[\mathrm{A}]^{\mathrm{T}}[\mathrm{K}]^{\mathrm{e}}[\mathrm{A}]$
$[\mathrm{F}]^{\mathrm{T}}=[\mathrm{A}]^{\mathrm{T}}[\mathrm{F}]^{\mathrm{e}}$
Where $[A]^{T}$ and $[A]$ are the matrix providing transformation from local to global system, they contain all known numbers.

### 2.4.6 Compute additional results

By solving the equation system $[\mathrm{F}]=[\mathrm{K}][\delta]$, nodal displacements can be found. By following the similar procedure as mentioned above, internal force matrix [f] can also be found. [f] can be expressed as $[\mathrm{f}]=[\mathrm{D}][\mathrm{A}]^{(6)}$, where [D] is a known member within stiffness matrix and $[\mathrm{A}]$ is a transformation matrix containing all known numbers. By solving [f], basing on the relationship $[\mathrm{M}]=[\mathrm{f}][\delta],[\mathrm{M}]$, the moments can be determined. And finally stress and strain can be solved.

## Chapter 3: Comparison of KENSLABS and STAAD-III

### 3.1 Background on Finite element software

Various computer software has been developed with the application of Finite Element Method; the high-speed computation of PC makes the prediction of pavement distress and stress analysis much more convenient today than the past. EverFE delivered by Washington State Department of Transportation and University of Washington (1997); ${ }^{(5)}$ KENSLABS developed by Huang (1985) are both finite element program to investigate concrete pavement performance and rehabilitation alternatives. SAP2000 and STAAD are both the most commonly used finite element program for structural engineers to simulate plane and space structures responses.

KENSLABS developed by Huang (1985) and his colleagues is a concrete pavement stress analysis program basing on finite element method. The program allows the study of a maximum of 6 slabs, 7 joints and 420 nodes, and each slab can have a maximum of 15 nodes in $x$ direction and 15 nodes in y direction. ${ }^{(1)}$ The following example will elaborate how a combined warping and loading stress analysis case be solved by using KENSLABS program.

### 3.2 Design example 1 - the study on combined effect of warping and loading

A concrete slab, 10 meter ( 32.8 feet) long, 3 meter ( 9.8 feet) wide and 20 centimeter ( 7.9 inch ) thick, is placed on a modulus of subgrade reaction of $55 \mathrm{MN} / \mathrm{m}^{3}$ ( 202.7 pci ). The pavement is subjected to a temperature differential of $0.5^{\circ} \mathrm{C}$ per centimeter $\left(2.28^{\circ} \mathrm{F}\right.$ per inch) at night when a $45 \mathrm{KN}(10100 \mathrm{lbs})$ single axle load is applied on the edge of the slab over a circular area with a
contact pressure of $600 \mathrm{KPa}(87 \mathrm{psi})$ as shown below in Fig. 24. Determine the combined pressure due to curling and loading at the edge beneath the load.


Determine $\sigma_{c}$ due to combined effect of warping and loading

Fig. 24 Pavement design example 1

### 3.2.1 Analysis using $K E N S L A B S$

In order to save computer storage and running time of computation, since the slab and loading condition is symmetrical about Y -axis, only half of the slab needs to be discretized into rectangular finite elements. We usually number the nodes by 'from bottom to top' and 'from left to right' sequence. The general rule of numbering the nodes is trying to keep the maximum difference between nodal numbers on the opposite corners of the element low. The following Fig. 25 and Fig. 26 show how half of the slab was discretized and nodes were numbered. Half slab has been discretized into $5 \times 7=35$ elements, and half slab structure contains $6 \times 8=48$ nodes.


Fig. 25 Discretize half slab for design example 1 (KENSLABS)


Fig. 26 Numbering of nodes for design example 1 (KENSLABS)

KENSLABS can analyze the pavements under the combined effects of temperature loading and physical loading. Since the pavement is subjected to a temperature differential $0.5^{\circ} \mathrm{C}$ per centimeter $\left(2.28^{\circ} \mathrm{F}\right.$ per inch) throughout the thickness of the slab, $\Delta \mathrm{t}$, the temperature differential between the top and the bottom of the slab is $0.5 \times 20=10^{\circ} \mathrm{C}\left(18^{\circ} \mathrm{F}\right)$. With knowing 45 KN $(10100 \mathrm{lbs})$ load is applied uniformly on the circular area with a contact pressure 600 KPa ( 87
psi ), we need to convert the actual circular tire contact area to equivalent rectangular area (as shown in Fig. 27 below) which equals to $45 \times 10^{4} / 600=750 \mathrm{~cm}^{2}(116$ sq. inch). On half of the slab, 22.5 KN ( 5050 lbs ) load is uniformly applied on a approximate $11.5 \mathrm{~cm} \times 33 \mathrm{~cm}$ ( 4.5 inch $\times 13$ inch) rectangular area. When discretize the slab into finite elements, for concentrated loads, we always try to locate the external loads right on or close to the nodes. For uniformly distributed loads, we always try to locate the load boundaries right on or close to the boundaries of elements.


Actual area $=0.5227 L^{2}$, when actual area $=750 \mathrm{~cm}^{2}(116$ sq. inch $), L \approx 38 \mathrm{~cm}$ ( 15 inch)


Equivalent area $=0.6 L \times 0.8712 L \approx 23 \mathrm{~cm} \times 33 \mathrm{~cm}$ (9 inch $\times 13$ inch $)$
Fig. 27 Conversion of actual tire contact area into equivalent rectangular area ${ }^{(1)}$

Other necessary input information is: coefficient of thermal expansion of concrete slab is $9 \times 10^{-6}$ $\mathrm{mm} / \mathrm{mm} /{ }^{\circ} \mathrm{C}\left(5 \times 10^{-6} \mathrm{in} / \mathrm{in} /{ }^{\circ} \mathrm{F}\right) ; v$, Poisson's ratio of concrete is 0.15 ; E , the modulus of elasticity of concrete is $27.6 \times 10^{6} \mathrm{KPa}\left(4 \times 10^{6} \mathrm{psi}\right)$. The following Fig. 28 shows the stress contour
generated by $K E N S L A B S$ for the studied slab under combined effect of temperature loading and uniform loading. Maximum tensile stress at the bottom of the slab occurs at node ' 1 ', which is the corner of the slab where the uniform load applies; the maximum compressive stress at the bottom of slab occurs along the edge of the slab due to the warping effect.
example 1 - liquid foundation

Fig. 28 Stress contour due to combined effect of warping and loading (liquid foundation)
Under the combined effect of warping and loading, stress level at node ' 1 ' at the bottom of the slab is 1346 KPa as a tensile stress. Combined stresses at other locations can refer to Appendix A as attached with this report.


Fig. 29 Stresses at node ' 1 'due to combined effects of warping and loading

As shown above in Fig. 29, 'Stress $x$ ' represents the stress in $x$ direction, negative when bottom of the slab is in tension; 'Stress $y$ ' is the stress in $y$ direction, negative when bottom of the slab is in tension; 'Stress $x y$ ' represents shear stress in $x y$ plane. 'Major' and 'Minor' are major and minor principle stresses in slab when shear stress equals zero, which is used to compare with flexural stress of concrete, and 'Max Shear' represents maximum shear stress in slab. At node ' 1 ' 'Stress y' equals zero, since node ' 1 ' locates at the edge of slab.

Warping effect is a significant factor when designing concrete pavement joints; for the same concrete slab if without the warping effect, maximum tensile stress occurs at node ' 1 ' at the bottom of the slab is 2608 KPa as a tensile stress as shown in Fig. 30 and Fig. 31.
example 1 - without warping effect/liquid foundation

$\square$ Inderam Latad


Fig. 30 Stress contour due to uniform load alone (liquid foundation)


Fig. 31 Stresses at node '1' due to uniform load

The result is quite reasonable: as shown in Fig. 32 and Fig. 33 below, the studied concrete slab is experiencing a temperature differential of $0.5^{\circ} \mathrm{C}$ per centimeter $\left(2.28^{\circ} \mathrm{F}\right.$ per inch $)$ at night; the top of the concrete slab is cooler than bottom of the slab. The top tends to contract and the bottom tends to expand, however, the weight of the slab restrains it from expansion and contraction, tensile stress will be induced at the top and compressive stress at the bottom. This pair of the stress can balance part of the stress induced by uniform loading at the edge which is in opposite direction, thus compare with 2608 KPa when uniform load alone applied on the slab, the tensile stress at node ' 1 ' become 1346 KPa when warping effect is also considered.


Fig. 32 Combined effect of warping and loading at node ' 1 '


Fig. 33 Upward curling ${ }^{(l)}$

Above calculation was based on the assumption that concrete slab was built on liquid foundation; since $K E N S L A B$ can offer three different types of foundation for analysis: liquid, soild or layer; even though liquid foundation is a simple approach as it requires less computer time and storage, solid foundation is still a more realistic solution to treat concrete pavement problems.

In stead of providing modulus of subgrade reaction $k$ for liquid foundation, resilient modulus $M_{R}$ and Poisson's ratio $v$ of subgrade should be provided for solid foundation. Resilient modulus $\mathbf{M}_{\mathbf{R}}$ is the elastic modulus based on the recoverable strain under repeated loads, the value of $M_{R}$ for granular material and fine-grained soil is normally determined by repeated triaxial test. ${ }^{(1)} \mathrm{A}$ calibrated equation is recommended to determine $M_{R}$ with knowing subgrade reaction $k,{ }^{(1)} M_{R}=$ $18.8 \times \mathrm{k}$, where k is in pei and $\mathrm{M}_{\mathrm{R}}$ in psi. In this design example, $\mathrm{M}_{\mathrm{R}}=18.8 \times 202.7=3811 \mathrm{psi}$ ( 26276 KPa ), Poission's ratio of subgrade soil $v$ is assumed to be 0.45 .

Solid foundation is able to provide more realistic result since subgrade reaction k used in liquid foundation is not a true characteristic of soil behaviors. ${ }^{(1)}$ Comparing with calculated tensile
stress at node ' 1 ' on liquid foundation under the combined effect of warping and loading as 1346 KPa , stress level at node ' 1 ' become 1509 KPa on solid foundation as shown in Fig. 34.
example 1 - solid foundation
( Max. '-' Stress in X Dir. $=-1509.3$ * Mas. ' + ' Stress in $X$ Dir. $=1628.4 \mathrm{kPa}$


Fig. 34 Stress contour due to combined effect of warping and loading (solid foundation)

Another important feature of $K E N S L A B S$ is to evaluate the contact condition between concrete pavement and subgrade foundation. The above design example 1 was solved basing on the assumption that slab and foundation are always in full contact. Under the full contact condition, the precompression due to the weight of the slab is more than the deflection due to temperature curling effect; the spring supports of subgrade are always able to contact the slab within their elastic range.


Fig. 35 Negative reactive pressure at nodes

According to the report generated by $K E N S L A B S$ as attached Appendix A (liquid foundation), as shown in Fig. 35 above, support reactive pressures at highlighted nodes are negative in sign (compression positive). That means under temperature curling effect, the slab at these nodes will curl up, since the slab and subgrade are always in full contact within the elastic range of spring support of subgrade, the springs of subgrade will pull the slab back into position, tensile stresses are induced at these nodes as support reactions.

Besides the condition of full contact, the slab and foundation can be in partial contact when the slab is subjected to curling or pumping before any load applications. Under the partial contact condition, initial gaps between slab and subgrade may or may not exist depending on whether there is pumping or plastic deformation of subgrade induced by repeated heavy traffic. Under a high intensity of traffic, some supporting spring of subgrade will fail to function elastically; gaps will form at these locations.

KENSLABS is able to evaluate the contact condition for partial contact with applied load for both liquid and solid foundations, and a two-step analysis is recommended when analyzing such cases. Most importantly, in the case of partial contact, the weight of the slab must be considered in order to counterbalance the positive reactive forces.

In the case of liquid foundation, firstly, the gaps and precompression due to temperature curling, weight of the slab and initial gaps are pre-determined (gaps are positive in sign and precompression is negative in sign); secondly, using the gaps and precompression obtained from the first step to calculate the stresses and displacements under the applied load. In the case of solid foundation, the contact condition is determined by the reactive forces and precompression due to temperature curling, weight of the slab and initial gaps. Compressive reactive forces are positive in sign, which means slab and subgrade are in contact. Negative (tensile) reactive force means slab and subgrade are not in contact. All tensile reactive forces are assigned to zero and the program will automatically run iteration cycles until there is no negative (tensile) reactive force of subgrade. The procedure of two-step analysis will be briefly introduced in Appendix A.

### 3.2.2 Analysis using STAAD-III

$S T A A D-I I I$, developed by REI, Research Engineers Inc., is another powerful finite element program for analyzing linear and nonlinear, static and dynamic three-dimensional concrete, aluminum, timber and steel structures. The above stress analysis example will be solved once again by STAAD-III program.

STAAD-III grouped all types of structures into four categories: ${ }^{(7)}$ a $S P A C E$ structure is a threedimensional framed structure with loads applied on any plane; a PLANE structure is a twodimensional structure bound by a global X-Y coordinate system with loads in the same plane; a $T R U S S$ structure is a structure consists of truss members which can have only axial forces and no bending in members; a FLOOR structure is a two-dimensional or three-dimensional structure with no horizontal movement. In this case, we define the concrete pavement slab as a $S P A C E$ structure.

In order to specify the structure geometry, STAAD-III uses Cartesian coordinate system as global coordinate system to define joint locations and loading directions. The Cartesian coordinate system as shown in Fig. 36 follows the orthogonal right hand rule; the translational degrees of freedom are denoted by $u_{1}, u_{2}, u_{3}$ and the rotational degrees of freedom are denoted by $u_{4}, u_{5}$ and $\mathbf{u}_{6}$.


Fig. 36 Cartesian coordinate system ${ }^{(7)}$


Fig. 37 Association of global and local coordinate system ${ }^{\text {(7) }}$

The local coordinate system also follows the right hand rule, but it associates with each member. As shown in Fig. 37 above, the beam member with starting node " i " and end node " j ", the positive direction of local X axis is joining node " i " to " j " and projecting in the same direction. All element force output is in the local coordinate system.

Similar to other finite element software, in order to save computing time, similar elements should be numbered sequentially in STAAD-III; and when assigning nodes to elements, nodes should be specified either clockwise or counter clockwise. The program also automatically generate a node at element center; after running the program, element force output is available at the center of the element.
$S T A A D-I I I$ builds the finite element model through text input file. The text file contains a series of commands such as 'joint coordinates' and 'repeat', 'joint coordinate' command specifies joint
coordinates for specific elements in the global Cartesian coordinate system; 'repeat' command is used to define joint locations by same size increments. The following Fig. 38 and Fig. 39 show how the joint coordinates are specified and elements are numbered for design example 1.


Fig. 38 Discretize the slab for design example 1 (STAAD-III)


Fig. 39 Numbering the nodes for design example 1 (STAAD-III)

The whole slab has been discretized into $10 \times 22=220$ elements; and the whole structure contains $11 \times 23=253$ nodes as shown above. The $45 \mathrm{KN}(10100 \mathrm{lbs})$ uniformly distributed load is applied on a $23 \mathrm{~cm} \times 33 \mathrm{~cm} \approx 750 \mathrm{~cm}^{2}\left(116 \mathrm{inch}^{2}\right)$ area where element 101,102 and element 111, 112 are located. Similar as $K E N S L A B S$, when discretize the slab into finite elements, for concentrated loads, we always try to locate the external loads right on or close to the nodes; and for uniform distributed loads, we always try to locate the boundary of area load right on boundary of the elements.

STAAD-III can specify temperature load on members and elements by applying 'temp load' command. Within the 'temp load' command line, both $f_{1}$ and $f_{2}$ should be specified ${ }^{(7)}$; where $f_{1}$ is the change in temperature which will cause axial elongation in the members or uniform expansion in element; $f_{2}$ is the temperature differential from the top to the bottom of the member or element ( $T_{\text {top }}-T_{\text {botom }}$ ), it is the $f_{2}$ that cause the member to bend. In this case, $f_{1}=0$ and $f_{2}=-$ $10^{\circ} \mathrm{C}\left(18^{\circ} \mathrm{F}\right)$.

For analysis purpose, STAAD-III can separately calculate the stress under different loading cases with the using of 'load combination' command. In this case, four group of load cases are specified: load case 1 : uniformly distributed external load alone; load case 2 : temperature load alone (upward curling); load case 3: uniformly distributed load with upward curling ( $\mathrm{T}_{\text {top }}$ $\left.\mathrm{T}_{\text {botom }}=-10^{\circ} \mathrm{C}\right)$; load case 4: uniformly distributed load with downward curling $\left(\mathrm{T}_{\text {top }}-\mathrm{T}_{\text {bottom }}=\right.$ $10^{\circ} \mathrm{C}$ ).

The other necessary information to be provided is modulus of elasticity of concrete slab $\mathrm{E}=27.6$ $\times 10^{6} \mathrm{KPa}\left(4 \times 10^{6} \mathrm{psi}\right)$ and Poisson's ratio $v=0.15$. STAAD-III is also capable of modeling elastic spring support for subgrade foundation by using 'support' and 'elastic mat' command, in this case, subgrade reaction $\mathrm{k}=55 \mathrm{MN} / \mathrm{m}^{3}$ ( 202.7 pci ), this means the analysis is basing on the assumption that the slab is built on liquid foundation.

After running the STAAD-III input file, a report containing joint displacements, support reactions, member end forces, element principle stress, shear force $Q_{x}, Q_{y}$, membrane force $F_{x}$,
$F_{y}, F_{x y}, M_{x}, M_{y}$ and $M_{x y}$ will be printed. The positive directions of element forces are as shown below in Fig. 40.


Fig. 40 Sign convention of element forces ${ }^{(7)}$

Refer to attached Appendix B for detailed report of results; the critical stresses at the centre of element '101' under uniformly distributed load are listed as below.

Element forces at centre of element '101':
Load case 1 (uniformly distributed load alone):
Bottom of slab: 2229 KPa (tension)
Load case 2 (upward curling):
Bottom of slab: -1246 KPa (compression)
Load case 3 (uniformly distributed load with upward curling):
Bottom of slab: 988 KPa (tension)
Load case 4 (uniformed distributed load with downward curling):
Bottom of slab: 3474 KPa (tension)

Comparing the result from $K E N S L A B S$ and $S T A A D-I I I$, based on the assumption that the slab is built on liquid foundation, when only external load applies, 2608 KPa at node ' 1 ' was calculated by KENSLABS and 2229 KPa at the center of element '101' was calculated by STAAD-III. When considering external load combined with upward curling effect, 1346 KPa at node ' 1 ' was calculated by $K E N S L A B S$ and 988 KPa at the center of element ' 101 ' was calculated by $S T A A D$ III.

Many factors may affect the precision of the calculation. First of all, the size of finite elements can be one of the important factors. Normally the coarser the finite element mesh, the smaller stresses and deflections we can get, thus the coarser mesh may lead to unsafe design. Secondly, KENSLABS program provides the stresses at nodes, while STAAD-III provides the stresses at the center of the elements.

As a finite element program designed specifically for pavement stress analysis, KENSLABS has several advantages in term of simplicity. For KENSLABS, When the slab and loading exhibit symmetry, only one-half or one-quarter of the slab need to be considered. While for STAAD-III, the finite element program for general plane and space structures, the program was not tailored to recognize 'free edge', 'joint' or 'centerline' of the slab like KENSLABS. Especially for the design example 1 , the slab is subjected to combined effect of edge loading at center of slab and temperature loading, the whole pavement slab need to be studied since center portion of slab experience critical stresses when the whole slab curling upward or downward. Besides, KENSLABS is able to provide straightforward answers as far as we are concerned in the pavement design problems, for instance, stress at the comer of the slab. Furthermore,

KENSLABS has three foundation options to choose, and solid foundation is believed to be the solution close to real conditions. The ability to evaluate contact condition between slab and subgrade is another advantage of $K E N S L A B S$ program. In reality, the slab and subgrade are not always in full contact due to temperature curling or pumping. For old pavements under high intensity of traffic, plastic deformation will form at some locations of subgrade at where initial gaps exist. KENSLABS is able to determine the contact condition for partial contact cases with applied load; the stresses and displacements of slab can be analyzed basing on pre-determined gaps and precompression.

### 3.3 Design example 2 - the study on dowel bars at transverse joint

We will now use both $K E N S L A B S$ and $S T A A D-I I I$ to solve another stress analysis example dealing with dowel bars at transverse joint as load transferring devices.


Fig. 41 Pavement design example 2

As shown in Fig. 41 above, a concrete slab has a width of 3.6 meter ( 141.7 inch) wide, a thickness of 20 centimeter ( 7.9 inches) and a modulus of subgrade reaction of $80 \mathrm{MN} / \mathrm{m}^{3}$ ( 294.9 pci). Two 30 centimeter ( 11.8 inch) by 30 centimeter ( 11.8 inch) loaded area, each weighing 60 $\mathrm{KN}(13470 \mathrm{lbs})$ and spaced at 210 centimeter ( 82.7 inch ) apart, are applied at the joint with the outside loaded area adjacent to the pavement edge as shown below. Determine the maximum bearing stress between concrete and dowel. The joint has an opening of 0.5 centimeter ( 0.2 inch) and the dowels are 2.5 centimeter ( 0.98 inch) in diameter and 30 centimeter ( 11.8 inch) on centers.

### 3.3.1 Analysis using STAAD-III



Fig. 42 Discretize the slab for design example 2 (STAAD-III)


Fig. 43 Numbering of nodes for design example 2 (STAAD-III)

Firstly, we use STAAD-III program to solve the problem. In this case, 3 m (118 inches) of the pavement slab on both sides of the transverse joint is studied. 30 cm ( 11.8 inch) by 30 cm ( 11.8 inch) square elements are defined; there are totally $12 \times 20=240$ finite elements. Fig. 42 and Fig. 43 above showed how the joint coordinates are specified and elements are numbered.

In order to properly simulate the dowel bars, we have to fully understand how dowel bars function at transverse joint. Dowel bars are fixed at one end and free to move at other end, they are used across transverse joints to transfer only vertical loads. Thus dowel bars can be simulated as a group of steel beams members fixed at one end and hinged at other end between two concrete slabs. From the given information, 2.5 cm ( 1 inch ) diameter dowel bars are arranged at

30 cm ( 11.8 inch ) center to center, there are totally 13 steel beam members are defined, for instance, the steel beam member connecting node 131 and node 144 represent one dowel bar.

Since dowel bars do not transfer any moment or horizontal movement, STAAD-III allows specification of release of degrees of freedom for specific members by using "release" command. In this case, since dowel bars transfer only vertical load along $Y$-axis, member force $F_{x}, F_{z}$ and moment $\mathrm{M}_{\mathrm{x}}, \mathrm{M}_{\mathrm{y}}$ and $\mathrm{M}_{\mathrm{z}}$ has to be released.


Fig. 44 Simulate dowel bars as steel beam members (STAAD-III)


Element 109
Fig. 45 Uniformly distributed load on element 109 \& 116 (STAAD-III)

Finally, $60 \mathrm{KN}(13470 \mathrm{lbs})$ uniformly distributed loads are applied on two 30 cm (11.8 inch) by 30 cm (11.8 inch) areas, which are 210 cm ( 82.7 inch) apart, STAAD-III can specify uniformly distributed load applied directly on elements by the 'pressure' command. As show in Fig. 45 below, area loads are applied on element 109 and 116 , uniform pressure equals to $60 \times 10^{4} / 900$ $\mathrm{cm}^{2}=667 \mathrm{KPa}(96.7 \mathrm{psi})$.

Other necessary information shall be provided including modulus of elasticity of concrete slab E $=27.6 \times 10^{6} \mathrm{KPa}\left(4 \times 10^{6} \mathrm{psi}\right)$; Poisson's ratio $v=0.15$; modulus of elasticity of steel dowel beam $E_{d}=2 \times 10^{8} \mathrm{KPa}\left(29 \times 10^{6} \mathrm{psi}\right)$; cross sectional area of steel bar $\mathrm{A}=\pi \times \mathrm{d}^{2} / 4=4.9 \mathrm{~cm}^{2}$
( 0.76 square inch); moment of inertia of steel dowel beam $I_{y}=I_{z}=\pi \times d^{4} / 64=1.92 \mathrm{~cm}^{4}$ ( 0.045 inch ${ }^{4}$ ) and torsional constant $\mathrm{I}_{\mathrm{x}}=\pi \times \mathrm{r}^{4} / 2=3.83 \mathrm{~cm}^{4}\left(0.09\right.$ inch $\left.^{4}\right) . S T A A D$-III is also capable of modeling elastic spring support for subgrade foundation by using 'support' and 'elastic mat' command, in this case, subgrade reaction $k=80 \mathrm{MN} / \mathrm{m}^{3}$ (294.9 pci). This means the analysis is based on the assumption that the slab is built on liquid foundation.

The steel member stress are listed as below, a detail report of results is attached in Appendix C. Maximum shear force occur at steel members under area loads are as shown below.


Fig. 46 Maximum stress and deflection at critical locations

## Member end forces:

Member 301, shear- $\mathrm{Y}=-13.11 \mathrm{KN}(-2937$ pounds $)$
Member 302, shear- $\mathrm{Y}=-11.88 \mathrm{KN}(-2661$ pounds $)$
Member 308, shear- $\mathrm{Y}=-10.78 \mathrm{KN}(-2415$ pounds $)$
Member 309, shear- $Y=-10.82 \mathrm{KN}(-2424$ pounds $)$

The relative stiffness of dowel bars embedded in concrete, $\beta$, can be defined by $\beta=[(\mathrm{K} \times \mathrm{d}) /(4$ $\left.\left.\times \mathrm{E}_{\mathrm{d}} \times \mathrm{I}_{\mathrm{d}}\right)\right]^{1 / 4} \cdot{ }^{(1)}$ Where K is the modulus of dowel support, in this case we assume $\mathrm{K}=406950$ $\mathrm{KN} / \mathrm{m}^{3}{ }^{(1)}\left(1.5 \times 10^{6} \mathrm{pci}\right)$; the modulus of elasticity of steel dowel $\mathrm{E}_{\mathrm{d}}=2 \times 10^{8} \mathrm{KPa}\left(29 \times 10^{6}\right.$ psi); moment of inertia of dowel bar $\mathrm{I}_{\mathrm{d}}=\pi \times \mathrm{d}^{4} / 64=1.92 \mathrm{~cm}^{4}\left(0.045\right.$ inch $\left.^{4}\right)$; diameter of dowel bars $\mathrm{d}=2.5 \mathrm{~cm}$ ( 0.98 inch $)$. Thus $\beta=1.85 \mathrm{~cm}(0.73$ inch $)$.

Since the maximum shear force on the steel member ' 301 ' $\mathrm{P}_{\mathrm{t}}$ is 13.11 KN ( 2937 pounds), concrete bearing stress $\sigma_{b}$ can be determined by $\sigma_{b}=K \times P_{t} \times(2+\beta \times z) /\left(4 \times \beta^{3} \times E_{d} \times I_{d}\right)$, where the joint width $\mathrm{z}=0.5 \mathrm{~cm}(0.2$ inch $) . \sigma_{\mathrm{b}}=4655 \mathrm{psi}(32 \mathrm{Mpa})$.

Relative stiffness of dowel bar $\ell=\left[\left(\mathrm{E} \times \mathrm{h}^{3}\right) /\left(12 \times\left(1-v^{2}\right) \times \mathrm{k}\right]^{1 / 4}\right.$, (1) where modulus of elasticity of concrete $E=27.6 \times 10^{6} \mathrm{KPa}\left(4 \times 10^{6} \mathrm{psi}\right)$; thickness of the slab $\mathrm{h}=20 \mathrm{~cm}$ (7.9 inches); Poisson's ratio of concrete $v=0.15$ and modulus of subgrade reaction $\mathrm{k}=80 \mathrm{MN} / \mathrm{m}^{3}$ ( 294.9 pci ). $\ell=27.5$ inch ( 70 cm ). From the above discussion about dowel group reaction, we normally assume that the shear in each dowel decreases with the distance of the dowel from the point of loading. Being maximum for the dowel under or nearest to the point of loading and zero at a distance of $1.8 \ell, 1.8 \times \ell=126 \mathrm{~cm}$; since two uniformly distributed area loads are 210 cm apart,
the shear force on steel member 308 will not affect the concrete bearing stress under steel member 301. Thus, the maximum bearing stress between concrete and dowel is 32 Mpa ( 4655 psi).

### 3.3.2 Analysis using KENSLABS:

Now we use KENSLABS to solve the same problem, a different set of finite element arrangement is laid out as shown in Fig. 47 and Fig. 48 as below. 4 m ( 157.5 inch) length of the pavement slab on both sides of transverse joint is studied. On one slab, there are total $8 \times 5=40$ elements and $9 \times 6=54$ nodes. The system of units for this example is English (length in inch, force in pound, stress in psi and dowel K value and subgrade reaction in pci).

KENSLABS can define area loads by specifying the boundaries of the load, in this case, the boundary along X-axis of two area loads is the same, from 145.7 inch ( 370 cm ) to 157.5 inch $(400 \mathrm{~cm})$; the Y boundaries are from 0 to 11.8 inch ( 30 cm ) and from 82.7 inch $(210 \mathrm{~cm})$ to 94.5 inch ( 240 cm ).


Fig. 47 Discretize the slab for design example 2 (KENSLABS)


Fig. 48 Numbering of nodes for design example 2 (KENSLABS)

Other necessary information is modulus of elasticity of concrete slab $E=4 \times 10^{6} \mathrm{psi}\left(27.6 \times 10^{6}\right.$ $\mathrm{KPa})$; Poisson's ratio $v=0.15$; modulus of subgrade reaction $\mathrm{k}=295 \mathrm{pci}\left(80 \mathrm{MN} / \mathrm{m}^{3}\right)$; dowel bars are 0.98 inch ( 2.5 cm ) in diameter and 11.8 inch ( 30 cm ) on centers, thus totally 13 dowel bars are used at the transverse joint; joint is 0.2 inch ( 0.5 cm ) wide. Two uniformly distributed area loads 13470 pounds ( 60 KN ) were applied on two 11.8 inch ( 30 cm ) by 11.8 inch ( 30 cm ) areas, load pressure equals to $13470 /(11.8 \times 11.8)=96.7 \mathrm{psi}(667 \mathrm{KPa})$. Assume modulus of dowel support $\mathrm{K}=1.5 \times 10^{6} \mathrm{pci}\left(406950 \mathrm{KN} / \mathrm{m}^{3}\right)$.

Since $K E N S L A B S$ is a computer program designed specifically for concrete pavement stress analysis, after running the program, the report contains the critical stresses such as bearing stress of concrete and shear stress of dowel along joints. Refer to attached Appendix $D$ for more
detailed report; maximum shear force in dowel bars and maximum concrete bearing stress is listed as below.

Maximum shear force at the dowel bars occur at the first dowel which connects node ' 46 ' to node ' 55 ', the force value equals to 2653 pounds ( 11.8 KN ). Maximum bearing stress of concrete occurs at node ' 46 '; the stress value equals 4226 psi ( 29 Mpa ). Stress contour is as shown in Fig. 49 below.
example 2-1iquit Eoundation
© Mas. ' - ' Stress in X Dir. $=-115.9$ 太 Max. ' + ' Stress in X Dir. $=167.8 \mathrm{psi}$
a Unationm Loxd


Fig. 49 Stress contour (liquid foundation)
FOR JOINT HO. 1 , PHEAR IH ONE DOFEL RAR (FATPD) AT THE NODES IS: $\begin{array}{rrrrrrrrrr}46 & -2652.8 & 47 & -1778.6 & 48 & -813.9 & 49 & -288.3 & 50 & -357.4 \\ 51 & -962.3 & 52 & -1851.8 & 53 & -1367.7 & 54 & 160.3 & & \end{array}$
EOR JOINT NO. 1 BEARING STRESS (BEARS) OF CONCRETE AND SHEAR STRESS
(SHEARS) OF DOFELS AT THE NODES ARE:

| 46 | -4226.2 | -3516.9 | 47 | -2833.5 | -2357.9 | 48 | -1296.6 | -1079.0 |
| ---: | ---: | ---: | ---: | ---: | ---: | ---: | ---: | ---: |
| 49 | -459.4 | -382.3 | 50 | -569.4 | -473.8 | 51 | -1533.0 | -1275.7 |
| 52 | -2950.2 | -2455.1 | 53 | -2179.0 | -1813.2 | 54 | -255.4 | -212.5 |

Fig. 50 Maximum shear in one dowel and maximum bearing stress

Now if we change the foundation as solid foundation while remain rest of the information. In stead of providing modulus of subgrade reaction $k$ for liquid foundation, resilient modulus $\mathrm{M}_{\mathrm{R}}$ and Poisson's ratio $v$ of subgrade should be provided for solid foundation. A calibrated equation is recommended to determine $M_{R}$ with knowing subgrade reaction $k,{ }^{(1)} M_{R}=18.8 \times k$, where $k$ is in pci and $\mathrm{M}_{\mathrm{R}}$ in psi. In this design example, $\mathrm{M}_{\mathrm{R}}=18.8 \times 295=5546 \mathrm{psi}(38238 \mathrm{KPa})$, Poission's ratio of subgrade soil $v$ is assumed to be 0.45 .

Comparing with the results basing on liquid foundation, maximum shear force occurs at dowel bar connecting node ' 46 ' and ' 55 ' as 2653 pounds ( 11.8 KN ) and maximum bearing stress of concrete occurs at node ' 46 ' as 4226 psi ( 29 Mpa ); when the analysis is based on solid foundation as shown in Fig. 51 and Fig. 52 below, maximum shear force occurs at the dowel bar connecting node ' 46 ' and node ' 55 ' as 2760 pounds ( 12.3 KN ) and maximum bearing stress of concrete occurs at node ' 46 ' as $4397 \mathrm{psi}(30 \mathrm{Mpa}$ ). Solid foundation is able to provide more realistic results since subgrade reaction $k$ used in liquid foundation is not a true characteristic of soil behavior. ${ }^{(1)}$


Fig. 51 Stress contour (solid foundation)

FOR JOIHT HO. 1 SHEAR IH OHE DOWEL BAR (FAJPD) AT THE HODES IS:

| 46 | -2760.1 | 47 | -1953.2 | 48 | -1043.2 | 49 | -542.6 | 50 | -615.0 |
| :--- | :--- | :--- | :--- | :--- | :--- | :--- | :--- | :--- | :--- |
| 51 | -1222.4 | 52 | -2127.5 | 53 | -1650.2 | 54 | -353.6 |  |  |

FOR TOINT ND. 1 BEARING STRESS (BEARS) OF CONCRETE AND SHEAR STRESS (SHEARS) OF DOHELS AT THE HODES ARE:

| 46 | -4397.2 | -3659.2 | 47 | -3111.6 | -2589.4 | 48 | -1661.9 | -1383.0 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 49 | -864.4 | -719.3 | 50 | -979.7 | -815.3 | 51 | -1947.4 | -1620.5 |
| 52 | $-3389.4$ | $-2820.5$ | 53 | -2629.0 | -2187.8 | 54 | -563.4 | -468.8 |

Fig. 52 Maximum shear in one dowel and maximum bearing stress

Comparing the result of two programs, when we use $S T A A D-I I I$, maximum member shear force occur at member '301' which is the first dowel located at the edge of the slab, shear- $\mathrm{Y}=-13.11$ KN (-2937 pounds); maximum bearing stress between concrete and that dowel is 32 Mpa ( 4655 psi). For $K E N S L A B S$, if under the assumption that the slab was built on liquid foundation, maximum shear in one dowel bar occurs under node '46' where locates the first dowel at the
edge of the slab, shear- $Y=-2653$ pounds ( -11.8 KN ); maximum bearing stress of concrete between concrete and that dowel is $4226 \mathrm{psi}(29 \mathrm{Mpa})$.

As we discussed before, many factors may affect the accuracy of the calculation, the size of elements can be one important factor. Normally the coarser the finite element mesh, the smaller stresses and deflections we can get, thus the coarser mesh may lead to unsafe design. Secondly, KENSLABS program provides the stresses at nodes, while STAAD-III provides the stresses at the center of the elements.

As a finite element program designed specifically for pavement stress analysis, KENSLABS has several advantages in term of simplicity. For the design example $2, K E N S L A B S$ is able to provide straightforward answers that we concern in the pavement design, for instance, concrete bearing stress under the dowel bar. Furthermore, KENSLABS has three foundation options to choose, and solid foundation is believed to be the solution close to real conditions.

## Chapter 4: Conclusions

Over the past few decades, pavement engineering has gone through considerable developments. Even though Empirical Method is a simple design approach which has been adopted by many agencies, there are still some drawbacks. One of the drawbacks might be that the set of calibrated equations or Nomograph can only be applied to common environmental, material and traffic conditions; when these conditions change for specific jobs, the design outputs will no longer be reliable. As a design method combining engineering experience and the analysis of mechanical characteristics of materials, the Mechanistic-Empirical method has been widely accepted and the transition from Empirical Design Method to Mechanistic-Empirical Method is becoming a nearfuture trend.

The most practical and widely used stress analysis method for flexible pavement is Burmister's Elastic Layered Theory. Horizontal tensile strain at the bottom of the asphalt layer and the vertical compressive strain at the top of the subgrade layer are two critical measurements for quantitatively predicting fatigue and rutting damage under repeated traffic load. The most practical method for analyzing rigid pavement is Finite Element Method. Transverse cracking and joint faulting are two major types of distress that adversely affect the concrete pavement performance. Transverse cracking is the effect of fatigue damage, and joint faulting is the result of a combination of many factors: repeated heavy axle load, poor load transfer across the joint; erosion of the supporting base material and upward curling of the slab, etc. Joint faulting has adverse impact on riding quality.

KENSLABS and STAAD-III are both commercial Finite Element software. They are both capable of analyzing stress state and deflection of concrete pavement under applied loads and temperature loads. For KENSLABS, when the slab and loading exhibit symmetry, only one-half or one-quarter of the slab need to be considered. While for STAAD-III, the finite element program for general plane and space structures, the program is not tailored to recognize 'free edge', 'joint' or 'centerline' of the slab, the whole pavement slab information need to be input into the system.

KENSLABS is designed specifically for studying concrete pavement; it is able to provide straightforward solutions for pavement design purpose. For instance, the bearing stress of concrete under dowel bar and stress at corner of the slab under load. Moreover, KENSLABS has three types of foundation, liquid, solid and layered. Since the modulus of subgrade reaction k is not a true characteristic of subgrade soil behaviors, solid foundation is a more realistic solution to treat pavement problems. The ability to evaluate contact condition between slab and subgrade is another advantage of KENSLABS program. In reality, the slab and subgrade are not always in full contact due to temperature curling or pumping. For old pavements under high intensity of traffic, plastic deformation will form at some locations of subgrade at where initial gaps exist. KENSLABS is able to determine the contact condition for partial contact cases with applied load; the stresses and displacements of slab can be analyzed basing on pre-determined gaps and precompression.

As one of the Finite Element software commonly used by many structure engineers, STAAD-III is more suitable for studying general plane and space structures. Comparing with KENSLABS,

STAAD-III is a relatively complicated Finite Element software. Training and experience are necessary for writing the text input file properly. Even though $S T A A D-I I I$ can provide precise calculations for general plane and space structures containing more elements and nodes, KENSLABS is more suitable for pavement design purposes.

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## Appendix A: Use KENSLABS to solve pavement design example 1

## 1. General Information screen.

|  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: |
| TITLE Hesign example 1 |  |  |  |  |
| Type of foundation [0=liquid, 1=solid, 2=layer] [NFOUND] 0 | 0 | Default options are shown by black dots. If not true, please click the other button. |  |  |
| Damage analysis [ $0=$ no, $1=P$ CA criteria, $2=$ user specified] [NDAMA] 0 | 0 |  |  |  |
| Number of periods per year [ ${ }_{\text {NPY] }} 1$ | 1 |  |  |  |
| Number of load groups <br> [NLG] | 1 | T. with unitorm load <br> ${ }^{-}$without uniform load |  |  |
| Number of slab layers [(NLAYER] 1 | 1 |  |  |  |
| Bond between two slab layers [0=unbonded, 1=bonded) [NBOND] 0 | 0 |  |  |  |
| Number of slabs [ ${ }^{\text {a }}$ [NLAB] 1 | 1 | with temperature gradient and/or checking contact |  |  |
| Number of joints $\quad$ [NJOINT] 0 | 0 |  |  |  |
| Nodal number for checking convergence <br> [NNCK] 1 | 1 | without temperature gradient and/or checking contact |  |  |
| Number of nodes for stress printout [ [NPRINT] 0 | 0 |  |  |  |
|  | 0 |  |  |  |
| $N u m b e r ~ o f ~ n o d e s ~ o n ~$ <br> $Y$ | 6 | $\ulcorner$ with concentrated load <br> if without concentrated load |  |  |
| More detailed pintout [ $0=$ no. $1=y \mathrm{~s}$ ]  <br> [MDPO] 1 | 1 |  |  |  |
| Number of nodes with different thicknesses of slab layer 1 [NAT]] 0 | 0 |  |  |  |
| Number of nodes with different thicknesses of slab layer 2 [NAT2] 0 | 0 | Print Data Sef 7 |  |  |
| System of units [0=English. $1=5$ ] $\quad$ [ 0 UNIT] 1 |  |  |  |  |

- NFOUND - type of foundation: ' 0 ' for liquid foundation, ' 1 ' for solid foundation and ' 2 ' for layered foundation. Modulus of subgrade reaction k (pci or $\mathrm{KN} / \mathrm{m}^{3}$ ) in the case of liquid foundation, resilient modulus $\mathrm{M}_{\mathrm{R}}$ (psi or KPa) and Poisson's ration $v$ in the case of solid foundation should be provided at the foundation information screen as shown below. Layered foundation (up to six layers) is recommended when the slab is placed on one or more layers of granular materials. Thickness, elastic modulus and Poisson's ratio of each foundation layer shall be specified.

- SUBMOD - modulus of subgrade reaction $k$ (pci or $\mathrm{KN} / \mathrm{m}^{3}$ ). Provide one modulus of subgrade reaction k if the foundation is uniform and assign NAS $=$ ' 0 ' (refer to 'NAS' below).
- NAS - numbers of additional modulus of subgrade reaction if the foundation is not uniform. Maximum 120 additional $k$ and nodal numbers at where subgrade modulus is different can be assigned.


## Solid Foundation



Poisson's ratio of subgrade [PRS]
0.45

- YMS - Young's modulus (resilient modulus) of subgrade in the case of rigid foundation (psi or KPa ). Resilient modulus $\mathrm{M}_{\mathrm{R}}$ is the elastic modulus based on recoverable strain under repeated loads, the value of $\mathrm{M}_{\mathrm{R}}$ for granular material and fine-grained soil is normally determined by repeated triaxial test. A calibrated equation is recommended to determine $M_{R}$ with knowing subgrade reaction $k, M_{R}$ $=18.8 \times \mathrm{k}$, where k is in pci and $\mathrm{M}_{\mathrm{R}}$ in psi.
- PRS - Poisson's ratio of subgrade for rigid foundation.
- NDAMA - damage analysis, ' 0 ' for no damage analysis is required; ' 1 ' for damage analysis based on fatigue equations recommended by Portland Cement Association and ' 2 ' for fatigue damage analysis based on user specified fatigue coefficients. Damage analysis is based on fatigue cracking only, the allowable number of repetitions of loading is expressed as $\log \mathrm{N}_{\mathrm{f}}=\mathrm{f}_{1}-\mathrm{f}_{2} \times\left(\sigma / \mathrm{S}_{\mathrm{c}}\right)$, where $\sigma$ is the flexural stress in concrete slab and $S_{c}$ is the modulus of rupture of concrete, $f_{1}$ and $f_{2}$ are coefficients. When the program is asked to perform damage analysis, a damage analysis screen as shown below will ask for information such as modulus of rupture of concrete pavement slab $\mathrm{S}_{\mathrm{c}}$ (psi or KPa) and total number of load repetitions for each load group during each period (refer to NPY and NLG below).


Strength and Fatigue Cogficients of Each Slab Layer
Unit . . . . kPa.


- PMR - modulus of rupture of concrete pavement slab $\mathrm{S}_{\mathrm{c}}$ (psi or KPa).

- TNLR - total predicted number of load repetitions for each load group in each period.
- NPY - number of periods per year. Maximum 12 periods can be specified for each year for damage analysis. Modulus of subgrade reaction k varies with the season of the year, after the allowable number of load repetitions is determined based on predicted number of load repetitions for each load group in each period, pavement design life can be estimated.
- NLG - number of load groups. Axle loads can be divided into maximum of 12 groups for damage analysis. In the case of no applied load and only temperature curling effect is considered, $\mathrm{NLG}=$ ' 1 '.
- NLAYER - number of slab layers, maximum 2 layers of slab can be specified in the case of asphalt overlay on top of concrete slab as a rehabilitation method or concrete slab over a cement-treated base.
- NBOND - bond between two slab layers, ' 0 ' for unbonded and ' 1 ' for bonded slab layers. In the case of bonded slab layers, flexural stress in the concrete is reduced by a composite moment of inertia about neutral axis; while in the case of unbonded slab layers, each layer works as an independent slab with the same displacements at nodes.
- NSLAB - number of slabs, maximum 6 slabs can be studied.
- NJOINT - number of joints, maximum 7 joints can be specified. In the case of single slab, NJOINT should be assigned ' 0 '.
- NNCK - nodal number for checking convergence. The program recommends using nodal number under the heaviest load to check convergence. In design example 1 , node ' 1 ' under uniform load is assigned for checking convergence.
- NPRINT - number of nodes for stress printout, stresses at maximum 420 nodes are printed. When NPRINT $=$ ' 0 ', the stresses at every node will be computed and printed.
- NSX - number of nodes on X-axis of symmetry, maximum 50 nodes can be defined. When NSX $=$ ' 0 ', it represents $X$-axis is not an axis of symmetry; when NSX $=$ ' 1 ', it represents X -axis of symmetry is along a joint. In design example $1, \mathrm{X}$-axis is not an axis of symmetry, thus NSX $=$ ' 0 '.
- NSY - number of nodes on Y-axis of symmetry, maximum 50 nodes can be defined. When NSY $=$ ' 0 ', it represents $Y$-axis is not an axis of symmetry; when NSY $=$ ' 1 ', it represents $Y$-axis of symmetry is along a joint. In design example 1, there are 6 nodes on Y-axis of symmetry, thus NSY $=$ ' 6 '.
- MDPO-more detailed printout, ' 1 ' for yes and ' 0 ' for no.
- NAT1 - number of nodes with different thickness of pavement slab layer 1 , maximum 120 nodes can be specified with different thickness. When NAT1 = ' 0 ', the slab layer 1 is of uniform thickness. In design example $1, \mathrm{NAT1}={ }^{\prime} 0$ '.
- NAT2 - number of nodes with different thickness of pavement slab layer 2, maximum 120 nodes can be specified with different thickness. When NAT2 $=$ ' 0 ', the slab layer 2 is of uniform thickness. In design example 1, NAT2 $=$ ' 0 '.
- NUNIT - system of units, ' 0 ' for English (length in inch, force in pounds, stress in psi, subgrade reaction and dowel support K value in pci and temperature in ${ }^{\circ} \mathrm{F}$ ) and ' 1 ' for SI (length in cm , force in KN , stress in KPa , subgrade reaction and dowel support K value in $\mathrm{KN} / \mathrm{m}^{3}$ and temperature in ${ }^{\circ} \mathrm{C}$ ).


## 2. Temperature load screen.

| Chrinf ind Crnhat monitiph |  |  |
| :---: | :---: | :---: |
| Number of nodes not in contact | [NOTCON] | 10 |
| Number of nodes with initial gaps | (NGAP) | 0 |
| Input of gaps from previous problem (1=yes, 0=no) | [INPUT) | 0 |
| Temperature curling [ $1=$ yes, $0=$ no] | (NTEMP) | 1 |
| Weight of slabs [1=yes, 0=no) | (NW/T) | 0 |
| Maximum number of cycles for checking contact | (NCYCLE) | 1 |
| Temperature differential between top and bottom in C | (TEMP] | 10 |
| Coefficient of thermal expansion per C | [CT] | 0.000009 |
| Tolerance for iteration | (DEL) | 0.001 |
| Maximum allowable deflection in cm | [FMAX] | 2.54 |

- NOTCON - number of nodes not in contact, maximum 120 nodes can be specified where slab and subgrade are assumed not in contact. The concrete slab and subgrade foundation can be in full contact or partial contact depending on whether there is separation due to curling or pumping before any load applications. Nodal numbers at which the slab and subgrade are not in contact have to be specified, and at those nodes subgrade reactive forces are assumed to be 0 .
In the case of liquid foundation, at these nodes initially not in contact, the slab and subgrade may or may not be in contact with load applications depending on how many iteration cycles (refer to NCYCLE below) is specified. If only one iteration cycle is specified, the slab and foundation at these nodes will never be in contact with the load applications. In the case of solid foundation, the nodes specified as not in contact will never be in contact, thus in the case of solid foundation NOTCON should be assigned ' 0 ' unless users are very sure that these nodes will definitely be out of contact.

= NODNC - nodal numbers at where the slab and subgrade are not in contact (nodal numbers do not have to be in sequence).
- NGAP - number of nodes with initial gaps, maximum 120 nodes can be specified. When the pavement slab and subgrade foundation are assumed to be in partial contact, initial gap may or may not exist depending on whether there is pumping or plastic deformation of subgrade induced by repeated heavy traffic. For old pavements under high intensity of traffic, plastic deformation will form at the subgrade; elastic support of subgrade will fail to function at some nodes. For new pavements, each elastic spring of foundation is in good condition thus no initial gap exists. In the case of partial contact will no initial gaps, NGAP $=$ ' 0 '; otherwise, nodal numbers at where initial gaps exist and the sizes of the gap shall be specified.

Initial gaps, temperature curling and the weight of the slab are used to determine contact condition at nodes before any load applications for partially contacted problems; a twostep analysis is recommended when using $K E N S L A B S$ to solve such problems (refer to INPUT below).

- INPUT - input of gaps from previous problem, ' 1 ' for yes and '0' for no. KENSLABS studies partial contact condition with applied load by two-step analysis: In the case of liquid foundation, (1) determine the gap and precompression between slab and subgrade due to temperature curling, self-weight and initial gaps; (2) use the gap and precompression obtained from first step to determine the stresses and displacements under applied load. As shown below, in 'data set 1 ', INPUT $=$ ' 0 ', NCYCLE $=10$ and NWT $=$ ' 1 ' (refer to NCYCLE and NWT below); in 'data set 2 ', INPUT = ' 1 ', NCYCLE $=10$ and NWT $=$ ' 0 '. These two data sets are established in one study case, the first data
set studies the gaps and precompression of slab under temperature curling, self-weight and initial gaps; the second data set determines the stresses and displacement of slab under applied load. These two steps are executed in the same run and one immediately after the other.


In the case of solid foundation, similar to the two-step analysis for liquid foundation, the contact condition of slab and subgrade is determined by analyzing reactive forces and precompression due to temperature curling, initial gaps and self-weight of the slab. When the reactive forces are compressive forces that mean slab and subgrade are in contact, when the reactive forces are negative (tensile) forces, they are assigned to zero and the program runs iteration cycles until no negative (tensile) reactive force is calculated at any node.

- NTEMP - temperature curling, ' 1 ' for yes and ' 0 ' for no.
- NWT - weight of slabs, assign ' 0 ' if weight is not considered and assign ' 1 ' if weight is considered. In the case of full contact of slab and foundation, it is not necessary to consider the weight of the slab; while in the case of partial contact, the weight of the slab must be considered.
Under the full contact condition, the weight of the slab will cause uniform precompression before temperature effect and load application, the slab and foundation will remain in contact under temperature gradient because the deflections due to curling are smaller than the precompression. The stresses and deflections due to curling and loading can be determined separately and each independently of the other.
Under the partial contact condition, in order to determine the stresses and deflection due to applied load, the deformed shape of the slab before any load applications must be determined first. Thus, a two-step analysis is recommended for partial contact condition with applied load. As stated above for INPUT, in dataset 1 , NWT = ' 1 ', the gaps and precompression due to self-weight of the slab, initial gap and temperature curling is
firstly determined; in dataset $2, N W T=$ ' 0 ', the stresses and deflection due to applied load is calculated basing on the gaps and precompression obtained from the first step.
- NCYCLE - maximum number of cycles for checking contact condition. In the case of full contact, $\operatorname{NCYCLE}=$ ' 0 ' and in the case of partial contact, $\operatorname{NCYCLE~}=$ ' 10 '.
In the case of liquid foundation, the program initially assumes the slab and foundation are in full contact for all partially contacted problems and calculate the gaps and precompression due to temperature curling, initial gaps and self-weight of the slab. If it turn out there is no gap between slab and subgrade due to temperature curling, initial gap and self-weight, no iteration cycle is necessary; otherwise up to 10 iterations cycles will be performed until the gaps at nodes calculated are same in sign and within tolerance of iteration (as per DEL below) as that in the previous cycle.
In the case of solid foundation, reactive forces and precompression is used to determine the contact condition for slab under temperature curling, self-weight and initial gaps. If the reactive forces at nodes calculated are all positive in sign (compressive forces), that means the slab and subgrade are in full contact, no iteration cycle is needed; otherwise, all negative reactive forces are assigned to zero and the program will run iteration cycles until there is no negative reactive force calculated.
- TEMP - temperature differential between top and bottom, TEMP $=$ temperature at bottom of slab - temperature at top of slab. Positive for upward curling and negative for downward curling.
- CT - coefficient of thermal expansion of concrete, recommended value is $5 \times 10^{-6}$ in $/ \mathrm{in} /{ }^{\circ} \mathrm{F}$ or $9 \times 10^{-6} / \mathrm{mm} / \mathrm{mm} /{ }^{\circ} \mathrm{C}$.
- DEL - tolerance for iteration, 0.001 is suggested by the program.
- FMAX - maximum allowable vertical deflection at node NNCK (as per above). 1.0 inch $(2.54 \mathrm{~cm})$ is recommended by the program, if the vertical displacement calculated by the program is greater than FMAX, the program will stop.


## 3. Slab information screen.



- NX - number of nodes in X direction for each slab, maximum 15 nodes can be defined. In design example $1, \mathrm{NX}=$ ' 8 '.
- NY - number of nodes in Y direction for each slab, maximum 15 nodes can be defined. In design example $1, \mathrm{NY}=$ ' 6 '.

- X - X coordinate of each node on each slab starting from 0 and increase from left to right.

* Y - Y coordinate of each node on each slab starting from 0 and increase from bottom to top.

- T - thickness of slabs when uniform (cm or inch).
- PR - Poisson's ratio of the pavement slab.
- YM - Young's modulus of concrete pavement slab (psi or KPa).


## 4. Uniform load screen.

## Lqaded Areas for Load Group No. 1

Use <Ctrl>-<Del> to delete a line, <Ctrl>-<Ins> to insert a line, and <Del

| Unit |  | cm | cm | cm | Cm | kFa |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| Load Sequence | LS | XL1 | XL2 | YLT | YL2 | QQ |
| 1 | 1 | 0 | 11.5 | 0 | 33 | 600 |

- LS - slab number on which load is applied.
- XL1 - left limits of loaded area in X local coordinate system.
- XL2 - right limits of loaded area in X local coordinate system.
- YL1 - lower limits of loaded area in Y local coordinate system.
- YL2 - upper limits of loaded area in Y local coordinate system.

- QQ - tire contact pressure of each loaded area (psi or KPa).


## INPUT FILE NAME -C:\KENPAVE\design example 1.TXT

NUMBER OF PROBLEMS TO BE SOLVED = 1

| TITLE -example 1 - liquid foundation |  |  |
| :--- | :--- | :--- | :--- |
| TYPE OF FOUNDATION (NFOUND) | $=$ | 0 |
| TYPE OF DAMAGE ANALYSIS (NDAMA) | $=$ | 0 |
| NUMBER OF PERIODS PER YEAR (NPY) | $=$ | 1 |
| NUMBER OF LOAD GROUPS (NLG) | $=$ | 1 |
| TOTAL NUMBER OF SLABS (NSLAB) | $=$ | 1 |
| TOTAL NUMBER OF JOINTS (NJOINT) | $=$ | 0 |

ARRANGEMENT OF SLABS

| SLAB | NO. NODES (NX) | NO. NODES (NY) | JOINT NO. AT FOUR SIDES | (JONO) |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| NO. | IN X DIRECTION | IN Y DIRECTION | LEFT | RIGHT | BOTTOM | TOP |
| 1 | 8 | 6 | 0 | 0 | 0 | 0 |

```
NUMBER OF LAYERS (NLAYER)-----------------------------------------
NODAL NUMBER USED TO CHECK CONVERGENCE (NNCK)--------..-------------1
NUMBER OF NODES NOT IN CONTACT (NOTCON)----------------------------
NUMBER OF GAPS (NGAP)---------------------------------------------
NUMBER OF POINTS FOR PRINTOUT (NPRINT)---------------------------
CODE FOR INPUT OF GAPS OR PRECOMPRESSIONS (INPUT)--------------- 0
BOND BETWEEN TWO LAYERS (NBOND)--.---------------------------------
CONDITION OF WARPING (NTEMP)-.--.----------------------------------
CODE INDICATING WHETHER SLAB WEIGHT IS CONSIDERED (NWT)--------= 0
MAX NO. OF CYCLES FOR CHECKING CONTACT (NCYCLE)------------------- 1
NUMBER OF ADDITIONAL THICKNESSES FOR SLAB LAYER 1 (NAT1)-------= 0
NUMBER OF ADDITIONAL THICKNESSES FOR SLAB LAYER 2 (NAT2)-------= 0
NUMBER OF POINTS ON X AXIS OF SYMMETRY (NSX)----------------------
NUMBER OF POINTS ON Y AXIS OF SYMMETRY (NSY)---------------------- 6
MORE DETAILED PRINTOUT FOR EACH CONTACT CYCLE (MDPO)------------ 1
TOLERANCE FOR ITERATIONS (DEL)------------------------------------0.001
MAXIMUM ALLOWABLE VERTICAL DISPLACEMENT (FMAX)------------------= 2.54
DIFFERENCE IN TEMP. BETWEEN TOP AND BOTTOM OF SLAB (TEMP) -.-.-= 10
COEFFICIENT OF THERMAL EXPANSION (CT)----------------------------
```

SYSTEM OF UNITS (NUNIT)------------------------------------------1
Length in cm , force in kN , stress in kPa , unit weight in $\mathrm{kN} / \mathrm{m}^{\wedge} 3$ subgrade and dowel K value in $\mathrm{NN} / \mathrm{m}^{\wedge} 3$, and temperature in C

FOR SLAB NO. I COORDINATES OF FINITE ELEMENT GRID ARE:

$\mathrm{X}=$| 0 | 20 | 50 | 100 | 200 | 300 | 400 | 500 |
| :--- | :--- | :--- | :--- | :--- | :--- | :--- | :--- |


$\mathrm{x}=$| 0 | 33 | 60 | 100 | 200 | 300 |
| :--- | :--- | :--- | :--- | :--- | :--- | :--- |


| LAYER | THICKNESS (T) | POISSON'S | YOUNG'S |
| :---: | :---: | :---: | :---: |
| NO. |  |  | RATIO (PR) |
| I | 20.00000 | 0.15000 | MODULUS (YM) |
|  |  | $2.760 \mathrm{E}+07$ |  |

No. OF LOADED AREAS (NUDL) FOR EACH LOAD GROUP ARE: 1
NO. OF NODAL FORCES (NCNF) AND MOMENTS (NCMX AND NCMY) ARE: 0 o 0
FOR LOAD GROUP NO. 1 LOADS ARE APPLIED AS FOLLOWS:
SLAB NO. $X$ COORDINATES $Y$ COORDINATES

| (LS) | (XLI) | (XL2) | (YLI) | (YL2) | (QQ) |
| :---: | :---: | :---: | :---: | :---: | :---: |
| 1 | 0.00000 | 11.50000 | 0.00000 | 33.00000 | 600.00000 |

NODES ON Y AXIS OF SYMMETRY (NODSY) ARE: $1 \begin{array}{llllllllllll} & 2 & 3 & 4 & 5 & 6\end{array}$
FOUNDATION ADJUSTMENT FACTOR (FSAF) FOR EACH PERIOD ARE: 1
NUMBER OF ADDITIONAL SUBGRADE MODULI (NAS) TO BE READ IN-.-.-. 0


SILAB NO., INITIAL NODAL NUMBER (INITNP), LAST NODAL NUMBER (LASTNP), INITIAL ELEMENT NO. (INITEN), AND LAST ELEMENT NO. (LASTEN) ARE;

NODAL COORDINATES (XN AND YN) OF INDIVIDUAL SLAB ARE:

| 1 | 0.000 | 0.000 | 2 | 0.000 | 33.000 | 3 | 0.000 | 60.000 |
| ---: | ---: | ---: | ---: | ---: | ---: | ---: | ---: | ---: |
| 4 | 0.000 | 100.000 | 5 | 0.000 | 200.000 | 6 | 0.000 | 300.000 |
| 7 | 20.000 | 0.000 | 8 | 20.000 | 33.000 | 9 | 20.000 | 60.000 |
| 10 | 20.000 | 100.000 | 11 | 20.000 | 200.000 | 12 | 20.000 | 300.000 |
| 13 | 50.000 | 0.000 | 14 | 50.000 | 33.000 | 15 | 50.000 | 60.000 |
| 16 | 50.000 | 100.000 | 17 | 50.000 | 200.000 | 18 | 50.000 | 300.000 |
| 19 | 100.000 | 0.000 | 20 | 100.000 | 33.000 | 21 | 100.000 | 60.000 |
| 22 | 100.000 | 100.000 | 23 | 100.000 | 200.000 | 24 | 100.000 | 300.000 |
| 25 | 200.000 | 0.000 | 26 | 200.000 | 33.000 | 27 | 200.000 | 60.000 |
| 28 | 200.000 | 100.000 | 29 | 200.000 | 200.000 | 30 | 200.000 | 300.000 |
| 31 | 300.000 | 0.000 | 32 | 300.000 | 33.000 | 33 | 300.000 | 60.000 |
| 34 | 300.000 | 100.000 | 35 | 300.000 | 200.000 | 36 | 300.000 | 300.000 |
| 37 | 400.000 | 0.000 | 38 | 400.000 | 33.000 | 39 | 400.000 | 60.000 |
| 40 | 400.000 | 100.000 | 41 | 400.000 | 200.000 | 42 | 400.000 | 300.000 |
| 43 | 500.000 | 0.000 | 44 | 500.000 | 33.000 | 45 | 500.000 | 60.000 |
| 46 | 500.000 | 100.000 | 47 | 500.000 | 200.000 | 48 | 500.000 | 300.000 |

AMOUNT OF INITIAL CURLING AND GAP (CURL) AT THE NODES IS

| 1 | 0.05063 | 2 | 0.03080 | 3 | 0.01823 | 4 | 0.00563 | 5 | 0.00563 |
| ---: | ---: | ---: | ---: | ---: | ---: | ---: | ---: | ---: | ---: |
| 6 | 0.05063 | 7 | 0.05153 | 8 | 0.03170 | 9 | 0.01913 | 10 | 0.00653 |
| 11 | 0.00653 | 12 | 0.05153 | 13 | 0.05625 | 14 | 0.03643 | 15 | 0.02385 |
| 16 | 0.01125 | 17 | 0.01125 | 18 | 0.05625 | 19 | 0.07313 | 20 | 0.05330 |
| 21 | 0.04073 | 22 | 0.02813 | 23 | 0.02813 | 24 | 0.07313 | 25 | 0.14063 |
| 26 | 0.12080 | 27 | 0.10823 | 28 | 0.09563 | 29 | 0.09563 | 30 | 0.14063 |
| 31 | 0.25313 | 32 | 0.23330 | 33 | 0.22073 | 34 | 0.20813 | 35 | 0.20813 |
| 36 | 0.25313 | 37 | 0.41063 | 38 | 0.39080 | 39 | 0.37823 | 40 | 0.36563 |
| 41 | 0.36563 | 42 | 0.41063 | 43 | 0.61313 | 44 | 0.59330 | 45 | 0.58073 |
| 46 | 0.56813 | 47 | 0.56813 | 48 | 0.61313 |  |  |  |  |

LOADS ARE APPLIED ON THE ELEMENT NO. (NE) WITH COORDINATES (XDA AND YDA)

$$
\begin{array}{cccccc}
1 & -1.000 & 0.150 & -1.000 & 1.000 & 0.060
\end{array}
$$

HALF BAND WIDTH (NB) $=24$
PERIOD 1 LOAD GROUP 1 AND CYCLE NO. 1

| DEFLECTIONS OF SLABS | (F) ARE: | (DOWNWARD POSITIVE) |  |  |  |  |  |  |  |
| :---: | ---: | ---: | ---: | ---: | ---: | ---: | ---: | ---: | ---: | ---: |
| 1 | 0.01871 | 2 | 0.02365 | 3 | 0.02452 | 4 | 0.02318 | 5 | 0.01164 |
| 6 | -0.02154 | 7 | 0.01711 | 8 | 0.02259 | 9 | 0.02388 | 10 | 0.02287 |
| 11 | 0.01161 | 12 | -0.02151 | 13 | 0.01096 | 14 | 0.01831 | 15 | 0.02099 |
| 16 | 0.02139 | 17 | 0.01146 | 18 | -0.02136 | 19 | -0.00132 | 20 | 0.00921 |
| 21 | 0.01431 | 22 | 0.01761 | 23 | 0.01108 | 24 | -0.02080 | 25 | -0.01741 |
| 26 | -0.00292 | 27 | 0.00522 | 28 | 0.01239 | 29 | 0.01143 | 30 | -0.01835 |
| 31 | -0.02103 | 32 | -0.00516 | 33 | 0.00411 | 34 | 0.01275 | 35 | 0.01413 |
| 36 | -0.01419 | 37 | -0.02305 | 38 | -0.00715 | 39 | 0.00227 | 40 | 0.01122 |


| 41 | 0.01322 | 42 | -0.01402 | 43 | -0.04439 | 44 | -0.02885 | 45 | -0.01958 |
| ---: | ---: | ---: | ---: | ---: | ---: | ---: | ---: | ---: | ---: |
| 46 | -0.01076 | 47 | -0.00872 | 48 | -0.03506 |  |  |  |  |

NODAL NUMBER AND REACTIVE PRESSURE (SUBR) ARE: (COMPRESSION POSITIVE)

|  |  |  |  |  |  |  |  |  |  |
| ---: | ---: | ---: | ---: | ---: | ---: | ---: | ---: | ---: | ---: |
| 1 | 10.290 | 2 | 13.005 | 3 | 13.488 | 4 | 12.749 | 5 | 6.401 |
| 6 | -11.848 | 7 | 9.409 | 8 | 12.426 | 9 | 13.134 | 10 | 12.579 |
| 11 | 6.385 | 12 | -11.831 | 13 | 6.026 | 14 | 10.068 | 15 | 11.544 |
| 16 | 11.767 | 17 | 6.303 | 18 | -11.745 | 19 | -0.726 | 20 | .5 .068 |
| 21 | 7.872 | 22 | 9.684 | 23 | 6.096 | 24 | -11.441 | 25 | -9.576 |
| 26 | -1.605 | 27 | 2.872 | 28 | 6.815 | 29 | 6.288 | 30 | -10.095 |
| 31 | -11.567 | 32 | -2.836 | 33 | 2.258 | 34 | 7.010 | 35 | 7.770 |
| 36 | -7.803 | 37 | -12.675 | 38 | -3.932 | 39 | 1.251 | 40 | 6.174 |
| 41 | 7.272 | 42 | -7.710 | 43 | -24.417 | 44 | -15.869 | 45 | -10.770 |
| 46 | -5.917 | 47 | -4.797 | 48 | -19.284 |  |  |  |  |


| NODE | ROTAT.X | ROTAT.Y | NODE | ROTAT.X | ROTAT. Y | NODE | RORAT.X |  |
| ---: | ---: | ---: | ---: | ---: | ---: | ---: | ---: | ---: | ---: |
| 1 | $4.575 E-04$ | $-1.219 E-18$ | 2 | $4.469 E-04$ | $-8.919 \mathrm{E}-19$ | 3 | $4.081 \mathrm{E}-04$ | $9.852 \mathrm{E}-19$ |
| 4 | $2.808 \mathrm{E}-04$ | $4.748 \mathrm{E}-18$ | 5 | $-3.505 \mathrm{E}-05$ | $8.498 \mathrm{E}-18$ | 6 | $-1.410 \mathrm{E}-04$ | $4.349 \mathrm{E}-18$ |
| 7 | $4.374 \mathrm{E}-04$ | $-5.681 \mathrm{E}-05$ | 8 | $4.310 \mathrm{E}-04$ | $-8.537 \mathrm{E}-06$ | 9 | $3.957 \mathrm{E}-04$ | $2.742 \mathrm{E}-05$ |
| 10 | $2.752 \mathrm{E}-04$ | $5.958 \mathrm{E}-05$ | 11 | $-3.641 \mathrm{E}-05$ | $8.703 \mathrm{E}-05$ | 12 | $-1.414 \mathrm{E}-04$ | $9.300 \mathrm{E}-05$ |
| 13 | $3.744 \mathrm{E}-04$ | $-1.529 \mathrm{E}-05$ | 14 | $3.770 \mathrm{E}-04$ | $5.167 \mathrm{E}-05$ | 15 | $3.496 \mathrm{E}-04$ | $1.034 \mathrm{E}-04$ |
| 16 | $2.498 \mathrm{E}-04$ | $1.601 \mathrm{E}-04$ | 17 | $-4.295 \mathrm{E}-05$ | $2.184 \mathrm{E}-04$ | 18 | $-1.432 \mathrm{E}-04$ | $2.324 \mathrm{E}-04$ |
| 19 | $2.731 \mathrm{E}-04$ | $2.244 \mathrm{E}-04$ | 20 | $2.841 \mathrm{E}-04$ | $2.764 \mathrm{E}-04$ | 21 | $2.653 \mathrm{E}-04$ | $3.191 \mathrm{E}-04$ |
| 22 | $1.916 \mathrm{E}-04$ | $3.727 \mathrm{E}-04$ | 23 | $-6.054 \mathrm{E}-05$ | $4.431 \mathrm{E}-04$ | 24 | $-1.492 \mathrm{E}-04$ | $4.649 \mathrm{E}-04$ |
| 25 | $1.515 \mathrm{E}-04$ | $8.078 \mathrm{E}-04$ | 26 | $1.671 \mathrm{E}-04$ | $8.341 \mathrm{E}-04$ | 27 | $1.578 \mathrm{E}-04$ | $8.540 \mathrm{E}-04$ |
| 28 | $1.068 \mathrm{E}-04$ | $8.793 \mathrm{E}-04$ | 29 | $-9.276 \mathrm{E}-05$ | $9.176 \mathrm{E}-04$ | 30 | $-1.645 \mathrm{E}-04$ | $9.357 \mathrm{E}-04$ |
| 31 | $1.112 \mathrm{E}-04$ | $1.347 \mathrm{E}-03$ | 32 | $1.247 \mathrm{E}-04$ | $1.354 \mathrm{E}-03$ | 33 | $1.177 \mathrm{E}-04$ | $1.359 \mathrm{E}-03$ |
| 34 | $7.374 \mathrm{E}-05$ | $1.366 \mathrm{E}-03$ | 35 | $-1.088 \mathrm{E}-04$ | $1.378 \mathrm{E}-03$ | 36 | $-1.802 \mathrm{E}-04$ | $1.390 \mathrm{E}-03$ |
| 37 | $1.151 \mathrm{E}-04$ | $1.723 \mathrm{E}-03$ | 38 | $1.204 \mathrm{E}-04$ | $1.722 \mathrm{E}-03$ | 39 | $1.104 \mathrm{E}-04$ | $1.721 \mathrm{E}-03$ |
| 40 | $6.626 \mathrm{E}-05$ | $1.722 \mathrm{E}-03$ | 41 | $-1.147 \mathrm{E}-04$ | $1.724 \mathrm{E}-03$ | 42 | $-2.001 \mathrm{E}-04$ | $1.733 \mathrm{E}-03$ |
| 43 | $1.308 \mathrm{E}-04$ | $1.854 \mathrm{E}-03$ | 44 | $1.276 \mathrm{E}-04$ | $1.850 \mathrm{E}-03$ | 45 | $1.149 \mathrm{E}-04$ | $1.846 \mathrm{E}-03$ |
| 46 | $6.808 \mathrm{E}-05$ | $1.841 \mathrm{E}-03$ | 47 | $-1.179 \mathrm{E}-04$ | $1.839 \mathrm{E}-03$ | 48 | $-2.190 \mathrm{E}-04$ | $1.854 \mathrm{E}-03$ |

## SUM OF FORCES (FOSUM) $=22.8$ <br> SUM OF REACTIONS (SUBSUM) $=22.8$

| NODE LAYER | STRESS X | STRESS Y | STRESS XY | MAX.SHEAR | MAJOR | MINOR |  |
| :---: | ---: | ---: | ---: | ---: | ---: | ---: | ---: |
| 1 | 1 | -1346.006 | 0.000 | 0.000 | 673.003 | 0.000 | -1346.006 |
| 2 | 1 | -401.183 | -20.808 | 0.000 | 190.187 | -20.808 | -401.183 |
| 3 | 1 | 427.547 | 793.160 | 0.000 | 182.807 | 793.160 | 427.547 |
| 4 | 1 | 978.423 | 1168.842 | 0.000 | 95.210 | 1168.842 | 978.423 |
| 5 | 1 | 1345.533 | 971.715 | 0.000 | 186.909 | 1345.533 | 971.715 |
| 6 | 1 | 1281.328 | 0.000 | 0.000 | 640.664 | 1281.328 | 0.000 |
| 7 | 1 | -240.984 | 0.000 | 0.000 | 120.492 | 0.000 | -240.984 |
| 8 | 1 | 180.014 | 97.598 | -334.787 | 337.314 | 476.120 | -198.508 |
| 9 | 1 | 564.321 | 726.355 | -270.310 | 282.190 | 927.527 | 363.148 |
| 10 | 1 | 1013.099 | 1153.134 | -134.177 | 151.347 | 1234.463 | 931.770 |
| 11 | 1 | 1347.078 | 968.468 | -31.578 | 191.921 | 1349.694 | 965.852 |
| 12 | 1 | 1280.926 | 0.000 | 0.000 | 640.463 | 1280.926 | 0.000 |
| 13 | 1 | 961.863 | 0.000 | 0.000 | 480.932 | 961.863 | 0.000 |
| 14 | 1 | 988.311 | 243.326 | -472.843 | 601.940 | 1217.758 | 13.879 |
| 15 | 1 | 1036.229 | 625.643 | -418.405 | 466.056 | 1296.992 | 364.880 |
| 16 | 1 | 1175.383 | 1065.669 | -256.341 | 262.145 | 1382.670 | 858.381 |
| 17 | 1 | 1357.332 | 950.168 | -70.455 | 215.429 | 1369.179 | 938.321 |
| 18 | 1 | 1280.764 | 0.000 | 0.000 | 640.382 | 1280.764 | 0.000 |
| 19 | 1 | 1659.486 | 0.000 | 0.000 | 829.743 | 1659.486 | 0.000 |
| 20 | 1 | 1558.555 | 289.802 | -379.799 | 739.379 | 1663.557 | 184.800 |
| 21 | 1 | 1517.934 | 559.267 | -362.324 | 600.866 | 1639.466 | 437.735 |


| 22 | 1 | 1473.186 | 912.781 | -267.521 | 387.403 | 1580.386 | 805.581 |
| ---: | ---: | ---: | ---: | ---: | ---: | ---: | ---: |
| 23 | 1 | 1407.287 | 896.665 | -93.467 | 271.882 | 1423.858 | 880.094 |
| 24 | 1 | 1286.094 | 0.000 | 0.000 | 643.047 | 1286.094 | 0.000 |
| 25 | 1 | 1659.202 | 0.000 | 0.000 | 829.601 | 1659.202 | 0.000 |
| 26 | 1 | 1604.426 | 225.567 | -185.677 | 713.995 | 1628.991 | 201.001 |
| 27 | 1 | 1581.998 | 444.005 | -168.691 | 593.476 | 1606.478 | 419.526 |
| 28 | 1 | 1555.137 | 735.256 | -135.669 | 431.807 | 1577.004 | 713.390 |
| 29 | 1 | 1460.786 | 809.275 | -56.793 | 330.669 | 1465.700 | 804.362 |
| 30 | 1 | 1314.837 | 0.000 | 0.000 | 657.419 | 1314.837 | 0.000 |
| 31 | 1 | 1332.498 | 0.000 | 0.000 | 666.249 | 1332.498 | 0.000 |
| 32 | 1 | 1322.657 | 173.268 | -48.927 | 576.774 | 1324.736 | 171.189 |
| 33 | 1 | 1328.335 | 367.243 | -45.249 | 482.672 | 1330.461 | 365.117 |
| 34 | 1 | 1341.526 | 641.312 | -39.637 | 352.343 | 1343.762 | 639.076 |
| 35 | 1 | 1316.622 | 763.934 | -22.375 | 277.248 | 1317.526 | 763.029 |
| 36 | 1 | 1196.054 | 0.000 | 0.000 | 598.027 | 1196.054 | 0.000 |
| 37 | 1 | 727.058 | 0.000 | 0.000 | 363.529 | 727.058 | 0.000 |
| 38 | 1 | 746.666 | 133.304 | 5.165 | 306.724 | 746.709 | 133.261 |
| 39 | 1 | 769.148 | 298.912 | -0.527 | 235.119 | 769.149 | 298.911 |
| 40 | 1 | 802.532 | 547.364 | -4.489 | 127.663 | 802.611 | 547.285 |
| 41 | 1 | 815.072 | 692.182 | -8.183 | 61.988 | 815.614 | 691.639 |
| 42 | 1 | 690.082 | 0.000 | 0.000 | 345.041 | 690.082 | 0.000 |
| 43 | 1 | 0.000 | 0.000 | 0.000 | 0.000 | 0.000 | 0.000 |
| 44 | 1 | 0.000 | 55.105 | 0.000 | 27.553 | 55.105 | 0.000 |
| 45 | 1 | 0.000 | 201.547 | 0.000 | 100.773 | 201.547 | 0.000 |
| 46 | 1 | 0.000 | 442.107 | 0.000 | 221.054 | 442.107 | 0.000 |
| 47 | 1 | 0.000 | 585.670 | 0.000 | 292.835 | 585.670 | 0.000 |
| 48 | 1 | 0.000 | 0.000 | 0.000 | 0.000 | 0.000 | 0.000 |

MAXIMUM STRESS (SMAX) IN LAYER 1 IS 1663.557 (NODE 20 )

MAXIMUM NEGATIVE STRESS IN X DIRECTION = MAXIMUM POSITIVE STRESS IN X DIRECTION = MAXIMUM NEGATIVE STRESS IN Y DIRECTION = MAXIMUM POSITIVE STRESS IN Y DIRECTION =
-1346.0 (NODE 1)
1659.5 ( NODE 19)
-20.8 (NODE 2 )
1168.8 ( NODE 4 )

## Appendix B: Use STAAD-III to solve pavement design example 1

```
****************************************************
* S T A A D - III *
* Revision 22.3b_. *
* Proprietary Program of *
* Research Engineers, Inc. *
* Date= AUG 15, 2005 *
* Time= 11:40:52 *
* Build No. 2500 *
* USER ID: Kawneer Company Canada Limited
**************************************************
    1. STAAD SPACE WARPING-1
    2. UNIT METER KN
    3. JOINT COORDINATES
    4. 1 0 0 0 3 0 0 0.33 1
    5.4}40000.611100031
    6. REPEAT ALL 10 0.4885 0 0
    7. 122 5 0 0 124 5 0 0.33 1
    8. 125 5 0 0.6 132 5 0 3 1
    9. REPEAT ALL 1 0.115 0 0
10. 144 5.6035 0 0 146 5.6035 0 0.33 1
11. 147 5.6035 0 0.6 154 5.6035 0 3 1
12. REPEAT ALL 9 0.4885 0 0
13. *FINISH
14. ELEMENT INCIDENCES
15. 1 1 1 2 13 12 TO 10 1 I
16. REPEAT 21 10 11
17. ELEMENT PROPERTIES
18. I TO 220 TH 0.2
19. UNIT KN METER
20. CONSTANTS
21. E 27.6E6 ELEM 1 TO 220
22. POISSON 0.15 ELEM 1 TO 220
23. ALPHA 9E-6 ELEM 1 TO 220
24. SUPPORT
25. 1 TO 253 ELASTIC MAT DIRECTION Y SUB 55000
26. LOAD 1
27. ELEMENT LOAD
28. 101 102 111 112 PR GY -600
29. LOAD 2
30. TEMP LOAD
31. 1 TO 220 TEMP 0 -10
32. LOAD COMB 3
33. 1 1 2 1
34. LOAD COMB 4
35. 1 1 2-1
36. PERFORM ANAEYSIS
```*
37. PRINT JOINT DISPL LIST 100 TO 132

JOINT DISPLACEMENT (CM RADIANS) STRUCTURE TYPE = SPACE
JOINT LOAD X-TRANS Y-TRANS Z-TRANS X-ROTAN Y-ROTAN Z-ROTAN
\begin{tabular}{|c|c|c|c|c|c|c|c|}
\hline 100 & 1 & . 0000 & -. 0399 & . 0000 & -. 0003 & . 0000 & -. 0003 \\
\hline & 2 & . 0000 & . 0274 & . 0000 & . 0006 & . 0000 & . 0000 \\
\hline & 3 & . 0000 & -. 0125 & . 0000 & . 0003 & . 0000 & -. 0002 \\
\hline & 4 & . 0000 & -. 0673 & . 0000 & -. 0009 & . 0000 & -. 0003 \\
\hline 101 & 1 & . 0000 & -. 0351 & . 0000 & -. 0003 & . 0000 & -. 0002 \\
\hline & 2 & . 0000 & . 0183 & . 0000 & . 0005 & . 0000 & . 0000 \\
\hline & 3 & . 0000 & -. 0168 & . 0000 & . 0002 & . 0000 & -. 0002 \\
\hline & 4 & . 0000 & -. 0533 & . 0000 & -. 0008 & . 0000 & -. 0003 \\
\hline 102 & 1 & . 0000 & -. 0303 & . 0000 & -. 0003 & . 0000 & -. 0002 \\
\hline & 2 & . 0000 & . 0105 & . 0000 & . 0004 & . 0000 & . 0000 \\
\hline & 3 & . 0000 & -. 0198 & . 0000 & . 0001 & . 0000 & -. 0002 \\
\hline & 4 & . 0000 & -. 0408 & . 0000 & -. 0007 & . 0000 & -. 0002 \\
\hline 103 & 1 & . 0000 & -. 0229 & . 0000 & -. 0003 & . 0000 & -. 0001 \\
\hline & 2 & . 0000 & . 0006 & . 0000 & . 0003 & . 0000 & . 0000 \\
\hline & 3 & 0000 & -. 0223 & . 0000 & . 0000 & . 0000 & -. 0001 \\
\hline & 4 & . 0000 & -. 0235 & . 0000 & -. 0006 & . 0000 & -. 0002 \\
\hline 104 & 1 & . 0000 & -. 0148 & . 0000 & -. 0002 & . 0000 & -. 00001 \\
\hline & 2 & . 0000 & -. 0076 & . 0000 & . 0002 & . 0000 & . 0000 \\
\hline & 3 & . 0000 & -. 0224 & . 0000 & . 0000 & . 0000 & -. 0001 \\
\hline & 4 & . 0000 & -. 0072 & . 0000 & -. 0004 & . 0000 & -. 0001 \\
\hline 105 & 1 & . 0000 & -. 0086 & . 0000 & -. 0002 & . 0000 & -. 0001 \\
\hline & 2 & . 0000 & -. 0116 & . 0000 & . 0001 & . 0000 & . 0000 \\
\hline & 3 & . 0000 & -. 0202 & . 0000 & -. 0001 & . 0000 & . 0000 \\
\hline & 4 & . 0000 & . 0031 & . 0000 & -. 0002 & . 0000 & -. 0001 \\
\hline 106 & 1 & . 0000 & -. 0041 & . 0000 & -. 0001 & . 0000 & . 0000 \\
\hline & 2 & . 0000 & -. 0121 & . 0000 & . 0000 & . 0000 & . 0000 \\
\hline & 3 & . 0000 & -. 0162 & . 0000 & -. 0001 & . 0000 & . 0000 \\
\hline & 4 & . 0000 & . 0080 & . 0000 & -. 0001 & . 0000 & . 0000 \\
\hline 107 & 1 & . 0000 & -. 0011 & . 0000 & -. 0001 & . 0000 & . 0000 \\
\hline & 2 & . 0000 & -. 0091 & . 0000 & -. 0001 & . 0000 & . 0000 \\
\hline & 3 & . 0000 & -. 0102 & . 0000 & -. 0002 & . 0000 & . 0000 \\
\hline & 4 & . 0000 & . 0080 & . 0000 & . 0001 & . 0000 & . 0000 \\
\hline 108 & 1 & . 0000 & . 0010 & . 0000 & -. 0001 & . 0000 & . 0000 \\
\hline & 2 & . 0000 & -. 0022 & . 0000 & -. 0003 & . 0000 & . 0000 \\
\hline & 3 & . 0000 & -. 0011 & . 0000 & -. 0003 & . 0000 & . 0000 \\
\hline & 4 & . 0000 & . 0032 & . 0000 & . 0002 & . 0000 & . 0000 \\
\hline 109 & 1 & . 0000 & . 0026 & . 0000 & . 0000 & . 0000 & . 0000 \\
\hline & 2 & . 0000 & . 0094 & . 0000 & -. 0004 & . 0000 & . 0000 \\
\hline & 3 & . 0000 & . 0120 & . 0000 & -. 0005 & . 0000 & . 0000 \\
\hline & 4 & . 0000 & -. 0068 & . 0000 & . 0004 & . 0000 & . 0000 \\
\hline 110 & 1 & . 0000 & . 0040 & . 0000 & . 0000 & . 0000 & . 0000 \\
\hline & 2 & . 0000 & . 0266 & . 0000 & -. 0006 & . 0000 & . 0000 \\
\hline & 3 & . 0000 & . 0306 & . 0000 & -. 0006 & . 0000 & . 0000 \\
\hline & 4 & . 0000 & -. 0225 & . 0000 & . 0005 & . 0000 & . 0000 \\
\hline 111 & 1 & . 0000 & -. 0507 & . 0000 & -. 0004 & . 0000 & \\
\hline & 2 & . 0000 & . 0276 & . 0000 & . 0006 & . 0000 & \\
\hline & 3 & . 0000 & -. 0231 & . 0000 & . 0002 & . 0000 & \\
\hline & 4 & . 0000 & -. 0783 & . 0000 & -. 0010 & . 0000 & -. 0001 \\
\hline
\end{tabular}

\begin{tabular}{|c|c|c|c|c|c|c|c|}
\hline \multirow[t]{4}{*}{112} & 1 & . 0000 & -. 0441 & . 0000 & -. 0004 & . 0000 & -. 0001 \\
\hline & 2 & . 0000 & . 0184 & . 0000 & . 0005 & . .0000 & . 0000 \\
\hline & 3 & . 0000 & -. 0257 & . 0000 & . 0001 & . 0000 & -. 0001 \\
\hline & 4 & . 0000 & -. 0625 & . 0000 & -. 0009 & . 0000 & -. 0001 \\
\hline \multirow[t]{4}{*}{113} & 1 & . 0000 & -. 0377 & . 0000 & -. 0004 & . 0000 & -. 0001 \\
\hline & 2 & . 0000 & . 0107 & . 0000 & . 0004 & . 0000 & . 0000 \\
\hline & 3 & . 0000 & -. 0270 & . 0000 & . 0001 & . 0000 & -. 0001 \\
\hline & 4 & . 0000 & -. 0483 & . 0000 & -. 0008 & . 0000 & -. 0001 \\
\hline \multirow[t]{4}{*}{114} & 1 & . 0000 & -. 0278 & . 0000 & -. 0003 & . 0000 & . 0000 \\
\hline & 2 & . 0000 & . 0008 & . 0000 & . 0003 & . 0000 & . 0000 \\
\hline & 3 & . 0000 & -. 0271 & . 0000 & . 0000 & . 0000 & . 0000 \\
\hline & 4 & . 0000 & -. 0286 & . 0000 & -. 0006 & . 0000 & . 0000 \\
\hline \multirow[t]{4}{*}{115} & 1 & . 0000 & -. 0177 & . 0000 & -. 0003 & . 0000 & . 0000 \\
\hline & 2 & . 0000 & -. 0074 & . 0000 & . 0002 & . 0000 & . 0000 \\
\hline & 3 & . 0000 & -. 0251 & . 0000 & -. 0001 & . 0000 & . 0000 \\
\hline & 4 & . 0000 & -. 0103 & . 0000 & -. 0004 & . 0000 & . 0000 \\
\hline \multirow[t]{4}{*}{116} & 1 & . 0000 & -. 0102 & . 0000 & -. 0002 & . 0000 & . 0000 \\
\hline & 2 & . 0000 & -. 0115 & . 0000 & . 0001 & . 0000 & . 0000 \\
\hline & 3 & . 0000 & -. 0217 & . 0000 & -. 0001 & . 0000 & . 0000 \\
\hline & 4 & . 0000 & . 0013 & . 0000 & -. 0002 & . 0000 & . 0000 \\
\hline \multirow[t]{4}{*}{117} & 1 & . 0000 & -. 0050 & . 0000 & -. 0001 & . 0000 & . 0000 \\
\hline & 2 & . 0000 & -. 0119 & . 0000 & . 0000 & . 0000 & . 0000 \\
\hline & 3 & . 0000 & -. 0170 & . 0000 & -. 0002 & . 0000 & . 0000 \\
\hline & 4 & . 0000 & . 0069 & . 0000 & -. 0001 & . 0000 & . 0000 \\
\hline \multirow[t]{4}{*}{118} & 1 & . 0000 & -. 0015 & . 0000 & -. 0001 & . 0000 & . 0000 \\
\hline & 2 & . 0000 & -. 0089 & . 0000 & -. 0001 & . 0000 & . 0000 \\
\hline & 3 & . 0000 & -. 0105 & . 0000 & -. 0002 & . 0000 & . 0000 \\
\hline & 4 & . 0000 & . 0074 & . 0000 & . 0001 & . 0000 & .0000 \\
\hline \multirow[t]{4}{*}{119} & 1 & . 0000 & . 0008 & . 0000 & -. 0001 & . 0000 & . 0000 \\
\hline & 2 & . 0000 & -. 0020 & . 0000 & -. 0003 & . 0000 & . 0000 \\
\hline & 3 & . 0000 & -. 0012 & . 0000 & -. 0003 & . 0000 & . 0000 \\
\hline & 4 & . 0000 & . 0028 & . 0000 & . 0002 & . 0000 & . 0000 \\
\hline \multirow[t]{4}{*}{120} & 1 & . 0000 & . 0026 & . 0000 & . 0000 & .0000 & . 0000 \\
\hline & 2 & . 0000 & . 0096 & . 0000 & -. 0004 & . 0000 & . 0000 \\
\hline & 3 & . 0000 & . 0122 & . 0000 & -. 0005 & . 0000 & . 0000 \\
\hline & 4 & . 0000 & -. 0070 & . 0000 & . 0004 & . 0000 & . 0000 \\
\hline \multirow[t]{4}{*}{121} & 1 & . 0000 & . 0041 & . 0000 & . 0000 & . 0000 & . 0000 \\
\hline & 2 & . 0000 & . 0267 & . 0000 & -. 0006 & . 0000 & . 0000 \\
\hline & 3 & . 0000 & . 0308 & . 0000 & -. 0006 & . 0000 & . 0000 \\
\hline & 4 & . 0000 & -. 0226 & . 0000 & . 0005 & . 0000 & . 0000 \\
\hline \multirow[t]{4}{*}{122} & 1 & . 0000 & -. 0515 & . 0000 & -. 0004 & . 0000 & . 0000 \\
\hline & 2 & . 0000 & . 0276 & . 0000 & . 0006 & . 0000 & . 0000 \\
\hline & 3 & . 0000 & -. 0239 & . 0000 & . 0002 & . 0000 & . 0000 \\
\hline & 4 & . 0000 & -. 0790 & . 0000 & -. 0010 & . 0000 & . 0000 \\
\hline \multirow[t]{4}{*}{123} & 1 & . 0000 & -. 0447 & . 0000 & -. 0004 & . 0000 & . 0000 \\
\hline & 2 & . 0000 & . 0184 & . 0000 & . 0005 & . 0000 & . 0000 \\
\hline & 3 & . 0000 & -. 0263 & . 0000 & . 0001 & . 0000 & . 0000 \\
\hline & 4 & . 0000 & -. 0632 & . 0000 & -. 0009 & . 0000 & . 0000 \\
\hline
\end{tabular}
\begin{tabular}{|c|c|c|c|c|c|c|c|}
\hline \multicolumn{3}{|l|}{JOINT DISPLACEMENT (CM} & RADIANS) & \multicolumn{3}{|l|}{) STRUCTURE TYPE = SPACE} & \\
\hline JOINT & LOAD & X-TRANS & Y-TRANS & Z-TRANS & X-ROTAN & Y-ROTAN & Z-ROTAN \\
\hline \multirow[t]{4}{*}{124} & 1 & . 0000 & -. 0381 & . 0000 & -. 0004 & . 0000 & 0000 \\
\hline & 2 & .0000 & . 0107 & . 0000 & .0004
.0004 & . 0000 & . 0000 \\
\hline & 3 & .0000 & -. 0274 & . 0000 & . 0000 & . 0000 & 0000 \\
\hline & 4 & . 0000 & -. 0488 & . 0000 & -. 0008 & . 0000 & . 0000 \\
\hline \multirow[t]{4}{*}{125} & 1 & . 0000 & -. 0281 & . 0000 & -. 0003 & . 0000 & . 0000 \\
\hline & 2 & . 0000 & . 0008 & . 0000 & . 0003 & . 0000 & . 0000 \\
\hline & 3 & . 0000 & -. 0273 & . 0000 & . 0000 & . 0000 & . 0000 \\
\hline & 4 & . 0000 & -. 0288 & . 0000 & -. 0006 & . 0000 & . 0000 \\
\hline \multirow[t]{4}{*}{126} & 1 & . 0000 & -. 0178 & .0000 & -. 00003 & . 0000 & . 0000 \\
\hline & 2 & . 0000 & -. 0074 & . 0000 & . 0002 & . 0000 & . 0000 \\
\hline & 3 & . 0000 & -. 0252 & . 0000 & -. 0001 & . 0000 & . 0000 \\
\hline & 4 & . 0000 & -. 0104 & . 0000 & -. 0004 & . 0000 & . 0000 \\
\hline \multirow[t]{4}{*}{127} & 1 & . 0000 & -. 0103 & . 0000 & -. 0002 & . 0000 & . 0000 \\
\hline & 2 & . 0000 & -. 0115 & . 0000 & . 0001 & . 0000 & . 0000 \\
\hline & 3 & . 0000 & -. 0217 & . 0000 & -. 0001 & . 0000 & . 0000 \\
\hline & 4 & . 0000 & . 0012 & . 0000 & -. 0002 & . 0000 & . 0000 \\
\hline \multirow[t]{4}{*}{128} & 1 & . 0000 & -. 0051 & . 0000 & -. 0001 & . 0000 & . 0000 \\
\hline & 2 & . 0000 & -. 0119 & . 0000 & . 0000 & . 0000 & . 0000 \\
\hline & 3 & . 0000 & -. 0170 & . 0000 & -. 0002 & . 0000 & . 0000 \\
\hline & 4 & . 0000 & . 0069 & . 0000 & -. 0001 & . 0000 & . 0000 \\
\hline \multirow[t]{4}{*}{129} & 1 & . 0000 & -. 0016 & . 0000 & -. 0001 & . 0000 & . 0000 \\
\hline & 2 & . 0000 & -. 0089 & . 0000 & -. 0001 & . 0000 & . 0000 \\
\hline & 3 & . 0000 & -. 0105 & . 0000 & -. 0002 & . 0000 & . 0000 \\
\hline & 4 & . 0000 & . 0074 & . 0000 & . 0001 & . 0000 & . 0000 \\
\hline \multirow[t]{4}{*}{130} & 1 & . 0000 & . 0008 & . 0000 & -. 0001 & . 0000 & . 0000 \\
\hline & 2 & . 0000 & -. 0020 & . 0000 & -. 0003 & . 0000 & . 0000 \\
\hline & 3 & . 0000 & -. 0012 & . 0000 & -. 0003 & . 0000 & . 0000 \\
\hline & 4 & . 0000 & . 0028 & . 0000 & . 0002 & . 0000 & . 0000 \\
\hline \multirow[t]{4}{*}{131} & 1 & . 0000 & . 0026 & . 0000 & . 0000 & . 0000 & .0000 \\
\hline & 2 & . 0000 & . 0096 & . 0000 & -. 0004 & . 0000 & . 0000 \\
\hline & 3 & . 0000 & . 0122 & . 0000 & -. 0005 & . 0000 & .0000 \\
\hline & 4 & . 0000 & -. 0070 & . 0000 & . 0004 & . 0000 & . 0000 \\
\hline \multirow[t]{4}{*}{132} & 1 & . 0000 & . 0041 & . 0000 & . 0000 & . 0000 & . 0000 \\
\hline & 2 & . 0000 & . 0267 & . 0000 & -. 0006 & . 0000 & . 0000 \\
\hline & 3 & . 0000 & . 0308 & . 0000 & -. 0006 & . 0000 & . 0000 \\
\hline & 4 & . 0000 & -. 0226 & . 0000 & . 0005 & . 0000 & . 0000 \\
\hline
\end{tabular}
\begin{tabular}{|c|c|c|c|c|c|c|c|}
\hline \multicolumn{2}{|l|}{ELEMENT FORCES} & \multicolumn{2}{|l|}{FORCE, LENGTH UNITS \(=\mathrm{KN}\)} & \multicolumn{2}{|l|}{METE} & \multicolumn{2}{|l|}{\multirow[t]{2}{*}{}} \\
\hline \multicolumn{2}{|l|}{FORCE OR STRESS} & \multicolumn{2}{|l|}{\(=\) FORCE/UN} & MOMENT & FORCE-LENGTH/UNIT WIDTH & & \\
\hline ELEMENT I & LOAD & \[
\begin{gathered}
\text { QX } \\
\text { VONT }
\end{gathered}
\] & \[
\begin{gathered}
\text { QY } \\
\text { VONB }
\end{gathered}
\] & \[
\begin{aligned}
& M X \\
& \mathrm{FX}
\end{aligned}
\] & \[
\begin{aligned}
& \text { MY } \\
& \mathrm{FY}
\end{aligned}
\] & & \[
\begin{aligned}
& \text { MXY } \\
& \text { FXY }
\end{aligned}
\] \\
\hline \multirow[t]{2}{*}{100} & 1 & -2.58 & 2.10 & . 14 & . 05 & & -. 09 \\
\hline & & 30.28 & 30.28 & . 00 & . 00 & & -. 00 \\
\hline \multirow[t]{4}{*}{BOTT:} & : SMAX \(=\) & 29.66 & SMIN \(=-1.21\) & TMAX \(=\) & 15.43 & ANGLE= & -33.2 \\
\hline & : SMAX= & 1.21 & SMIN \(=-29.66\) & TMAX \(=\) & 15.43 & ANGLE= & -33.2 \\
\hline & 2 & -11.35 & . 04 & . 43 & 8.56 & ANGLE & . 02 \\
\hline & & 1252.48 & 1252.48 & . 00 & . 00 & & . 00 \\
\hline TOP : & : SMAX \(=\) & 1283.75 & SMIN \(=65.09\) & TMAX \(=\) & 609.33 & ANGLE= & -. 2 \\
\hline \multirow[t]{3}{*}{BOTT :} & : SMAX= & -65.09 & SMIN \(=-1283.75\) & TMAX \(=\) & 609.33 & ANGLE= & -. 2 \\
\hline & 3 & -13.93 & 2.14 & . 57 & 8.61 & & -. 07 \\
\hline & & 1251.39 & 1251.39 & . 00 & . 00 & & . 00 \\
\hline TOP : & : SMAX= & 1291.90 & SMIN \(=85.39\) & TMAX \(=\) & 603.25 & ANGLE= & . 5 \\
\hline \multirow[t]{3}{*}{BOTT :} & : SMAX \(=\) & -85.39 & SMIN \(=-1291.90\) & TMAX \(=\) & 603.25 & ANGLE= & . 5 \\
\hline & 4 & 8.77 & 2.07 & -. 30 & \[
-8.50
\] & & -. 12 \\
\hline & & 1254.30 & 1254.30 & . 00 & . 00 & & . 00 \\
\hline TOP : & : SMAX \(=\) & -44.46 & SMIN \(=-1275.94\) & TMAX \(=\) & 615.74 & ANGLE= & -. 8 \\
\hline BOTT : & : SMAX= & 1275.94 & SMIN \(=44.46\) & TMAX \(=\) & 615.74 & ANGLE= & -. 8 \\
\hline \multirow[t]{2}{*}{101} & 1 & -6.85 & -154.88 & . 18 & -14.82 & & -. 62 \\
\hline & & 2242.70 & 2242.70 & . 00 & . 00 & & . 00 \\
\hline TOP : & : SMAX= & 30.14 & SMIN \(=-2227.48\) & TMAX \(=\) & 1128.81 & ANGLE \(=\) & -2.4 \\
\hline \multirow[t]{3}{*}{BOTT:} & : SMAX \(=\) & 2227.48 & SMIN \(=\quad-30.14\) & TMAX \(=\) & 1128.81 & ANGLE= & -2.4 \\
\hline & 2 & 6.95 & . 36 & . 08 & 8.31 & & -. 02 \\
\hline & & 1239.68 & 1239.68 & . 00 & . 00 & & . 00 \\
\hline TOP : & : SMAX \(=\) & 1245.83 & SMIN= 12.39 & TMAX \(=\) & 616.72 & ANGLE= & . 1 \\
\hline \multirow[t]{3}{*}{BOTT :} & : SMAX \(=\) & -12.39 & SMIN \(=-1245.83\) & TMAX \(=\) & 616.72 & ANGLE= & . 1 \\
\hline & 3 & . 10 & -154.52 & . 26 & -6.52 & & -. 64 \\
\hline & & 1011.31 & 1011.31 & . 00 & . 00 & & . 00 \\
\hline TOP : & : SMAX= & 47.57 & SMIN \(=-986.69\) & TMAX \(=\) & 517.13 & ANGLE= & -5.3 \\
\hline \multirow[t]{3}{*}{BOTT :} & : SMAX \(=\) & 986.69 & SMIN \(=\quad-47.57\) & TMAX \(=\) & 517.13 & ANGLE= & -5.3 \\
\hline & 4 & -13.80 & -155.23 & . 09 & -23.13 & & -. 60 \\
\hline &  & 3479.98 & 3479.98 & . 00 & . 00 & & . 00 \\
\hline TOP : & : SMAX= & 16.27 & SMIN \(=-3471.82\) & TMAX \(=\) & 1744.04 & ANGLE= & -1.5 \\
\hline BOTT : & : SMAX \(=\) & 3471.82 & SMIN \(=\quad-16.27\) & TMAX \(=\) & 1744.04 & ANGLE= & -1.5 \\
\hline \multirow[t]{2}{*}{102} & 1 & 103.48 & -94.23 & \(-.43\) & -12.79 & & -. 85 \\
\hline & & 1899.59 & 1899.59 & . 00 & . 00 & & . 00 \\
\hline TOP : & : SMAX= & -55.47 & SMIN \(=-1926.72\) & TMAX \(=\) & 935.62 & ANGLE \(=\) & -3.9 \\
\hline \multirow[t]{3}{*}{BOTT:} & : SMAX= & 1926.72 & SMIN \(=\quad 55.47\) & TMAX \(=\) & 935.62 & ANGLE \(=\) & -3.9 \\
\hline & 2 & 16.04 & . 43 & . 41 & 8.40 & & -. 02 \\
\hline & & 1230.08 & 1230.08 & . 00 & . 00 & & . 00 \\
\hline TOP : & : SMAX \(=\) & 1259.75 & SMIN \(=\quad 61.66\) & TMAX \(=\) & 599.05 & ANGLE= & . 2 \\
\hline \multirow[t]{3}{*}{BOTT :} & : SMAX \(=\) & -61.66 & SMIN \(=-1259.75\) & TMAX \(=\) & 599.05 & ANGLE= & . 2 \\
\hline & 3 & 119.52 & -93.80 & -. 02 & -4.39 & & -. 87 \\
\hline & & 694.63 & 694.63 & . 00 & . 00 & & \(\begin{array}{r}.00 \\ \hline 108\end{array}\) \\
\hline TOP : & : SMAX \(=\) & 22.39 & SMIN \(=\quad-683.17\) & TMAX \(=\) & 352.78
352.78 & ANGLE= ANGLE= & \[
-10.8
\] \\
\hline BOTT : & : SMAX= & 683.17 & SMIN= \(\quad-22.39\) & TMAX \(=\) & 352.78 & ANGLE= & -10.8 \\
\hline
\end{tabular}

FORCE OR STRESS＝FORCE／UNIT WIDTH／THICK，MOMENT＝FORCE－LENGTH／UNIT WIDTH
\begin{tabular}{|c|c|c|c|c|c|c|c|}
\hline ELEMENT I & LOAD & \[
\begin{gathered}
\text { QX } \\
\text { VONT }
\end{gathered}
\] & \[
\begin{gathered}
\text { QY } \\
\text { VONB }
\end{gathered}
\] & \[
\begin{aligned}
& \text { MX } \\
& \text { FX }
\end{aligned}
\] & \[
\begin{aligned}
& \text { MY } \\
& \text { FY }
\end{aligned}
\] & & \[
\begin{aligned}
& \text { MXY } \\
& \text { FXY }
\end{aligned}
\] \\
\hline & 4 & 87.44 & －94．66 & －． 84 & －21．19 & & －． 82 \\
\hline & & 3124.19 & 3124.19 & ． 00 & ． 00 & & ． 00 \\
\hline TOP ： & ：SMAX＝ & －120．77 & SMIN \(=-3182.83\) & TMAX \(=\) & 1531.03 & ANGLE＝ & －2．3 \\
\hline BOTT： & ：SMAX＝ & 3182.83 & SMIN \(=120.77\) & TMAX \(=\) & 1531.03 & ANGLE＝ & －2．3 \\
\hline \multirow[t]{2}{*}{103} & 1 & 114.86 & －32．21 & 1.96 & －8．09 & & －． 59 \\
\hline & & 1393.20 & 1393.20 & ． 00 & ． 00 & & ． 00 \\
\hline TOP ： & SMAX \(=\) & 299.24 & SMIN \(=-1219.26\) & TMAX \(=\) & 759.25 & ANGLE＝ & －3．3 \\
\hline \multirow[t]{3}{*}{BOTT ：} & ：SMAX \(=\) & 1219.26 & SMIN \(=-299.24\) & TMAX \(=\) & 759.25 & ANGLE \(=\) & －3．3 \\
\hline & 2 & 24.47 & ． 24 & 1.21 & 8.58 & & －． 04 \\
\hline & & 1206.37 & 1206.37 & ． 00 & ． 00 & & ． 00 \\
\hline TOP ： & ：SMAX \(=\) & 1286.81 & SMIN \(=181.42\) & TMAX \(=\) & 552.69 & ANGLE＝ & ． 3 \\
\hline \multirow[t]{3}{*}{BOTT ：} & ：SMAX \(=\) & －181．42 & SMIN \(=-1286.81\) & TMAX \(=\) & 552.69 & ANGLE＝ & 3 \\
\hline & 3 & 139.33 & －31．97 & 3.17 & ． 48 & & －． 63 \\
\hline & & 472.71 & 472.71 & ． 00 & ． 00 & & 00 \\
\hline TOP ： & ：SMAX \(=\) & 496.46 & SMIN \(=\quad 51.75\) & TMAX \(=\) & 222.35 & ANGLE＝ & －12．5 \\
\hline \multirow[t]{3}{*}{BOTT ：} & ：SMAX＝ & －51．75 & SMIN \(=-496.46\) & TMAX \(=\) & 222.35 & ANGLE＝ & －12．5 \\
\hline & 4 & 90.38 & －32．45 & ． 75 & －16．67 & & －． 55 \\
\hline & & 2563.05 & 2563.05 & ． 00 & ． 00 & & ． 00 \\
\hline TOP ： & ：SMAX＝ & 115.24 & SMIN \(=-2503.48\) & TMAX \(=\) & 1309.36 & ANGLE＝ & －1． 8 \\
\hline BOTT ： & ：SMAX＝ & 2503.48 & SMIN \(=-115.24\) & TMAX \(=\) & 1309．36 & ANGLE＝ & －1．8 \\
\hline \multirow[t]{2}{*}{104} & 1 & 29.13 & －3．85 & 3.64 & －4．11 & & －． 25 \\
\hline & & 1008.89 & 1008.89 & \[
.00
\] & \[
.00
\] & & ． 00 \\
\hline TOP ： & ：SMAX \(=\) & 547.22 & SMIN \(=-617.04\) & TMAX \(=\) & 582.13 & ANGLE＝ & －1．8 \\
\hline \multirow[t]{3}{*}{BOTT ：} & ：SMAX \(=\) & 617.04 & SMIN \(=-547.22\) & TMAX \(=\) & 582.13 & ANGLE＝ & －1．8 \\
\hline & 2 & 26.15 & －． 16 & 2.51 & 8.84 & & －． 05 \\
\hline & & 1183.61 & 1183.61 & ． 00 & ． 00 & & ． 00 \\
\hline TOP ： & ：SMAX \(=\) & 1325.94 & SMIN＝ 376.03 & TMAX \(=\) & 474.96 & ANGLE＝ & ． 4 \\
\hline \multirow[t]{3}{*}{BOTT ：} & ：SMAX＝ & －376．03 & SMIN \(=-1325.94\) & TMAX \(=\) & 474.96 & ANGLE \(=\) & ． 4 \\
\hline & 3 & 55.29 & －4．01 & 6.15 & 4.73 & & －． 29 \\
\hline & & 839.97 & 839.97 & ． 00 & ． 00 & & ． 00 \\
\hline TOP ： & ：SMAX \(=\) & 930.92 & SMIN＝ 701.23 & TMAX \(=\) & 114.85 & ANGLE \(=\) & －11．3 \\
\hline \multirow[t]{3}{*}{BOTT ：} & ：SMAX \(=\) & －701．23 & SMIN＝－930．92 & TMAX \(=\) & 114.85 & ANGLE \(=\) & －11．3 \\
\hline & 4 & 2.98 & －3．69 & 1.13 & －12．95 & & －． 20 \\
\hline & & 2032.74 & 2032.74 & ． 00 & ． 00 & & ． 00 \\
\hline TOP ： & ：SMAX \(=\) & 170.39 & SMIN＝－1942．18 & TMAX \(=\) & 1056.29 & ANGLE＝ & －． 8 \\
\hline BOTT ： & ：SMAX \(=\) & 1942.18 & SMIN \(=-170.39\) & TMAX \(=\) & 1056． 29 & ANGLE＝ & －． 8 \\
\hline \multirow[t]{2}{*}{105} & 1 & －． 69 & －． 88 & 3.65 & －1．99 & & －． 09 \\
\hline & & 742.57 & 742.57 & ． 00 & \[
.00
\] & & ． 00 \\
\hline & ：SMAX \(=\) & 546.99 & SMIN \(=-298.35\) & TMAX \(=\) & 422.67 & ANGLE＝ & －． 9 \\
\hline \multirow[t]{3}{*}{BOTT ：} & ：SMAX \(=\) & 298.35 & SMIN \(=-546.99\) & TMAX \(=\) & 422.67 & ANGLE \(=\) & －． 9 \\
\hline & 2 & 16.56 & －． 41 & 3.70 & 9.06 & & ． 03 \\
\hline & & 1183.61 & 1183.61 & ． 00 & 4.00 & & ． 3 \\
\hline TOP ： & ：SMAX \(=\) & 1359.15 & SMIN＝\(\quad 555.18\) & TMAX \(=\) & 401.98
401.98 & ANGLE＝ & ． 3 \\
\hline BOTT ： & ：SMAX \(=\) & －555．18 & SMIN \(=-1359.15\) & TMAX \(=\) & 401.98 & ANGLE \(=\) & ． 3 \\
\hline
\end{tabular}

FORCE OR STRESS = FORCE/UNIT WIDTH/THICK, MOMENT = FORCE-LENGTH/UNIT WIDTH
\begin{tabular}{|c|c|c|c|c|c|c|c|c|}
\hline ELEMENT L & LOAD & \[
\begin{gathered}
\text { QX } \\
\text { VONT }
\end{gathered}
\] & QY & & \[
\begin{aligned}
& \text { MX } \\
& \text { FX }
\end{aligned}
\] & \[
\begin{aligned}
& M Y \\
& F Y
\end{aligned}
\] & & \[
\begin{aligned}
& \text { MXY } \\
& \text { FXY }
\end{aligned}
\] \\
\hline \multirow[t]{2}{*}{} & \multirow[t]{2}{*}{3} & 15.87 & -1. 2 & & 7.35 & 7.07 & & -. 12 \\
\hline & & 1082.55 & 1082.5 & & . 00 & . .00 & & . 00 \\
\hline \multirow[t]{4}{*}{BOTT:} & : SMAX \(=\) & 1109.19 & SMIN \(=\) & 1053.78 & TMAX \(=\) & 27.70 & ANGLE= & \(-2 i .1\) \\
\hline & : SMAX= & -1053.78 & SMIN= & -1109.19 & TMAX \(=\) & \[
27.70
\] & ANGLE= & -21.1 \\
\hline & \multirow[t]{2}{*}{4} & -17.25 & & & -. 06 & \[
-11.05
\] & ANGEE= & . .06 \\
\hline & & 1653.11 & 1653.1 & & . 00 & . 00 & & . 00 \\
\hline \multirow[t]{2}{*}{BOTT:} & : SMAX \(=\) & -8.40 & SMIN= & -1657.29 & TMAX \(=\) & 824.45 & ANGLE= & -. 3 \\
\hline & : SMAX \(=\) & 1657.29 & SMIN \(=\) & 8.40 & TMAX \(=\) & 824.45 & ANGLE= & -. 3 \\
\hline \multirow[t]{2}{*}{106} & \multirow[t]{2}{*}{1} & -10.64 & . 12 & & 2.98 & -. 96 & & -. 03 \\
\hline & & 533.36 & 533.3 & & . 00 & . 00 & & . 00 \\
\hline TOP : & : SMAX= & 446.62 & SMIN \(=\) & -143.93 & TMAX \(=\) & 295.28 & ANGLE= & -. 4 \\
\hline \multirow[t]{3}{*}{BOTT :} & : SMAX= & 143.93 & SMIN \(=\) & -446.62 & TMAX \(=\) & 295.28 & ANGLE= & -. 4 \\
\hline & \multirow[t]{2}{*}{2} & 2.44 & -. 5 & & 4.23 & 9.16 & & . 00 \\
\hline & & 1190.46 & 1190.4 & & . 00 & . 00 & & . 00 \\
\hline TOP : & SMAX \(=\) & 1373.32 & SMIN= & 634.74 & TMAX \(=\) & 369.29 & ANGLE \(=\) & 1 \\
\hline \multirow[t]{3}{*}{BOTT :} & : SMAX= & -634.74 & SMIN \(=\) & -1373.32 & TMAX \(=\) & 369.29 & ANGLE= & 1 \\
\hline & \multirow[t]{2}{*}{3} & -8.21 & -. 3 & & 7.21 & 8.20 & & -. 03 \\
\hline & & 1162.50 & 1162.50 & & . 00 & . 00 & & . 00 \\
\hline TOP : & : \(\operatorname{SMAX}=\) & 1229.57 & SMIN \(=\) & 1081.17 & TMAX \(=\) & 74.20 & ANGLE= & 1.9 \\
\hline \multirow[t]{3}{*}{BOTT:} & : SMAX= & -1081.17 & SMIN= & -1229.57 & TMAX \(=\) & 74.20 & ANGLE= & 1.9 \\
\hline & \multirow[t]{2}{*}{4} & -13.08 & & & -1.25 & -10.11 & & -. 02 \\
\hline & & 1432.46 & 1432.4 & & . 00 & . 00 & & . 00 \\
\hline TOP : & : SMAX \(=\) & -188.14 & SMIN \(=\) & -1517.23 & TMAX \(=\) & 664.55 & ANGLE= & -. 2 \\
\hline BOTT: & : SMAX \(=\) & 1517.23 & SMIN= & 188.14 & TMAX \(=\) & 664.55 & ANGLE= & -. 2 \\
\hline \multirow[t]{2}{*}{107} & \multirow[t]{2}{*}{1} & -13.34 & 3 & & 2.13 & -. 43 & & . 00 \\
\hline & & 356.32 & 356.3 & & . 00 & . 00 & & . 00 \\
\hline TOP : & SMAX \(=\) & 319.39 & SMIN= & -64.93 & TMAX \(=\) & 192.16 & ANGLE \(=\) & -. 1 \\
\hline \multirow[t]{3}{*}{BOTT :} & : SMAX= & 64.93 & SMIN= & -319.39 & TMAX \(=\) & 192.16 & ANGLE= & -. 1 \\
\hline & \multirow[t]{2}{*}{2} & -12.24 & -. 4 & & 3.96 & 9.11 & & . 02 \\
\hline & & 1186.25 & 1186.2 & & . 00 & . 00 & & . 00 \\
\hline TOP : & : SMAX= & 1365.88 & SMIN \(=\) & 593.71 & TMAX \(=\) & 386.09 & ANGLE= & -. 3 \\
\hline \multirow[t]{3}{*}{BOTT :} & : SMAX= & -593.71 & SMIN \(=\) & -1365.88 & TMAX \(=\) & 386.09 & ANGLE \(=\) & -. 3 \\
\hline & \multirow[t]{2}{*}{3} & -25.58 & -. 1 & & 6.09 & 8.67 & & . 02 \\
\hline & & 1156.86 & 1156.8 & & . 00 & . 00 & & . 00 \\
\hline TOP : & : SMAX= & 1300.96 & SMIN \(=\) & 913.09 & TMAX \(=\) & 193.94 & ANGLE= & -. 5 \\
\hline \multirow[t]{3}{*}{BOTT :} & : SMAX= & -913.09 & SMIN \(=\) & \(-1300.96\) & TMAX \(=\) & 193.94 & ANGLE= & -. 5 \\
\hline & \multirow[t]{2}{*}{4} & -1.10 & 7 & & \(-1.83\) & -9.54 & & -. 03 \\
\hline & & 1315.28 & 1315.2 & & . 00 & . 00 & & . 00 \\
\hline TOP : & : SMAX= & -274.33 & SMIN= & -1430.81 & TMAX \(=\) & 578.24 & ANGLE \(=\) & -. 2 \\
\hline BOTT: & : \(\operatorname{SMAX}=\) & 1430.81 & SMIN \(=\) & 274.33 & TMAX \(=\) & 578.24 & ANGLE & -. 2 \\
\hline \multirow[t]{2}{*}{108} & \multirow[t]{2}{*}{1} & -12.50 & , & & 1.31 & -. 18 & & . 01 \\
\hline & & 210.86 & 210.8 & & . 00 & . 00 & & . 00 \\
\hline TOP : & : SMAX= & 196.40 & SMIN \(=\) & -26.43 & TMAX \(=\) & 111.41 & ANGLE= & - 3 \\
\hline BOTT : & : SMAX= & 26.43 & SMIN= & -196.40 & TMAX \(=\) & 111.41 & ANGLE= & . 3 \\
\hline
\end{tabular}

FORCE OR STRESS = FORCE/UNIT WIDTH/THICK, MOMENT = FORCE-LENGTH/UNIT WIDTH
\begin{tabular}{|c|c|c|c|c|c|c|c|}
\hline ELEMENT I & LOAD & \[
\begin{gathered}
\text { QX } \\
\text { VONT }
\end{gathered}
\] & \[
\begin{gathered}
\text { QY } \\
\text { VONB }
\end{gathered}
\] & \[
\begin{aligned}
& M X \\
& F X
\end{aligned}
\] & \[
\begin{aligned}
& M Y \\
& F Y
\end{aligned}
\] & & \[
\begin{aligned}
& \text { MXY } \\
& \text { FXY }
\end{aligned}
\] \\
\hline \multirow[t]{2}{*}{} & \multirow[t]{2}{*}{2} & -23.46 & -. 30 & 2.96 & 8.92 & & . 04 \\
\hline & & 1180.64 & 1180.64 & . 00 & . 00 & & . 00 \\
\hline TOP : & : SMAX= & 1338.36 & SMIN \(=\quad 444.45\) & TMAX \(=\) & 446.95 & ANGLE= & -. 4 \\
\hline \multirow[t]{3}{*}{BOTT :} & : SMAX \(=\) & -444.45 & SMIN \(=-1338.36\) & TMAX \(=\) & 446.95 & ANGLE= & -. 4 \\
\hline & \multirow[t]{2}{*}{3} & -35.96 & . 04 & 4.27 & 8.75 & & . 05 \\
\hline & & 1136.30 & 1136.30 & . 00 & . 00 & & . 00 \\
\hline TOP : & SMAX \(=\) & 1311.97 & SMIN \(=640.80\) & TMAX = & 335.59 & ANGLE \(=\) & -. 7 \\
\hline \multirow[t]{3}{*}{BOTT :} & : SMAX= & -640.80 & SMIN \(=-1311.97\) & TMAX \(=\) & 335.59 & ANGLE= & -. 7 \\
\hline & \multirow[t]{2}{*}{4} & 10.96 & . 64 & \(-1.65\) & -9.10 & & -. 04 \\
\hline & & 1259.18 & 1259.18 & . 00 & . 00 & & . 00 \\
\hline TOP : & SMAX = & -248.08 & SMIN \(=-1364.76\) & TMAX \(=\) & 558.34 & ANGLE= & -. 3 \\
\hline BOTT : & : SMAX= & 1364.76 & SMIN \(=\quad 248.08\) & TMAX \(=\) & 558.34 & ANGLE= & -. 3 \\
\hline \multirow[t]{2}{*}{109} & \multirow[t]{2}{*}{1} & -10.05 & . 16 & . 62 & -. 04 & & . 01 \\
\hline & & 96.81 & 96.81 & . 00 & . 00 & & . 00 \\
\hline TOP : & : SMAX \(=\) & 93.65 & SMIN \(=\quad-6.03\) & TMAX \(=\) & 49.84 & ANGLE= & 5 \\
\hline \multirow[t]{3}{*}{BOTT :} & : SMAX \(=\) & 5.03 & SMIN \(=\quad-93.65\) & TMAX \(=\) & 49.84 & ANGLE= & . 5 \\
\hline & \multirow[t]{2}{*}{2} & -26.68 & -. 11 & 1.57 & 8.65 & & . 05 \\
\hline & & 1197.46 & 1197.46 & . 00 & . 00 & & . 00 \\
\hline TOP : & : SMAX= & 1297.75 & SMIN \(=235.62\) & TMAX \(=\) & 531.07 & ANGLE= & -. 4 \\
\hline \multirow[t]{3}{*}{BOTT :} & : SMAX \(=\) & -235.62 & SMIN \(=-1297.75\) & TMAX \(=\) & 531.07 & ANGLE= & -. 4 \\
\hline & \multirow[t]{2}{*}{3} & -36.74 & . 06 & 2.20 & 8.61 & & . 06 \\
\hline & & 1162.64 & 1162.64 & . 00 & . 00 & & . 00 \\
\hline TOP : & : SMAX \(=\) & 1291.76 & SMIN \(=329.24\) & TMAX \(=\) & 481.26 & ANGLE \(=\) & -. 5 \\
\hline \multirow[t]{3}{*}{BOTT:} & : SMAX = & -329.24 & SMIN \(=-1291.76\) & TMAX \(=\) & 481.26 & ANGLE \(=\) & -. 5 \\
\hline & \multirow[t]{2}{*}{4} & 16.63 & . 27 & -. 95 & -8.69 & & -. 05 \\
\hline & & 1238.87 & 1238.87 & . 00 & . 00 & & . 00 \\
\hline TOP : & SMAX= & -141.99 & SMIN \(=-1303.75\) & TMAX \(=\) & 580.88 & ANGLE= & -. 3 \\
\hline BOTT : & : SMAX \(=\) & 1303.75 & SMIN \(=141.99\) & TMAX \(=\) & 580.88 & ANGLE= & -. 3 \\
\hline \multirow[t]{2}{*}{110} & \multirow[t]{2}{*}{1} & -5.15 & . 77 & . 14 & . 02 & & . 00 \\
\hline & & 19.24 & 19.24 & . 00 & . 00 & & . 00 \\
\hline TOP : & : SMAX= & 20.47 & SMIN \(=\quad 2.75\) & TMAX \(=\) & 8.86 & ANGLE= & 2.0 \\
\hline \multirow[t]{3}{*}{BOTT :} & : SMAX= & -2.75 & SMIN \(=\quad-20.47\) & TMAX \(=\) & 8.86 & ANGLE= & 2.0 \\
\hline & \multirow[t]{3}{*}{2} & -16.13 & . 71 & . 40 & 8.37 & & . 04 \\
\hline & & 1227.04 & 1227.04 & . 00 & . 00 & & . 00 \\
\hline TOP : & & 1255.78 & SMIN= 59.66 & TMAX \(=\) & 598.06 & ANGLE= & -. 3 \\
\hline \multirow[t]{3}{*}{BOTT :} & : SMAX= & -59.66 & SMIN \(=-1255.78\) & TMAX \(=\) & 598.06 & ANGLE= & -. 3 \\
\hline & \multirow[t]{2}{*}{3} & -21.28 & 1.48 & 53 & 8.39 & & . 05 \\
\hline & & 1220.49 & 1220.49 & . 00 & . 00 & & . 00 \\
\hline TOP : & : SMAX= & 1258.56 & SMIN \(=\quad 80.10\) & TMAX \(=\) & 589.23 & ANGLE= & -. 3 \\
\hline \multirow[t]{3}{*}{BOTT:} & : SMAX= & -80.10 & SMIN \(=-1258.56\) & TMAX \(=\) & 589.23 & ANGLE= & -. 3 \\
\hline & \multirow[t]{2}{*}{4} & 10.98 & . 06 & -. 26 & -8.35 & & -. 04 \\
\hline & & 1233.85 & 1233.85 & . 00 & 606.89 & & . 00 \\
\hline TOP : & : SMAX= & -39.22 & SMIN \(=-1253.00\) & TMAX \(=\) & 606.89 & ANGLE \(=\) & -.3
-3 \\
\hline BOTT : & : SMAX \(=\) & 1253.00 & SMIN = \(\quad 39.22\) & TMAX \(=\) & 606.89 & ANGLE= & -. 3 \\
\hline
\end{tabular}

\footnotetext{
END OF ELEMENT FORCES*
}

\begin{tabular}{|c|c|c|c|c|c|c|c|}
\hline JOINT & LOAD & FORCE-X & FORCE-Y & FORCE-Z & MOM-X & MOM-Y & MOM Z \\
\hline -- & & \(\cdots\) & & & & & \\
\hline \multirow[t]{4}{*}{100} & 1 & . 00 & . 88 & . 00 & . 00 & . 00 & . 00 \\
\hline & 2 & . 00 & -. 61 & . 00 & . 00 & . 00 & . 00 \\
\hline & 3 & . 00 & . 28 & . 00 & . 00 & . 00 & . 00 \\
\hline & 4 & . 00 & 1.49 & . 00 & . 00 & . 00 & . 00 \\
\hline \multirow[t]{4}{*}{101} & 1 & . 00 & 1.56 & . 00 & . 00 & . 00 & . 00 \\
\hline & 2 & . 00 & -. 81 & . 00 & . 00 & . 00 & . 00 \\
\hline & 3 & . 00 & . 75 & . 00 & . 00 & . 00 & .00 \\
\hline & 4 & . 00 & 2.37 & . 00 & . 00 & . 00 & . 00 \\
\hline \multirow[t]{4}{*}{102} & 1 & . 00 & 1.77 & . 00 & . 00 & . 00 & . 00 \\
\hline & 2 & . 00 & -. 61 & . 00 & . 00 & . 00 & . 00 \\
\hline & 3 & . 00 & 1.16 & . 00 & . 00 & . 00 & . 00 \\
\hline & 4 & . 00 & 2.39 & . 00 & . 00 & . 00 & . 00 \\
\hline \multirow[t]{4}{*}{103} & 1 & . 00 & 1.89 & . 00 & . 00 & . 00 & . 00 \\
\hline & 2 & . 00 & -. 05 & . 00 & . 00 & . 00 & . 00 \\
\hline & 3 & . 00 & 1.84 & . 00 & . 00 & . 00 & . 00 \\
\hline & 4 & . 00 & 1.94 & . 00 & . 00 & . 00 & . 00 \\
\hline \multirow[t]{4}{*}{104} & 1 & .00 & 1.36 & . 00 & . 00 & . 00 & . 00 \\
\hline & 2 & . 00 & . 70 & . 00 & . 00 & . 00 & . 00 \\
\hline & 3 & . 00 & 2.06 & . 00 & . 00 & . 00 & . 00 \\
\hline & 4 & . 00 & . 67 & . 00 & . 00 & . 00 & . 00 \\
\hline \multirow[t]{4}{*}{105} & 1 & . 00 & . 79 & . 00 & . 00 & . 00 & . 00 \\
\hline & 2 & . 00 & 1.07 & . 00 & . 00 & . 00 & . 00 \\
\hline & 3 & . 00 & 1.86 & . 00 & . 00 & . 00 & . 00 \\
\hline & 4 & . 00 & -. 28 & . 00 & . 00 & . 00 & . 00 \\
\hline \multirow[t]{4}{*}{106} & 1 & . 00 & . 38 & . 00 & . 00 & . 00 & . 00 \\
\hline & 2 & . 00 & 1.12 & . 00 & . 00 & . 00 & . 00 \\
\hline & 3 & . 00 & 1.50 & . 00 & . 00 & . 00 & . 00 \\
\hline & 4 & . 00 & -. 74 & . 00 & . 00 & . 00 & . 00 \\
\hline \multirow[t]{4}{*}{107} & 1 & . 00 & . 10 & . 00 & . 00 & .00 & . 00 \\
\hline & 2 & . 00 & . 84 & . 00 & . 00 & . 00 & . 00 \\
\hline & 3 & . 00 & . 94 & . 00 & . 00 & . 00 & . 00 \\
\hline & - 4 & . 00 & -. 74 & . 00 & . 00 & . 00 & . 00 \\
\hline \multirow[t]{4}{*}{108} & 1 & . 00 & -. 10 & . 00 & . 00 & . 00 & . 00 \\
\hline & 2 & . 00 & . 20 & . 00 & . 00 & . 00 & . 00 \\
\hline & 3 & . 00 & . 10 & . 00 & . 00 & . 00 & . 00 \\
\hline & 4 & . 00 & -. 30 & . 00 & . 00 & . 00 & . 00 \\
\hline \multirow[t]{4}{*}{109} & 1 & . 00 & -. 24 & . 00 & . 00 & . 00 & . 00 \\
\hline & 2 & . 00 & -. 87 & . 00 & . 00 & . 00 & . 00 \\
\hline & 3 & . 00 & -1.11 & . 00 & . 00 & . 00 & . 00 \\
\hline & 4 & . 00 & . 63 & . 00 & . 00 & . 00 & . 00 \\
\hline \multirow[t]{4}{*}{110} & 1 & . 00 & -. 19 & . 00 & . 00 & . 00 & . 00 \\
\hline & 2 & . 00 & -1.22 & . 00 & . 00 & . 00 & . 00 \\
\hline & 3 & . 00 & -1.41 & . 00 & . 00 & . 00 & . 00 \\
\hline & 4 & . 00 & 1.04 & . 00 & . 00 & . 00 & .00
.00 \\
\hline \multirow[t]{4}{*}{111} & 1 & . 00 & . 69 & . 00 & .00
.00 & .00
.00 & . 00 \\
\hline & 2 & . 00 & -. 38 & . 00 & . 00 & . 00 & . 00 \\
\hline & 3 & . 00 & \(\begin{array}{r}.32 \\ \hline .07\end{array}\) & .00
.00 & . 00 & . 00 & . 00 \\
\hline & 4 & . 00 & 1.07 & . 00 & . 00 & . 00 & \\
\hline
\end{tabular}
\begin{tabular}{|c|c|c|c|c|c|c|c|}
\hline \multirow[t]{4}{*}{112} & 1 & . 00 & 1.21 & . 00 & . 00 & . 00 & 00 \\
\hline & 2 & . 00 & -. 50 & . 00 & . 00 & . 00 & . 00 \\
\hline & 3 & . 00 & . 70 & . 00 & . 00 & . 00 & . 00 \\
\hline & 4 & . 00 & 1.71 & . 00 & . 00 & . 00 & . 00 \\
\hline \multirow[t]{4}{*}{113} & 1 & . 00 & 1.36 & . 00 & . 00 & . 00 & . 00 \\
\hline & 2 & . 00 & -. 39 & . 00 & . 00 & . 00 & . 00 \\
\hline & 3 & . 00 & . 97 & . 00 & . 00 & . 00 & . 00 \\
\hline & 4 & . 00 & 1.74 & . 00 & . 00 & . 00 & . 00 \\
\hline \multirow[t]{4}{*}{114} & 1 & . 00 & 1.42 & . 00 & . 00 & . 00 & . 00 \\
\hline & 2 & . 00 & -. 04 & . 00 & . 00 & . 00 & . 00 \\
\hline & 3 & . 00 & 1.38 & . 00 & . 00 & . 00 & . 00 \\
\hline & 4 & . 00 & 1.45 & . 00 & . 00 & . 00 & . 00 \\
\hline \multirow[t]{4}{*}{115} & 1 & . 00 & 1.01 & . 00 & . 00 & . 00 & . 00 \\
\hline & 2 & . 00 & . 42 & . 00 & . 00 & . 00 & . 00 \\
\hline & 3 & . 00 & 1.43 & . 00 & . 00 & . 00 & . 00 \\
\hline & 4 & . 00 & . 58 & . 00 & . 00 & . 00 & . 00 \\
\hline \multirow[t]{4}{*}{116} & 1 & . 00 & . 58 & . 00 & . 00 & . 00 & . 00 \\
\hline & 2 & . 00 & . 65 & . 00 & . 00 & . 00 & . 00 \\
\hline & 3 & . 00 & 1.23 & . 00 & . 00 & . 00 & . 00 \\
\hline & 4 & . 00 & -. 07 & . 00 & . 00 & . 00 & . 00 \\
\hline \multirow[t]{4}{*}{117} & 1 & . 00 & . 29 & . 00 & . 00 & . 00 & . 00 \\
\hline & 2 & . 00 & . 68 & . 00 & . 00 & . 00 & . 00 \\
\hline & 3 & . 00 & . 97 & . 00 & . 00 & . 00 & . 00 \\
\hline & 4 & . 00 & -. 39 & . 00 & . 00 & . 00 & . 00 \\
\hline \multirow[t]{4}{*}{118} & 1 & . 00 & . 09 & . 00 & . 00 & . 00 & . 00 \\
\hline & 2 & . 00 & . 51 & . 00 & . 00 & . 00 & . 00 \\
\hline & 3 & . 00 & . 60 & . 00 & . 00 & . 00 & . 00 \\
\hline & 4 & . 00 & -. 42 & . 00 & . 00 & . 00 & . 00 \\
\hline \multirow[t]{4}{*}{119} & 1 & . 00 & -. 05 & . 00 & . 00 & . 00 & . 00 \\
\hline & 2 & . 00 & . 11 & . 00 & . 00 & . 00 & . 00 \\
\hline & 3 & . 00 & . 07 & . 00 & . 00 & . 00 & . 00 \\
\hline & 4 & . 00 & -. 16 & . 00 & . 00 & . 00 & . 00 \\
\hline \multirow[t]{4}{*}{120} & 1 & . 00 & -. 15 & . 00 & . 00 & . 00 & . 00 \\
\hline & 2 & . 00 & -. 55 & . 00 & . 00 & . 00 & . 00 \\
\hline & 3 & . 00 & -. 69 & . 00 & . 00 & . 00 & . 00 \\
\hline & 4 & . 00 & . 40 & . 00 & . 00 & . 00 & . 00 \\
\hline \multirow[t]{4}{*}{121} & 1 & . 00 & -. 12 & . 00 & . 00 & . 00 & . 00 \\
\hline & 2 & . 00 & -. 76 & . 00 & . 00 & . 00 & . 00 \\
\hline & 3 & . 00 & -. 88 & . 00 & . 00 & . 00 & . 00 \\
\hline & 4 & . 00 & . 64 & . 00 & . 00 & . 00 & . 00 \\
\hline \multirow[t]{4}{*}{122} & 1 & . 00 & . 27 & . 00 & . 00 & . 00 & . 00 \\
\hline & 2 & . 00 & -. 14 & . 00 & . 00 & . 00 & . 00 \\
\hline & 3 & . 00 & . 12 & . 00 & . 00 & . 00 & . 00 \\
\hline & 4 & . 00 & . 41 & . 00 & . 00 & . 00 & . 00 \\
\hline \multirow[t]{4}{*}{123} & 1 & . 00 & . 47 & . 00 & . 00 & . 00 & . 00 \\
\hline & 2 & . 00 & \(-.19\) & . 00 & . 00 & . 00 & . 00 \\
\hline & 3 & . 00 & . 27 & . 00 & . 00 & . 00 & . 00 \\
\hline & & . 00 & . 66 & . 00 & . 00 & . 00 & . 00 \\
\hline
\end{tabular}

JOINT LOAD FORCE-X FORCE
FORCE-Z
MOM-X
MOM-Y
MOM Z
\begin{tabular}{|c|c|c|c|c|c|c|c|}
\hline 124 & 1 & . 00 & . 52 & 00 & 00 & & \\
\hline & 2 & . 00 & -. 15 & . 00 & . 00 & . 00 & . 00 \\
\hline & 3 & . 00 & . 38 & . 00 & . 00 & . 00 & . 00 \\
\hline & 4 & . 00 & . 67 & . 00 & . 00 & . 00 & . 00 \\
\hline 125 & 1 & . 00 & . 54 & . 00 & . 00 & . 00 & . 00 \\
\hline & 2 & . 00 & -. 02 & . 00 & . 00 & . 00 & . 00 \\
\hline & 3 & .00 & . 53 & . 00 & . 00 & . 00 & . 00 \\
\hline & 4 & . 00 & . 56 & . 00 & . 00 & . 00 & . 00 \\
\hline 126 & 1 & . 00 & . 39 & . 00 & . 00 & . 00 & . 00 \\
\hline & 2 & . 00 & . 16 & . 00 & . 00 & . 00 & .00 \\
\hline & 3 & .00 & . 55 & . 00 & . 00 & . 00 & .00 \\
\hline & 4 & . 00 & . 23 & . 00 & . 00 & . 00 & . 00 \\
\hline 127 & 1 & . 00 & . 22 & . 00 & . 00 & . 00 & . 00 \\
\hline & 2 & . 00 & . 25 & . 00 & . 00 & . 00 & . 00 \\
\hline & 3 & . 00 & . 47 & . 00 & . 00 & . 00 & . 00 \\
\hline & 4 & . 00 & -. 03 & . 00 & . 00 & . 00 & . 00 \\
\hline 128 & 1 & . 00 & . 11 & . 00 & . 00 & . 00 & .00 \\
\hline & 2 & . 00 & . 26 & . 00 & . 00 & . 00 & . 00 \\
\hline & 3 & . 00 & . 37 & . 00 & . 00 & . 00 & . 00 \\
\hline & 4 & . 00 & -. 15 & . 00 & . 00 & . 00 & . 00 \\
\hline 129 & 1 & . 00 & . 03 & . 00 & . 00 & . 00 & 00 \\
\hline & 2 & . 00 & . 19 & . 00 & . 00 & . 00 & . 00 \\
\hline & 3 & . 00 & . 23 & . 00 & . 00 & . 00 & . 00 \\
\hline & 4 & . 00 & -. 16 & . 00 & .00 & . 00 & . 00 \\
\hline 130 & 1 & . 00 & -. 02 & . 00 & . 00 & . 00 & . 00 \\
\hline & 2 & . 00 & . 04 & . 00 & . 00 & . 00 & . 00 \\
\hline & 3 & . 00 & . 03 & . 00 & . 00 & . 00 & . 00 \\
\hline & 4 & . 00 & -. 06 & . 00 & . 00 & . 00 & . 00 \\
\hline 131 & 1 & . 00 & -. 06 & . 00 & . 00 & . 00 & . 00 \\
\hline & 2 & . 00 & -. 21 & . 00 & . 00 & . 00 & .00 \\
\hline & 3 & . 00 & -. 26 & . 00 & . 00 & . 00 & . 00 \\
\hline & 4 & . 00 & . 15 & . 00 & . 00 & . 00 & . 00 \\
\hline 132 & 1 & . 00 & \(-.04\) & . 00 & . 00 & . 00 & . 00 \\
\hline & 2 & . 00 & -. 29 & . 00 & . 00 & . 00 & . 00 \\
\hline & 3 & . 00 & -. 33 & . 00 & . 00 & . 00 & . 00 \\
\hline & 4 & . 00 & . 24 & . 00 & . 00 & . 00 & . 00 \\
\hline
\end{tabular}

\section*{Appendix C: Use STAAD-III to solve pavement design example 2}
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****************************************************

* S T A A D - III *
* Revision 22.3b *
* Proprietary Program of *
* Research Engineers, Inc. *
* Date= AUG 16, 2005 *
* Time= 9:16:3 *
* Build No. 2500 *
* USER ID: Kawneer Company Canada Limited

1. STAAD SPACE DOWEL-1
2. UNIT CM
3. JOINT COORDINATES
4. 1 0. 0. 0. 13 0 0 360 1
5. REPEAT 10 30 0 0
6. 144 300.5 0 0 156 300.5 0 360 1
7. REPEAT 10 30 0 0
8. MEMBER INCIDE
9. 301 131 144 313
9. ELEMENT INCIDENCES
10. 1 1 2 15 14 TO 12 1 1
11. REPEAT 9 12 13
12. 121 144 145 158 157 TO 132 1 1
13. REPEAT 9 12 13
14. MEMBER RELEASE
15. 301 TO 313 STAR'T FZ FX MX MY MZ
16. UNIT KN CM
17. MEMBER PROPERTIES
18.     * DOWEL BAR
19. 301 TO 313 PRIS AX 4.9 IX 3.83 IY 1.92 IZ 1.92
20. ELEMENT PROPERTIES
21. 1 TO 240 TH 20
22. UNIT KN METER
23. CONSTANTS
24. E 2E8 MEMBER 301 TO 313
25. E 27.6E6 ELEM 1 TO 240
26. POISSON 0.15 ELEM I TO 240
27. SUPPORT
28. 1 TO 286 ELASTIC MAT DIRECTION Y SUB 80000
29. LOAD 1
30. ELEMENT LOAD
31. 109 116 PR GY -667
32. PERFORM ANALYSIS
```
34. PRINT JOINT DISPL LIST 131 TO 143
\begin{tabular}{|c|c|c|c|c|c|c|c|}
\hline JOINT & LOAD & X -TRANS & Y-TRANS & Z-TRANS & X-ROTAN & Y-ROTAN & Z-ROTAN \\
\hline 131 & 1 & . 0000 & -. 0738 & . 0000 & -. 0004 & . 0000 & -. 0004 \\
\hline 132 & 1 & . 0000 & -. 0613 & . 0000 & -. 00004 & . 0000 & -. -.0004 \\
\hline 133 & 1 & . 0000 & -. 0490 & . 0000 & -. 00004 & . 0000 & -. 00003 \\
\hline 134 & 1 & . 0000 & -. 0396 & . 0000 & -. 0002 & . 0000 & -. .0003 \\
\hline 135 & 1 & . 0000 & -. 0339 & . 0000 & -. 0001 & . 0000 & -. 00002 \\
\hline 136 & 1 & . 0000 & -. 0316 & . 0000 & . 0000 & . 0000 & -. 00002 \\
\hline 137 & 1 & . 0000 & -. 0320 & . 0000 & . 0000 & . 0000 & -. 0002 \\
\hline 138 & 1 & . 0000 & -. 0332 & . 0000 & . 0000 & . 0000 & -. 0003 \\
\hline 139 & 1 & . 0000 & -. 0317 & . 0000 & -. 0001 & . 0000 & -. 0003 \\
\hline 140 & 1 & . 0000 & -. 0269 & . 0000 & -. 0002 & . 0000 & -. 0002 \\
\hline 141 & 1 & . 0000 & -. 0219 & . 0000 & -. 0002 & . 0000 & -. 0002 \\
\hline 142 & 1 & . 0000 & -. 0173 & . 0000 & -. 0001 & . 0000 & -. 0002 \\
\hline 143 & 1 & . 0000 & -. 0133 & . 0000 & -. 0001 & . 0000 & -. 0002 \\
\hline
\end{tabular}
35. PRINT MEMBER FORCE LIST 301 TO 313
\begin{tabular}{|c|c|c|c|c|c|c|c|c|}
\hline \multicolumn{3}{|l|}{MEMBER END FORCES} & \multicolumn{4}{|l|}{STRUCTURE TYPE = SPACE} & & \\
\hline \multicolumn{3}{|l|}{ALL UNITS ARE -- KN} & \multicolumn{2}{|l|}{METE} & & & & \\
\hline MEMBER & IOAD & JT & AXIAL & SHEAR-Y & SHEAR-Z & TORSION & MOM-Y & MOM-2 \\
\hline \multirow[t]{2}{*}{301} & 1 & 131 & . 00 & -13.11 & . 00 & . 00 & . 00 & . 00 \\
\hline & & 144 & . 00 & 13.11 & . 00 & . 00 & . 00 & -. 07 \\
\hline \multirow[t]{2}{*}{302} & 1 & 132 & . 00 & -11.88 & . 00 & . 00 & . 00 & . 00 \\
\hline & & \[
345
\] & . 00 & 11.88 & . 00 & . 00 & . 00 & -. 06 \\
\hline \multirow[t]{2}{*}{303} & 1 & 133 & . 00 & -1.05 & . 00 & . 00 & . 00 & . 00 \\
\hline & & 146 & . 00 & \[
1.05
\] & . 00 & . 00 & . 00 & -. 01 \\
\hline \multirow[t]{2}{*}{304} & 1 & 134 & . 00 & . 16 & . 00 & . 00 & . 00 & 00 \\
\hline & & 147 & . 00 & -. 16 & . 00 & . 00 & . 00 & . 00 \\
\hline \multirow[t]{2}{*}{305} & 1 & 135 & . 00 & . 21 & . 00 & .00 & . 00 & . 00 \\
\hline & & 148 & . 00 & -. 21 & . 00 & . 00 & . 00 & . 00 \\
\hline \multirow[t]{2}{*}{306} & 1 & 136 & . 00 & \[
-.10
\] & . 00 & . 00 & . 00 & . 00 \\
\hline & & 149 & \[
.00
\] & . 10 & . 00 & . 00 & .00 & . 00 \\
\hline \multirow[t]{2}{*}{307} & 1 & 137 & . 00 & \(-1.70\) & . 00 & . 00 & .00 & . 00 \\
\hline & & 150 & . 00 & 1.70 & . 00 & . 00 & . 00 & -. 01 \\
\hline \multirow[t]{2}{*}{308} & 1 & 138 & . 00 & -10.78 & . 00 & . 00 & . 00 & . 00 \\
\hline & & 151 & . 00 & 10.78 & . 00 & . 00 & . 00 & -. 05 \\
\hline \multirow[t]{2}{*}{309} & 1 & \[
139
\] & . 00 & -10.82 & . 00 & . 00 & . 00 & . 00 \\
\hline & & 152 & . 00 & 10.82 & . 00 & . 00 & . 00 & -. 05 \\
\hline \multirow[t]{2}{*}{310} & 1 & 140 & . 00 & -1.80 & . 00 & . 00 & . 00 & . 00 \\
\hline & & 153 & .00 & 1.80 & . 00 & .00 & . 00 & -. 01 \\
\hline \multirow[t]{2}{*}{311} & 1 & 141 & . 00 & -. 32 & . 00 & . 00 & . 00 & .00 \\
\hline & & 154 & . 00 & . 32 & . 00 & . 00 & . 00 & . 00 \\
\hline \multirow[t]{2}{*}{312} & 1 & 142 & . 00 & -. 02 & . 00 & . 00 & . 00 & . 00 \\
\hline & & 155 & . 00 & . 02 & . 00 & . 00 & . 00 & . 00 \\
\hline \multirow[t]{2}{*}{313} & 1 & 143 & . 00 & . 32 & . 00 & . 00 & .00 & . 00 \\
\hline & & 156 & . 00 & -. 32 & . 00 & . 00 & . 00 & . 00 \\
\hline
\end{tabular}
ELEMENT FORCES FORCE,LENGTH UNITS = KN METE

FORCE OR STRESS = FORCE/UNIT WIDTH/THICK, MOMENT = FORCE-LENGTH/UNIT WIDTH
\begin{tabular}{|c|c|c|c|c|c|c|c|c|}
\hline ELEMENT L & LOAD & \[
\begin{gathered}
\text { QX } \\
\text { VONT }
\end{gathered}
\] & \[
\begin{gathered}
\text { QY } \\
\text { VON }
\end{gathered}
\] & & \[
\begin{aligned}
& \text { MX } \\
& \text { FX }
\end{aligned}
\] & \[
\begin{aligned}
& M Y \\
& F Y
\end{aligned}
\] & & \[
\begin{aligned}
& \text { MXY } \\
& \text { FXY }
\end{aligned}
\] \\
\hline \multirow[t]{2}{*}{109} & \multirow[t]{2}{*}{1} & 64.60 & \multicolumn{2}{|l|}{42.18} & 1.04 & \multicolumn{2}{|l|}{-1.75} & -1.27 \\
\hline & & 491.91 & \multicolumn{2}{|l|}{491.91} & . 00 & . 00 & & 1. 00 \\
\hline \multirow[t]{2}{*}{BOTT:} & \multirow[t]{2}{*}{SMAX=} & 228.72 & SMIN= & -335.91 & TMAX \(=\) & 282.31 & ANGLE= & -2i.1 \\
\hline & & 335.91 & SMIN= & -228.72 & TMAX \(=\) & 282.31 & ANGLE= & -21.1 \\
\hline \multirow[t]{2}{*}{110} & \multirow[t]{2}{*}{1} & 150.34 & \multicolumn{2}{|l|}{26.91} & 3.40 & -. 79 & & -2.87 \\
\hline & & 943.98 & \multicolumn{2}{|l|}{943.98} & . 00 & . 00 & & . 00 \\
\hline \multirow[t]{2}{*}{BOTT:} & \multirow[t]{2}{*}{SMAX=} & 729.48 & SMIN= & -336.68 & TMAX \(=\) & 533.08 & ANGLE= & -26.9 \\
\hline & & \(=336.68\) & SMIN= & -729.48 & TMAX \(=\) & 533.08 & ANGLE \(=\) & -26.9 \\
\hline \multirow[t]{2}{*}{111} & \multirow[t]{2}{*}{1} & 70.62 & \multicolumn{2}{|l|}{-3. 10} & 6.41 & . 01 & & -2.11 \\
\hline & & 1106.09 & \multicolumn{2}{|l|}{1106.09} & . 00 & . 00 & & . 00 \\
\hline \multirow[t]{2}{*}{BOTT:} & : SMAX= & 1056.67 & SMIN= & -92.97 & TMAX \(=\) & 574.82 & ANGLE= & -16.7 \\
\hline & : SMAX \(=\) & 92.97 & SMIN= & -1056.67 & TMAX \(=\) & 574.82 & ANGLE= & \(-16.7\) \\
\hline \multirow[t]{2}{*}{112} & \multirow[t]{2}{*}{1} & 23.69 & \multicolumn{2}{|l|}{-6. 29} & 6.73 & . 35 & & -1.29 \\
\hline & & 1039.51 & \multicolumn{2}{|l|}{1039.51} & . 00 & . 00 & & . 00 \\
\hline TOP : & : SMAX \(=\) & 1046.77 & SMIN= & 14.67 & TMAX \(=\) & 516.05 & ANGLE= & -11.0 \\
\hline BOTT : & : SMAX= & -14.67 & SMIN= & -1046.77 & TMAX \(=\) & 516.05 & ANGLE= & -11.0 \\
\hline \multirow[t]{2}{*}{113} & \multirow[t]{2}{*}{1} & -15.39 & -5.5 & & 5.69 & . 32 & & -. 45 \\
\hline & & 839.38 & 839.3 & & . 00 & . 00 & & . 00 \\
\hline TOP : & : SMAX \(=\) & 859.52 & SMIN= & 41.83 & TMAX \(=\) & 408.84 & ANGLE= & -4.8 \\
\hline BOTT : & : SMAX= & - 41.83 & SMIN= & -859.52 & TMAX \(=\) & 408.84 & ANGLE= & -4.8 \\
\hline \multirow[t]{2}{*}{114} & \multirow[t]{2}{*}{1} & -58.67 & -. 1 & & 3.33 & -. 13 & & . 26 \\
\hline & & 514.01 & 514.0 & & . 00 & . 00 & & . 00 \\
\hline TOP : & : SMAX= & 502.11 & SMIN= & -23.04 & TMAX \(=\) & 262.57 & ANGLE= & 4.3 \\
\hline BOTT : & : SMAX= & - 23.04 & SMIN= & -502.11 & TMAX \(=\) & 262.57 & ANGLE= & 4.3 \\
\hline \multirow[t]{2}{*}{115} & \multirow[t]{2}{*}{1} & -137.99 & 14.3 & & -1.80 & -. 46 & & . 74 \\
\hline & & 310.17 & 310.1 & & . 00 & . 00 & & . 00 \\
\hline \multirow[t]{2}{*}{\[
\begin{aligned}
& \text { TOP } \\
& \text { BOTT }
\end{aligned}
\]} & : SMAX= & -19.61 & SMIN \(=\) & -319.51 & TMAX \(=\) & 149.95 & ANGLE= & -23.9 \\
\hline & : SMAX= & 319.51 & SMIN= & 19.61 & TMAX \(=\) & 149.95 & ANGLE= & -23.9 \\
\hline \multirow[t]{2}{*}{116} & \multirow[t]{2}{*}{1} & -3.23 & 25.1 & & -5.78 & -. 40 & & -. 42 \\
\hline & & 845.26 & 845.2 & & . 00 & . 00 & & . 00 \\
\hline & : SMAX \(=\) & -55.02 & SMIN= & -871.42 & TMAX \(=\) & 408.20 & ANGLE= & 4.4 \\
\hline BOTT : & : \(\operatorname{SMAX}=\) & 871.42 & SMIN= & 55.02 & TMAX \(=\) & 408.20 & ANGLE= & 4.4 \\
\hline \multirow[t]{2}{*}{117} & \multirow[t]{2}{*}{1} & 132.97 & 15.72 & & -3.44 & -. 54 & & -1.64 \\
\hline & & 642.48 & 642.48 & & . 00 & . 00 & & . 00 \\
\hline TOP : & : SMAX \(=\) & 30.21 & SMIN= & -626.84 & TMAX \(=\) & 328.53 & ANGLE= & 24.2 \\
\hline BOTT : & : SMAX= & 626.84 & SMIN \(=\) & -30.21 & TMAX \(=\) & 328.53 & ANGLE= & 24.2 \\
\hline \multirow[t]{2}{*}{118} & \multirow[t]{2}{*}{1} & 58.20 & 2.5 & & . 06 & -. 29 & & -1.34 \\
\hline & & 351.02 & 351.02 & & . 00 & . 00 & & . 00 \\
\hline
\end{tabular}

\begin{tabular}{|c|c|c|c|c|c|c|c|}
\hline \multicolumn{3}{|l|}{SUPPORT REACTIONS} & -UNIT KN & \multicolumn{4}{|l|}{E STRUCTURE TYPE = SPACE} \\
\hline JOINT & LOAD & FORCE-X & FORCE-Y & FORCE-Z & MOM-X & MOM-Y & MOM Z \\
\hline 131 & 1 & . 00 & 1.35 & . 00 & . 00 & . 00 & . 00 \\
\hline 132 & 1 & . 00 & 2.24 & . 00 & . 00 & . 00 & . 00 \\
\hline 133 & 1 & . 00 & 1.79 & . 00 & . 00 & . 00 & . 00 \\
\hline 134 & 1 & . 00 & 1.45 & . 00 & . 00 & . 00 & . 00 \\
\hline 135 & 1 & . 00 & 1.24 & . 00 & . 00 & . 00 & . 00 \\
\hline 136 & 1 & . 00 & 1.16 & . 00 & . 00 & . 00 & . 00 \\
\hline 137 & I & . 00 & 1.17 & . 00 & . 00 & . 00 & . 00 \\
\hline 138 & 1 & . 00 & 1.22 & . 00 & . 00 & . 00 & . 00 \\
\hline 139 & 1 & . 00 & 1.16 & . 00 & . 00 & . 00 & . 00 \\
\hline 140 & 1 & . 00 & . 99 & . 00 & . 00 & . 00 & . 00 \\
\hline 141 & 1 & . 00 & . 80 & . 00 & . 00 & . 00 & . 00 \\
\hline 142 & 1 & . 00 & . 63 & . 00 & . 00 & . 00 & . 00 \\
\hline 143 & 1 & . 00 & . 24 & . 00 & . 00 & . 00 & . 00 \\
\hline
\end{tabular}

\section*{Appendix D: Use KENSLABS to solve pavement design example 2}

\section*{1. General Information screen.}

Genera 1
TITLE design example 2
\begin{tabular}{|c|c|}
\hline Type of foundation [ \(0=\) liquid, 1=solid, 2-layer] [NFOUND] 0 & \multirow[t]{3}{*}{Default options are shown bs black dots. If not tive, pleas click the other bution.} \\
\hline Damage analysis [ \(0=\) no, 1=PCA criteria, \(2=u\) er specified] (NDAMA) 0 & \\
\hline Number of periods per year [NPY)1 & \\
\hline Number of load groups [ [NLG] 1 & \\
\hline Number of slab layers [ [NLAYER] 1 & With uniform load \\
\hline B ond between two slab layers [0=unbonded, 1=bonded) [NBOND] & without uniform load \\
\hline Number of slabs [ \({ }^{\text {a }}\) (NSLAB] 2 & \\
\hline Number of ioints [NJOINT] 1 & \(\bigcirc\) with temperature gradient and/or checking contact \\
\hline Nodal number for checking convergence [ [NNCK] 1 & \\
\hline Number of nodes for stress printout [ [NPRINT] \({ }^{\text {a }}\) & and/or checking contact \\
\hline Number of nodes on \(X\) axis of symmeliy \(\quad[\mathrm{NSX}] 0\) & \\
\hline Number of nodes on \(Y\) axis of symmetry \(\quad\) [NSY] 0 & with concentrated loa \\
\hline Hore detailed printout (0=no. \(1=\) yes) [ \({ }^{\text {a }}\) [MDPO] & \\
\hline Number of nodes with different thicknesses of slab layer 1 [NAT1] 0 & \\
\hline Number of nodes with different thicknesses of slab layer 2 (NAT2),0 & \\
\hline System of units ( \(0=\) English, \(1=51\) ] \({ }^{\text {a }}\) & Print Data Ser 7 \\
\hline
\end{tabular}

\section*{2. Slab Information screen.}


Use <Ctrl>-<Dels to delete a line, <Ctrl>-<lns> to insert a line. and < [

- JONO1/2/3/4 - joint number on four sides of each slab. Subscript ' 1 ' indicates the joint number on the left side of the slab, ' 2 ' on right side, ' 3 ' on the bottom and ' 4 ' on the top. If there is no joint on a given side, \(\mathrm{JONO}=' 0\) '.


\section*{3. Uniform load screen.}

Lorded Mas fom Mon Group No
23 M
Use <Ctrl>-<Del> to delete a line. <Ctrl>-<Ins> to insert a line, and <Del
Unit
\begin{tabular}{|c|c|c|c|c|c|c|}
\hline Urict. & & in. & in & in. & in. & psi \\
\hline Load Sequence & LS & XL1 & XL2 & YL1 & YL2 & QQ \\
\hline 1 & 1 & 145.7 & 157.5 & 0 & 11.8 & 96.7 \\
\hline 2 & 1 & 145.7 & 157.5 & 82.7 & 94.2 & 96.7 \\
\hline
\end{tabular}

\section*{4. Foundation Information screen.}


Double click (or press the \(E\)
pei
\begin{tabular}{|l|l|l|}
\hline & SUBME \\
\hline & 294.9 & NAS \\
\hline & 0 \\
\hline
\end{tabular}


\section*{5. Joint Information screen.}

\section*{Joint Infunation}

Double click anywhere on a line to get auxiliary form for NNAJ.
\begin{tabular}{|c|c|c|c|c|c|c|c|c|}
\hline Unit & psi & \multicolumn{2}{|l|}{in. -lbrin pai} & in & in. & 1 & 1 n & \\
\hline Joint No. & SPCON1 & SPCON2 & SCKV & BD & BS & WJ & GDC & NNAd \\
\hline - 1 & O & O & 1500000 & 98 & 11.6 & 2 & 0 & 9 \\
\hline
\end{tabular}
- SPCON1 - spring constant for shear transfer for each joint. SPCON1 = '0' if dowel bars are used. When grain interlock is used as loading transferring media, the stiffness of joint is represented by shear spring constant \(C_{\omega}\) and moment spring constant \(C_{\theta}\) as mentioned above in Chapter 2.
- SPCON2 - spring constant for moment transfer for each joint. SPCON2 = '0' if dowel bars are used. Since load is transferred across the joint mainly by shear, it is generally agreed that \(\mathrm{C}_{\theta}=0\) in most occasions.
- SCKV - modulus of dowel support K , also called steel-concrete K value (pci or \(\mathrm{KN} / \mathrm{m}^{3}\) ) as mentioned above in Chapter 2. Suggested \(K\) values range from \(3 \times 10^{5}\) to \(1.5 \times 10^{6} \mathrm{pci}\) ( 81.5 to \(409 \mathrm{GN} / \mathrm{m}^{3}\) ). If \(\mathrm{SPCON} 1 / 2 \not \boldsymbol{F}^{‘} 0\) ', \(\mathrm{SCKV}={ }^{\prime} 0\) '.
- BD - dowel bar diameter (inch or cm ). If \(\mathrm{SPCON} 1 / 2 \not \boldsymbol{\neq}^{\prime} 0^{\prime}, \mathrm{BD}={ }^{\prime} 0^{\prime}\).
- BS - dowel bar spacing (inch or cm ). If \(\mathrm{SPCON} 1 / 2 \neq{ }^{\prime} 0^{\prime}, \mathrm{BS}={ }^{\prime} 0\) '.
- WJ - width of the joint (inch or cm ). If \(\operatorname{SPCON} 1 / 2 \not \boldsymbol{\neq}^{\prime} 0^{\prime}, \mathrm{WJ}=' 0\) '.
- GDC - gap between dowel bars and concrete. In the case of no gap exists between dowel bars and concrete, \(\mathrm{GDC}=\) ' 0 '.
- NNAJ - number of nodes at each joint, maximum 15 nodes can be defined. In the case of dowel bars are uniformly distributed along the joint, \(\mathrm{NNAJ}=\) ' 0 '.

INPUT FILE NAME -C:\KENPAVE\design example 2.TXT
NUMBER OF PROBLEMS TO BE SOLVED = 1









CODE INDICATING WHETHER SLAB WEIGHT IS CONSIDERED (NWT)-----.--= 0
MAX NO. OF CYCLES FOR CHECKING CONTACT (NCYCLE)--.-.------------ 1
NUMBER OF ADDITIONAL THICKNESSES FOR SLAB LAYER 1 (NAT1)------= 0
NUMBER OF ADDITIONAL THICKNESSES FOR SLAB LAYER 2 (NAT2)------= 0
NUMBER OF POINTS ON X AXIS OF SYMMETRY (NSX) -......................... 0
NUMBER OF POINTS ON Y AXIS OF SYMMETRY (NSY)-........................... 0
MORE DETAILED PRINTOUT FOR EACH CONTACT CYCLE (MDPO)-.-......-. 0



(Length in in., force in lb, stress in psi, unit weigh in pcf subgrade and dowel \(K\) value in pci, and temperature in \(F\) )

FOR SLAB NO. I COORDINATES OF FINITE ELEMENT GRID ARE:
\(X=\begin{array}{llllll}X & 40 & 80 & 120 & 145.7 & 157.5\end{array}\)
\(Y=\begin{array}{lllllllll} & 11.8 & 25 & 39 & 53 & 67 & 82.7 & 110 & 141.7\end{array}\)
FOR SLAB NO. 2 COORDINATES OF FINITE ELEMENT GRID ARE:
\(\begin{array}{lllllll}\mathrm{X}= & 157.7 & 169.5 & 195.2 & 235.2 & 275.2 & 315.2\end{array}\)
\(Y=\begin{array}{lllllllll}0 & 11.8 & 25 & 39 & 53 & 67 & 82.7 & 110 & 141.7\end{array}\)
\begin{tabular}{cccc} 
LAYER & THICKNESS (T) & POISSON'S & YOUNG'S \\
NO. & & & RATIO (PR) \\
1 & 7.90000 & 0.15000 & MODULUS (YM) \\
& & & \(4.000 E+06\)
\end{tabular}

NO. OF LOADED AREAS (NUDL) FOR EACH LOAD GROUP ARE: 2
NO. OF NODAL FORCES (NCNF) AND MOMENTS (NCMX AND NCMY) ARE: 0000
FOR LOAD GROUP NO. 1 LOADS ARE APPLIED AS FOLLOWS:
\begin{tabular}{cccccc} 
SLAB NO. & \multicolumn{2}{c}{ X COORDINATES } & \multicolumn{2}{c}{ Y COORDINATES } & INTENSITY \\
(LS) & (XL1) & (XL2) & (YL1) & (YL2) & (QQ) \\
I & 145.70000 & 157.50000 & 0.00000 & 11.80000 & 96.70000 \\
1 & 145.70000 & 157.50000 & 82.70000 & 94.20000 & 96.70000
\end{tabular}

FOUNDATION ADJUSTMENT FACTOR (FSAF) FOR EACH PERIOD ARE: 1
NUMBER OF ADDITIONAL SUBGRADE MODULI (NAS) TO BE READ IN----- 0

NODAL COORDINATES (XN AND YN) OF INDIVIDUAL SLAB ARE:
\begin{tabular}{rrrrrrrrr}
1 & 0.000 & 0.000 & 2 & 0.000 & 11.800 & 3 & 0.000 & 25.000 \\
4 & 0.000 & 39.000 & 5 & 0.000 & 53.000 & 6 & 0.000 & 67.000 \\
7 & 0.000 & 82.700 & 8 & 0.000 & 110.000 & 9 & 0.000 & 141.700 \\
10 & 40.000 & 0.000 & 11 & 40.000 & 11.800 & 12 & 40.000 & 25.000 \\
13 & 40.000 & 39.000 & 14 & 40.000 & 53.000 & 15 & 40.000 & 67.000 \\
16 & 40.000 & 82.700 & 17 & 40.000 & 110.000 & 18 & 40.000 & 141.700 \\
19 & 80.000 & 0.000 & 20 & 80.000 & 11.800 & 21 & 80.000 & 25.000 \\
22 & 80.000 & 39.000 & 23 & 80.000 & 53.000 & 24 & 80.000 & 67.000 \\
25 & 80.000 & 82.700 & 26 & 80.000 & 110.000 & 27 & 80.000 & 141.700 \\
28 & 120.000 & 0.000 & 29 & 120.000 & 11.800 & 30 & 120.000 & 25.000 \\
31 & 120.000 & 39.000 & 32 & 120.000 & 53.000 & 33 & 120.000 & 67.000 \\
34 & 120.000 & 82.700 & 35 & 120.000 & 110.000 & 36 & 120.000 & 141.700 \\
37 & 145.700 & 0.000 & 38 & 145.700 & 11.800 & 39 & 145.700 & 25.000 \\
40 & 145.700 & 39.000 & 41 & 145.700 & 53.000 & 42 & 145.700 & 67.000 \\
43 & 145.700 & 82.700 & 44 & 145.700 & 110.000 & 45 & 145.700 & 141.700 \\
46 & 157.500 & 0.000 & 47 & 157.500 & 11.800 & 48 & 157.500 & 25.000 \\
49 & 157.500 & 39.000 & 50 & 157.500 & 53.000 & 51 & 157.500 & 67.000 \\
52 & 157.500 & 82.700 & 53 & 157.500 & 110.000 & 54 & 157.500 & 141.700 \\
55 & 157.700 & 0.000 & 56 & 157.700 & 11.800 & 57 & 157.700 & 25.000 \\
58 & 157.700 & 39.000 & 59 & 157.700 & 53.000 & 60 & 157.700 & 67.000 \\
61 & 157.700 & 82.700 & 62 & 157.700 & 110.000 & 63 & 157.700 & 141.700 \\
64 & 169.500 & 0.000 & 65 & 169.500 & 11.800 & 66 & 169.500 & 25.000 \\
67 & 169.500 & 39.000 & 68 & 169.500 & 53.000 & 69 & 169.500 & 67.000 \\
70 & 169.500 & 82.700 & 71 & 169.500 & 110.000 & 72 & 169.500 & 141.700 \\
73 & 195.200 & 0.000 & 74 & 195.200 & 11.800 & 75 & 195.200 & 25.000 \\
76 & 195.200 & 39.000 & 77 & 195.200 & 53.000 & 78 & 195.200 & 67.000 \\
79 & 195.200 & 82.700 & 80 & 195.200 & 110.000 & 81 & 195.200 & 141.700 \\
82 & 235.200 & 0.000 & 83 & 235.200 & 11.800 & 84 & 235.200 & 25.000 \\
85 & 235.200 & 39.000 & 86 & 235.200 & 53.000 & 87 & 235.200 & 67.000 \\
88 & 235.200 & 82.700 & 89 & 235.200 & 110.000 & 90 & 235.200 & 141.700 \\
91 & 275.200 & 0.000 & 92 & 275.200 & 11.800 & 93 & 275.200 & 25.000 \\
94 & 275.200 & 39.000 & 95 & 275.200 & 53.000 & 96 & 275.200 & 67.000 \\
97 & 275.200 & 82.700 & 98 & 275.200 & 110.000 & 99 & 275.200 & 141.700 \\
100 & 315.200 & 0.000 & 101 & 315.200 & 11.800 & 102 & 315.200 & 25.000 \\
103 & 315.200 & 39.000 & 104 & 315.200 & 53.000 & 105 & 315.200 & 67.000 \\
106 & 315.200 & 82.700 & 107 & 315.200 & 110.000 & 108 & 315.200 & 141.700
\end{tabular}

YOUNG MODULUS OF DOWEL BAR (YMSB) \(=2.900 \mathrm{E}+07\)
POISSON RATIO OF DOWEL BAR (PRSB) \(=0.30000\)
\begin{tabular}{ccccccccc} 
& & & \\
JOINT & SPRING & CONSTANT & MODULUS OF & DOWEL & DOWEL & JOINT & GAP & NODE \\
NO. & SHEAR & MOMENT & DOWEL SUP. & DIA. & SPACING & WIDTH & DOWEL & JOINT \\
& (SPCON1) & (SPCON2) & (SCKV) & (BD) & (BS) & (WJ) & (GDC) & (NNAJ) \\
1 & \(0.000 E+00\) & \(0.000 E+00\) & \(1.500 E+06\) & 0.980 & 11.800 & 0.200 & 0.00000 & 9
\end{tabular}

FOR JOINT NO. 1 DOWELS (BARNO) AT EACH NODE ARE: \(1 \begin{array}{lllllllll}1 & 1 & 1 & 1 & 1 & 1 & 1 & 1 & 1\end{array}\)

JOINT NO. 1 SPRING CONSTANT OF ONE DOWEL BAR \(=4.656 \mathrm{~F}+05\)
HALF BAND WIDTH (NB) \(=33\)
PERIOD I LOAD GROUP 1 AND CYCLE NO. 1
\begin{tabular}{rrrrrrrrrrrr} 
DEFLECTIONS OF SLABS & (F) ARE: & (DOWNWARD POSITIVE) & & \\
\hline 1 & -0.00028 & 2 & -0.00031 & 3 & -0.00033 & 4 & -0.00033 & -0.00031 \\
6 & -0.00027 & 7 & -0.00022 & 8 & -0.00014 & 9 & -0.00009 & 10 & -0.00032 \\
11 & -0.00046 & 12 & -0.00061 & 13 & -0.00071 & 14 & -0.00076 & 15 & -0.00076 \\
16 & -0.00072 & 17 & -0.00066 & 18 & -0.00073 & 19 & 0.00196 & 20 & 0.00128 \\
21 & 0.00059 & 22 & 0.00005 & 23 & -0.00026 & 24 & -0.00039 & 25 & -0.00040 \\
26 & -0.00048 & 27 & -0.00106 & 28 & 0.01197 & 29 & 0.00944 & 30 & 0.00690 \\
31 & 0.00495 & 32 & 0.00396 & 33 & 0.00373 & 34 & 0.00373 & 35 & 0.00293 \\
36 & 0.00063 & 37 & 0.02394 & 38 & 0.01924 & 39 & 0.01442 & 40 & 0.01085 \\
41 & 0.00925 & 42 & 0.00931 & 43 & 0.00997 & 44 & 0.00780 & 45 & 0.00354 \\
46 & 0.02891 & 47 & 0.02366 & 48 & 0.01795 & 49 & 0.01372 & 50 & 0.01187 \\
51 & 0.01208 & 52 & 0.01315 & 53 & 0.01027 & 54 & 0.00525 & 55 & 0.02321 \\
56 & 0.01984 & 57 & 0.01621 & 58 & 0.01310 & 59 & 0.01110 & 60 & 0.01001 \\
61 & 0.00918 & 62 & 0.00734 & 63 & 0.00491 & 64 & 0.01694 & 65 & 0.01442 \\
66 & 0.01173 & 67 & 0.00942 & 68 & 0.00788 & 69 & 0.00698 & 70 & 0.00630 \\
71 & 0.00501 & 72 & 0.00316 & 73 & 0.00655 & 74 & 0.00539 & 75 & 0.00415 \\
76 & 0.00310 & 77 & 0.00237 & 78 & 0.00194 & 79 & 0.00165 & 80 & 0.00119 \\
81 & 0.00031 & 82 & 0.00017 & 83 & -0.00009 & 84 & -0.00038 & 85 & -0.00060 \\
86 & -0.00074 & 87 & -0.00079 & 88 & -0.00078 & 89 & -0.00076 & 90 & -0.00095 \\
91 & -0.00055 & 92 & -0.00059 & 93 & -0.00062 & 94 & -0.00064 & 95 & -0.00064 \\
96 & -0.00061 & 97 & -0.00057 & 98 & -0.00050 & 99 & -0.00049 & 100 & -0.00011 \\
101 & -0.00011 & 102 & -0.00010 & 103 & -0.00009 & 104 & -0.00007 & 105 & -0.00004
\end{tabular}

FOR JOINT NO. I SHEAR (FAJ1) AND MOMENT (FAJ2) AT THE NODES ARE:
\begin{tabular}{rrrrrrrrr}
46 & -2652.8 & 0.0 & 47 & -1778.6 & 0.0 & 48 & -813.9 & 0.0 \\
49 & -288.3 & 0.0 & 50 & -357.4 & 0.0 & 51 & -962.3 & 0.0 \\
52 & -1851.8 & 0.0 & 53 & -1367.7 & 0.0 & 54 & -160.3 & 0.0
\end{tabular}

FOR JOINT NO. 1 SHEAR IN ONE DOWEL BAR (FAJPD) AT THE NODES IS:
\begin{tabular}{llllllllll}
46 & -2652.8 & 47 & -1778.6 & 48 & -813.9 & 49 & -288.3 & 50 & -357.4 \\
51 & -962.3 & 52 & -1851.8 & 53 & -1367.7 & 54 & -160.3 & &
\end{tabular}

FOR JOINT NO. 1 BEARING STRESS (BEARS) OF CONCRETE AND SHEAR STRESS
(SHEARS) OF DOWELS AT THE NODES ARE:
\begin{tabular}{rrrrrrrrr}
46 & -4226.2 & -3516.9 & 47 & -2833.5 & -2357.9 & 48 & -1296.6 & -1079.0 \\
49 & -459.4 & -382.3 & 50 & -569.4 & -473.8 & 51 & -1533.0 & -1275.7 \\
52 & -2950.2 & -2455.1 & 53 & -2179.0 & -1813.2 & 54 & -255.4 & -212.5
\end{tabular}

NODAL NUMBER AND REACTIVE PRESSURE (SUBR) ARE: (COMPRESSION POSITIVE)
\begin{tabular}{rrrrrrrrrr} 
\\
1 & -0.082 & 2 & -0.091 & 3 & -0.098 & 4 & -0.098 & 5 & -0.092 \\
6 & -0.080 & 7 & -0.064 & 8 & -0.040 & 9 & -0.027 & 10 & -0.093 \\
11 & -0.136 & 12 & -0.179 & 13 & -0.210 & 14 & -0.224 & 15 & -0.223 \\
16 & -0.212 & 17 & -0.195 & 18 & -0.214 & 19 & 0.577 & 20 & 0.378 \\
21 & 0.173 & 22 & 0.014 & 23 & -0.077 & 24 & -0.114 & 25 & -0.119 \\
26 & -0.142 & 27 & -0.313 & 28 & 3.529 & 29 & 2.785 & 30 & 2.036 \\
31 & 1.460 & 32 & 1.168 & 33 & 1.099 & 34 & 1.101 & 35 & 0.864 \\
36 & 0.187 & 37 & 7.059 & 38 & 5.674 & 39 & 4.252 & 40 & 3.198 \\
41 & 2.728 & 42 & 2.746 & 43 & 2.940 & 44 & 2.299 & 45 & 1.045 \\
46 & 8.526 & 47 & 6.979 & 48 & 5.295 & 49 & 4.045 & 50 & 3.499 \\
51 & 3.561 & 52 & 3.879 & 53 & 3.030 & 54 & 1.549 & 55 & 6.846
\end{tabular}
\begin{tabular}{rrrrrrrrrr}
56 & 5.852 & 57 & 4.779 & 58 & 3.862 & 59 & 3.273 & 60 & 2.951 \\
61 & 2.706 & 62 & 2.163 & 63 & 1.448 & 64 & 4.997 & 65 & 4.251 \\
66 & 3.459 & 67 & 2.778 & 68 & 2.323 & 69 & 2.057 & 70 & 1.858 \\
71 & 1.477 & 72 & 0.932 & 73 & 1.931 & 74 & 1.590 & 75 & 1.225 \\
76 & 0.914 & 77 & 0.700 & 78 & 0.572 & 79 & 0.486 & 80 & 0.352 \\
81 & 0.092 & 82 & 0.049 & 83 & -0.027 & 84 & -0.111 & 85 & -0.178 \\
86 & -0.217 & 87 & -0.232 & 88 & -0.230 & 89 & -0.226 & 90 & -0.279 \\
91 & -0.162 & 92 & -0.173 & 93 & -0.184 & 94 & -0.190 & 95 & -0.188 \\
96 & -0.180 & 97 & -0.167 & 98 & -0.148 & 99 & -0.144 & 100 & -0.032 \\
101 & -0.031 & 102 & -0.029 & 103 & -0.026 & 104 & -0.019 & 105 & -0.012 \\
106 & -0.002 & 107 & 0.012 & 108 & 0.024 & & & &
\end{tabular}

SUM OF FORCES (FOSUM) \(=26586.7\) SUM OF REACTIONS (SUBSUM) \(=26586.5\)
\begin{tabular}{|c|c|c|c|c|c|c|c|}
\hline NODE & LAYER & STRESS X & STRESS Y & STRESS XY & MAX.SHEAR & MAJOR & MINOR \\
\hline 1 & 1 & 0.000 & 0.000 & 0.000 & 0.000 & 0.000 & 0.000 \\
\hline 2 & 1 & 0.000 & 1.153 & 0.000 & 0.577 & 1.153 & 0.000 \\
\hline 3 & 1 & 0.000 & 1.991 & 0.000 & 0.995 & 1.991 & 0.000 \\
\hline 4 & 1 & 0.000 & 2.080 & 0.000 & 1.040 & 2.080 & 0.000 \\
\hline 5 & 1 & 0.000 & 1.575 & 0.000 & 0.788 & 1.575 & 0.000 \\
\hline 6 & 1 & 0.000 & 0.726 & 0.000 & 0.363 & 0.726 & 0.000 \\
\hline 7 & 1 & 0.000 & -0.368 & 0.000 & 0.184 & 0.000 & -0.368 \\
\hline 8 & 1 & 0.000 & -1.197 & 0.000 & 0.599 & 0.000 & -1.197 \\
\hline 9 & 1 & 0.000 & 0.000 & 0.000 & 0.000 & 0.000 & 0.000 \\
\hline 10 & 1 & 13.867 & 0.000 & 0.000 & 6.934 & 13.867 & 0.000 \\
\hline 11 & 1 & 12.338 & 3.496 & 8.187 & 9.304 & 17.222 & -1.387 \\
\hline 12 & 1 & 9.990 & 5.724 & 5.772 & 6.153 & 14.010 & 1.704 \\
\hline 13 & 1 & 7.756 & 6.156 & 3.810 & 3.893 & 10.849 & 3.063 \\
\hline 14 & 1 & 5.847 & 5.103 & 2.142 & 2.174 & 7.649 & 3.301 \\
\hline 15 & 1 & 4.141 & 3.147 & 0.914 & 1.040 & 4.685 & 2.604 \\
\hline 16 & 1 & 2.902 & 0.600 & 0.084 & 1.154 & 2.905 & 0.597 \\
\hline 17 & 1 & 2.358 & -2.857 & 1.341 & 2.932 & 2.682 & -3.182 \\
\hline 18 & 1 & -0.190 & 0.000 & 0.000 & 0.095 & 0.000 & -0.190 \\
\hline 19 & 1 & 78.442 & 0.000 & 0.000 & 39.221 & 78.442 & 0.000 \\
\hline 20 & 1 & 65.184 & 16.717 & 31.855 & 40.025 & 80.976 & 0.925 \\
\hline 21 & 1 & 55.654 & 26.270 & 24.709 & 28.747 & 69.709 & 12.215 \\
\hline 22 & 1 & 46.769 & 26.452 & 15.333 & 18.393 & 55.003 & 18.218 \\
\hline 23 & 1 & 41.380 & 21.296 & 6.481 & 11.952 & 43.290 & 19.386 \\
\hline 24 & 1 & 38.690 & 14.658 & 1.254 & 12.081 & 38.755 & 14.593 \\
\hline 25 & 1 & 34.446 & 5.330 & 1.034 & 14.595 & 34.482 & 5.293 \\
\hline 26 & 1 & 30.314 & \(-6.875\) & 8.356 & 20.385 & 32.105 & -8.666 \\
\hline 27 & 1 & 18.859 & 0.000 & 0.000 & 9.430 & 18.859 & 0.000 \\
\hline 28 & 1 & 162.783 & 0.000 & 0.000 & 81.391 & 162.783 & 0.000 \\
\hline 29 & 1 & 126.553 & 47.826 & 79.979 & 89.141 & 176.331 & -1.952 \\
\hline 30 & 1 & 94.411 & 78.372 & 69.461 & 69.922 & 156.314 & 16.469 \\
\hline 31 & 1 & 75.590 & 94.649 & 38.570 & 39.730 & 124.850 & 45.390 \\
\hline 32 & 1 & 69.531 & 76.414 & 7.866 & 8.586 & 81.558 & 64.387 \\
\hline 33 & 1 & 75.765 & 29.285 & -12.535 & 26.405 & 78.930 & 26.120 \\
\hline 34 & 1 & 91.133 & -11.753 & -0.718 & 51.448 & 91.138 & -11.758 \\
\hline 35 & 1 & 62.221 & -18.027 & 28.689 & 49.325 & 71.422 & -27.228
0.000 \\
\hline 36 & 1 & 38.453 & 0.000 & 0.000 & 19.226 & 38.453 & 0.000
-125.574 \\
\hline 37 & 1 & -125.574 & 0.000 & 0.000 & 62.787 & 0.000
72.979 & \\
\hline 38 & 1 & -44.995 & 15.727 & 82.184 & 87.612 & 72.979
178.170 & -102.246 \\
\hline 39 & 1 & 8.082 & 144.588 & 75.576 & 101.835 & 178.170 & -18.295 \\
\hline 40 & 1 & 28.942 & 171.916 & 40.442 & 82.134 & & 26.744 \\
\hline 41 & 1 & 26.884 & 145.548 & 4.084 & 59.472 & 145.688
76.249 & -2.007 \\
\hline 42 & 1 & 8.431 & 65.810 & -26.607 & 39.128 & 76.249 & -2.007 \\
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\end{tabular}
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